

# Communicating with a space probe using Software Defined Radio:

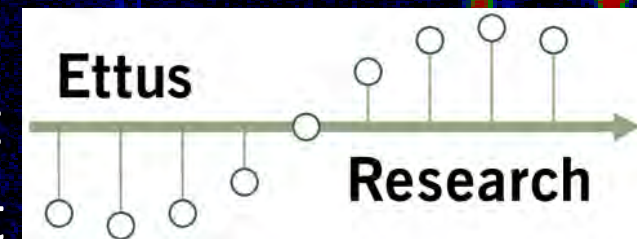
## The ISEE-3 Reboot Project

Balint Seeber

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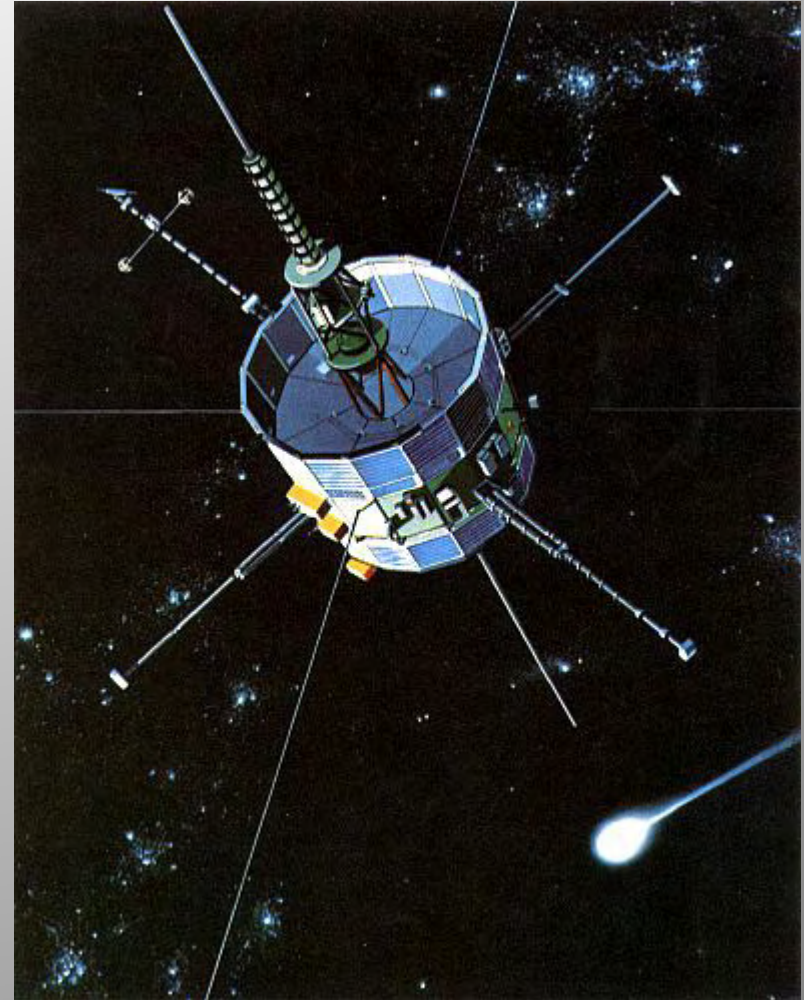
@spenchnet





# ISEE-3

- International Sun/Earth Explorer 3
- Launched: August 12, 1978
- Heliocentric Orbit
- Study interaction between solar wind and Earth's magnetic field

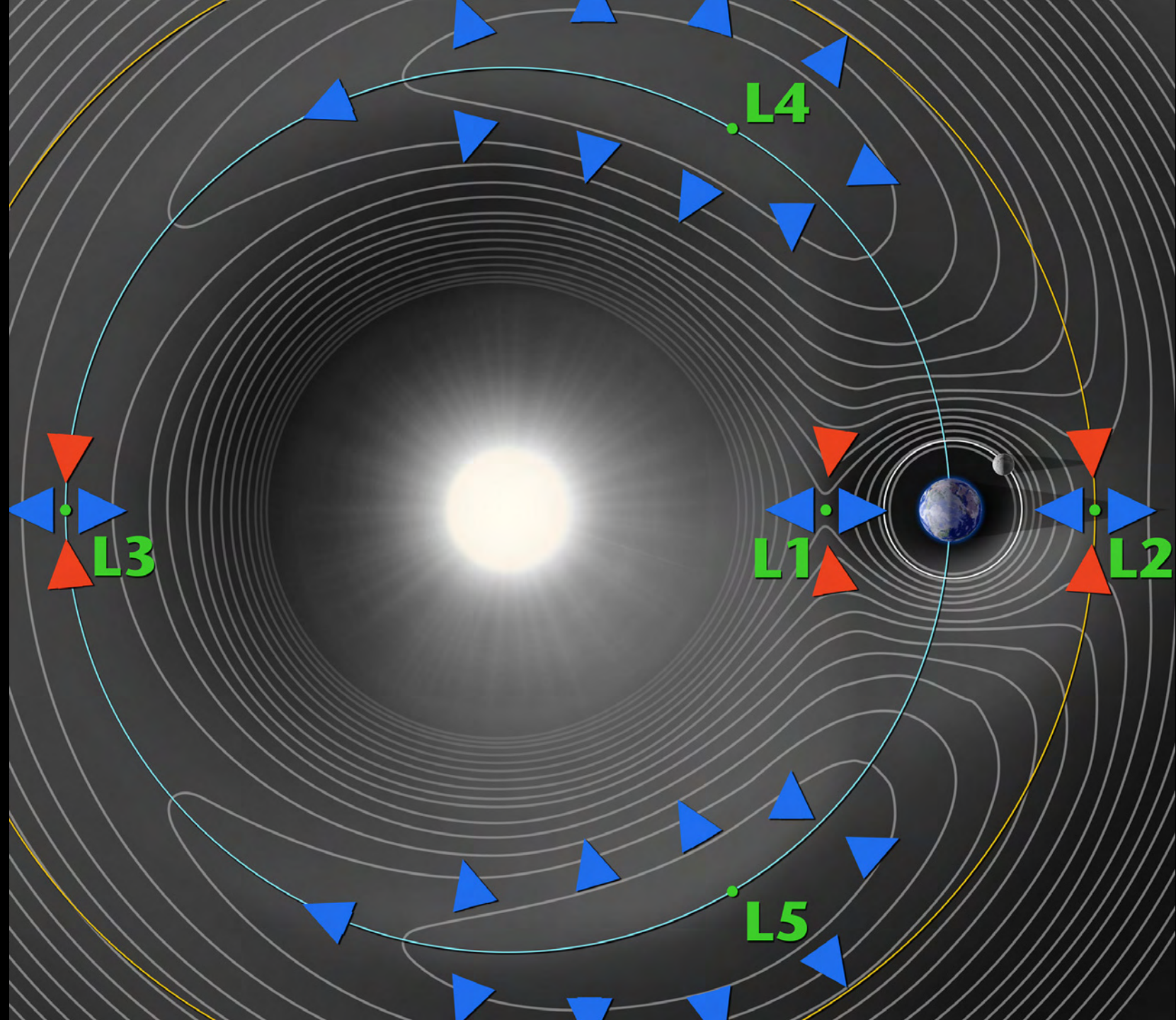




# ISEE-3

- Renamed ICE:  
International Cometary Explorer
- First spacecraft in halo orbit at an Earth-Sun L1 (Lagrange point)
- First spacecraft to pass through tail of a comet (Giacobini-Zinner)







COMET HALLEY  
3-28-86

COMET GIACOBINI-ZINNER  
9-11-85

9-1-82

11-23-83

11-23-82

9-27-83

12-22-83

3-30-83

6-30-83

HALO ORBIT  
6 MO. TRAVEL  
5 YR. ORBIT

MOON ORBIT

4-23-83

10-16-82

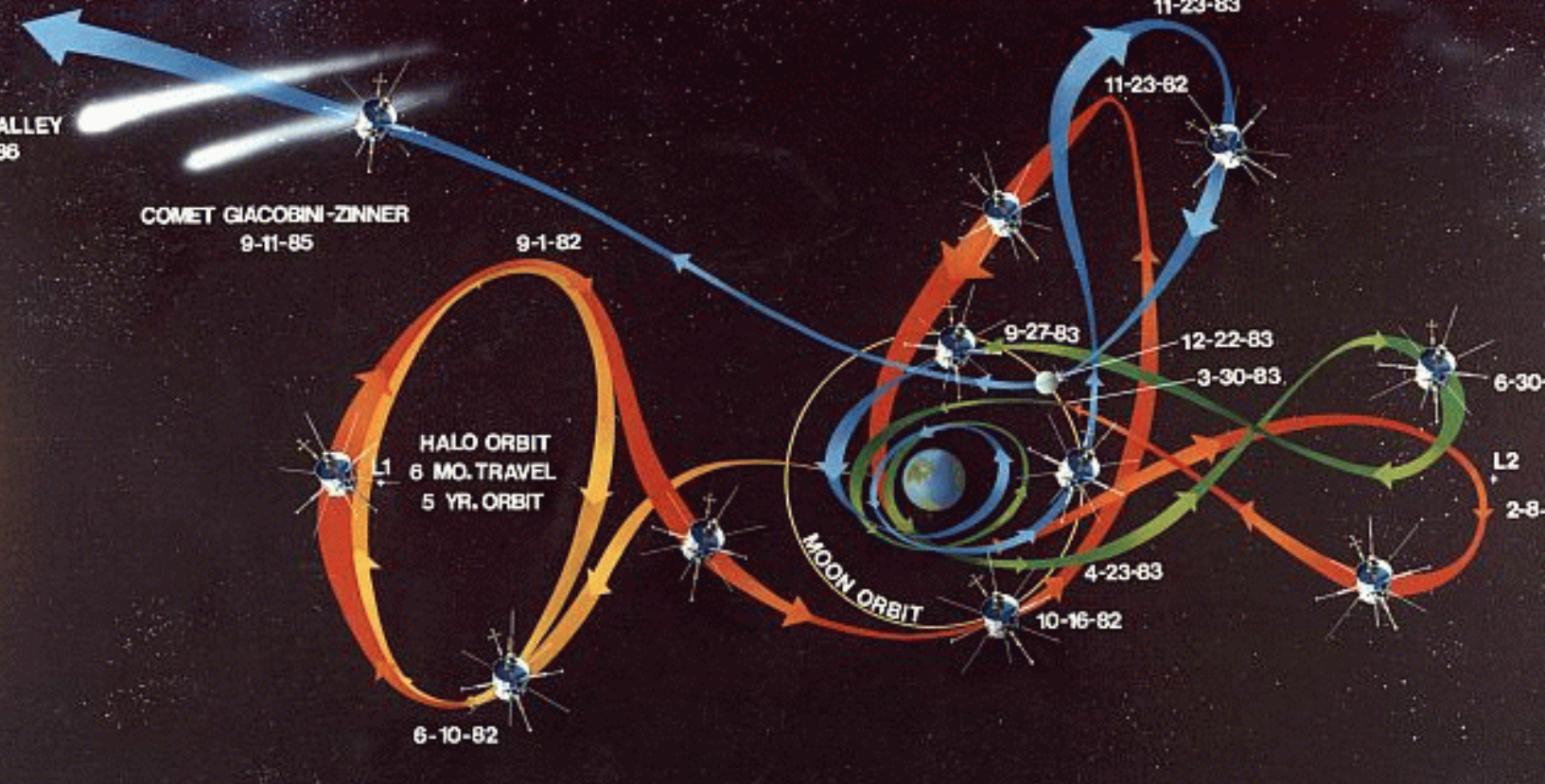
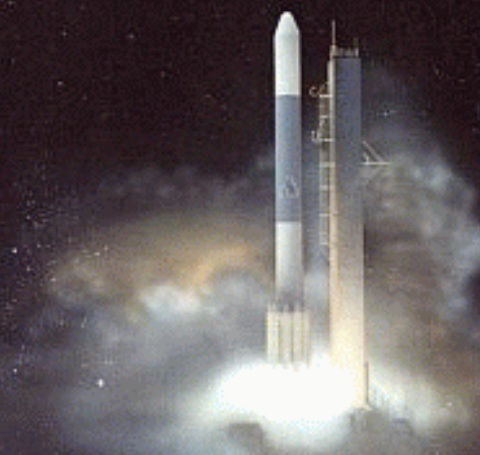
2-8-83

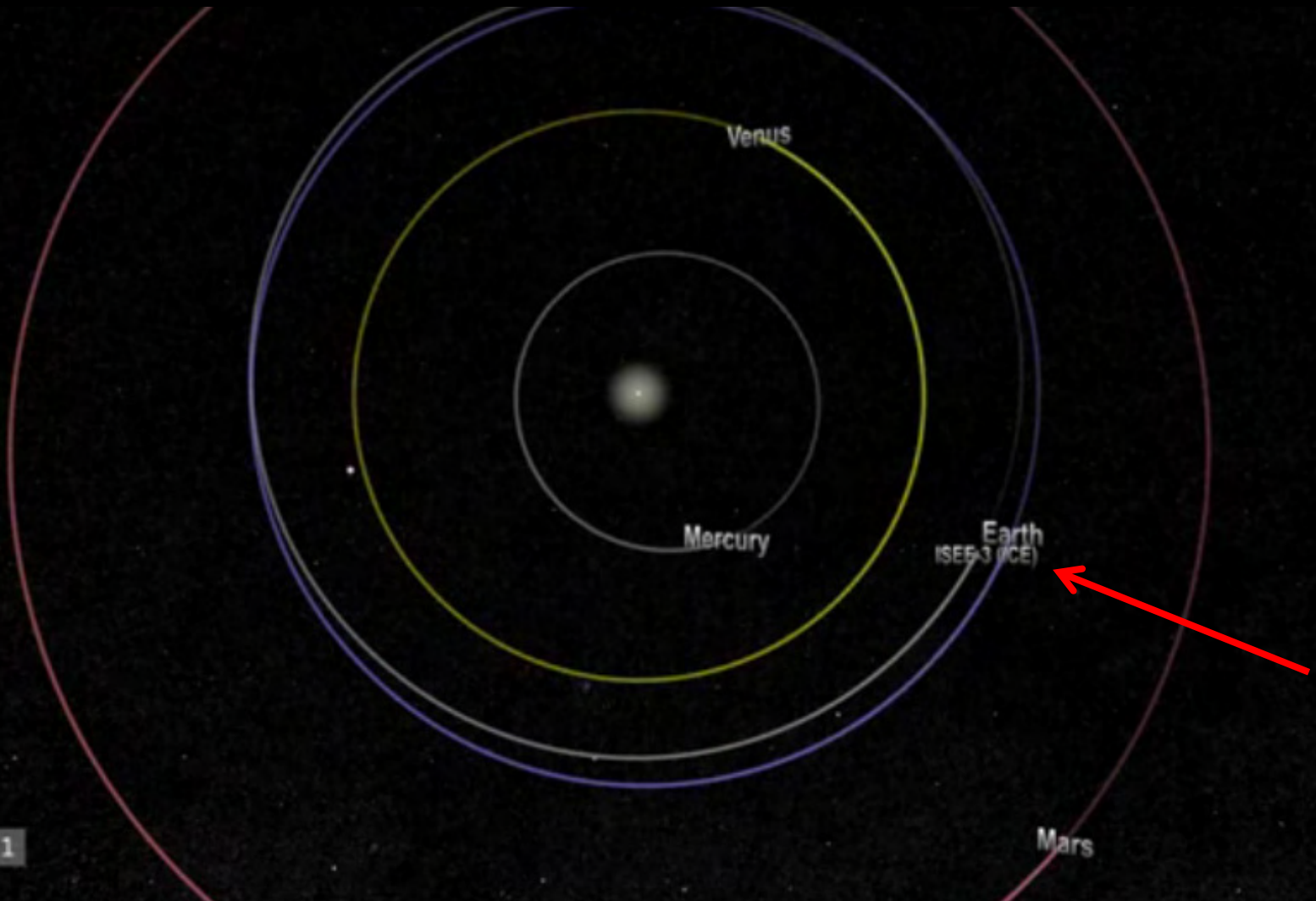
6-10-82

2012

**ISEE 3 MANEUVERS FROM LAUNCH  
TO HALO ORBIT  
TO COMET EXPLORATION**

DELTA 2914  
LAUNCHED AUGUST 12, 1978



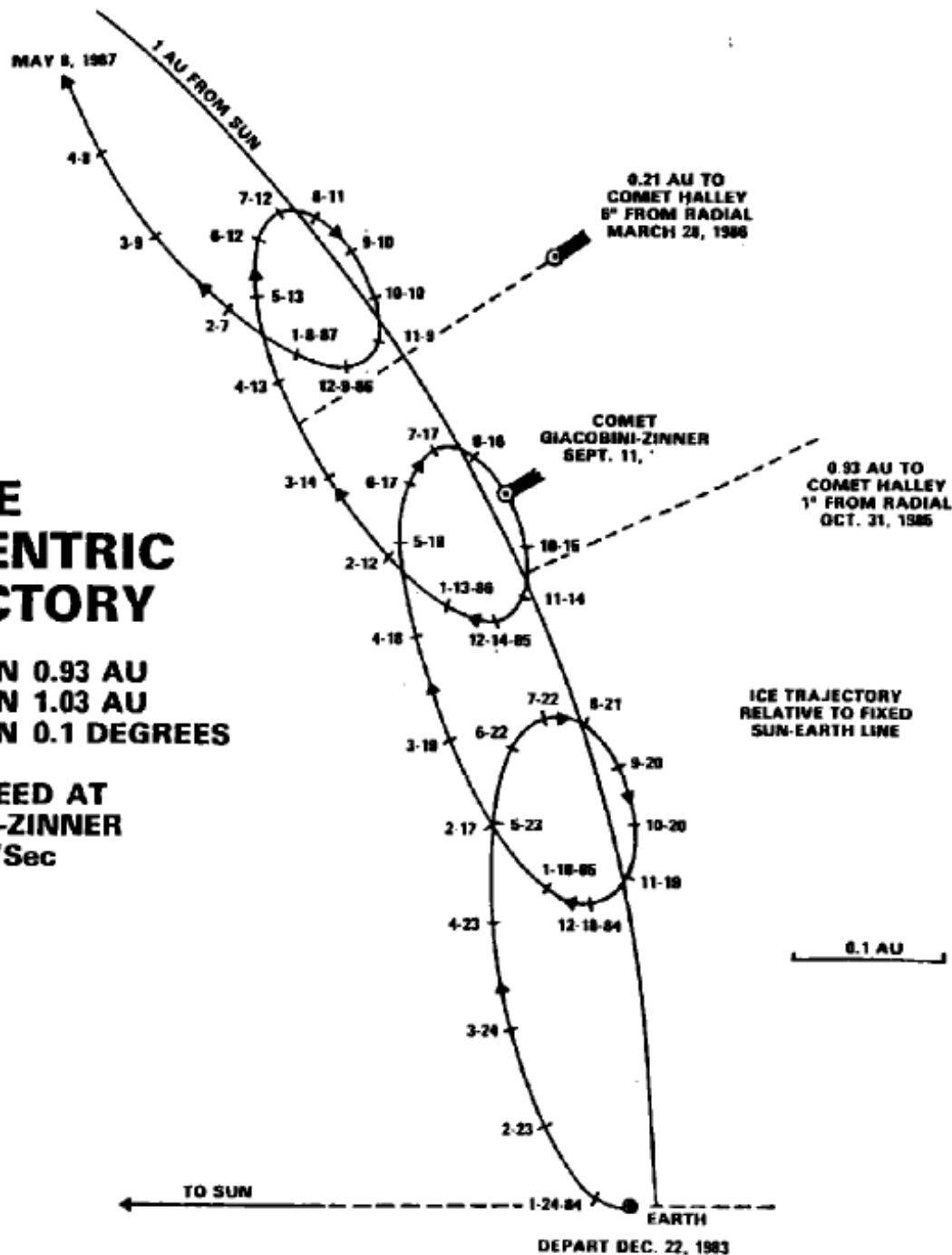


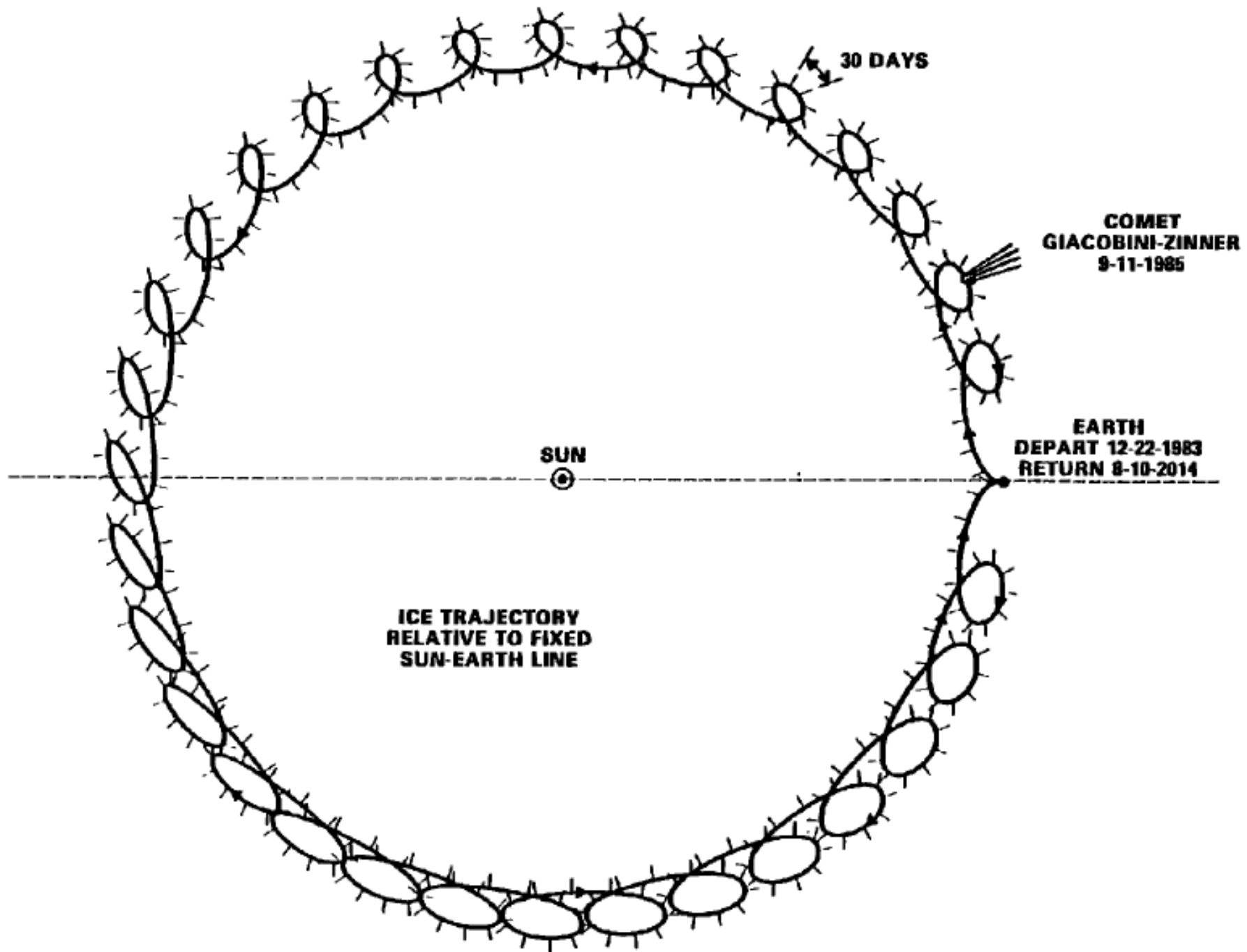
[http://en.wikipedia.org/wiki/File:ISEE-3 %28ICE%29 Revisits Earth.ogg](http://en.wikipedia.org/wiki/File:ISEE-3_%28ICE%29_Revisits_Earth.ogg)

# ICE HELIOCENTRIC TRAJECTORY

PERIHELION 0.93 AU  
APHELION 1.03 AU  
INCLINATION 0.1 DEGREES

FLYBY SPEED AT  
GIACOBINI-ZINNER  
21 Km/Sec

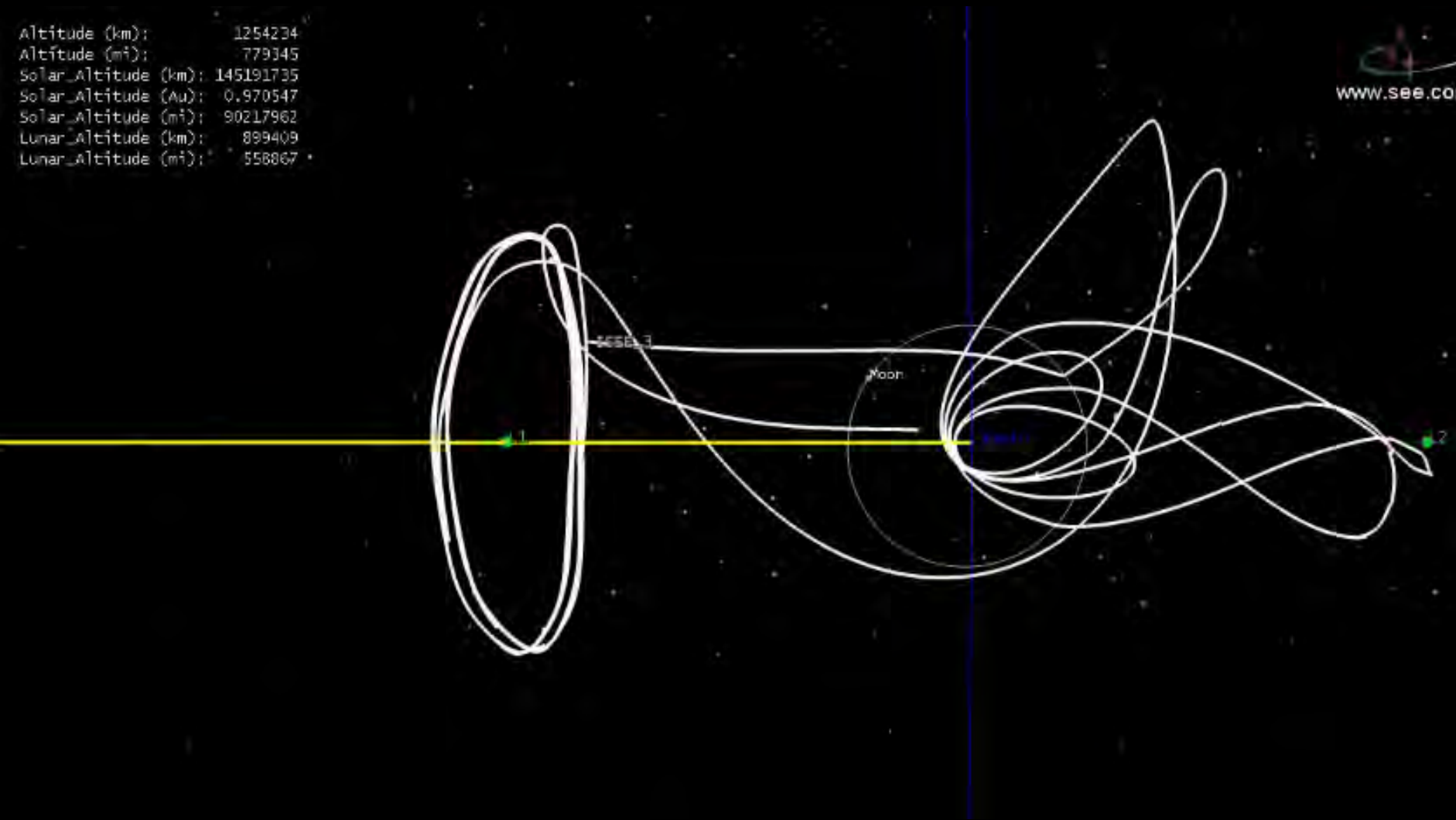








Altitude (km): 1254234  
Altitude (mi): 779345  
Solar\_Altitude (km): 145191735  
Solar\_Altitude (Au): 0.970547  
Solar\_Altitude (mi): 90217962  
Lunar\_Altitude (km): 899409  
Lunar\_Altitude (mi): 558867





**TOTAL S/C WEIGHT: 479 kg**

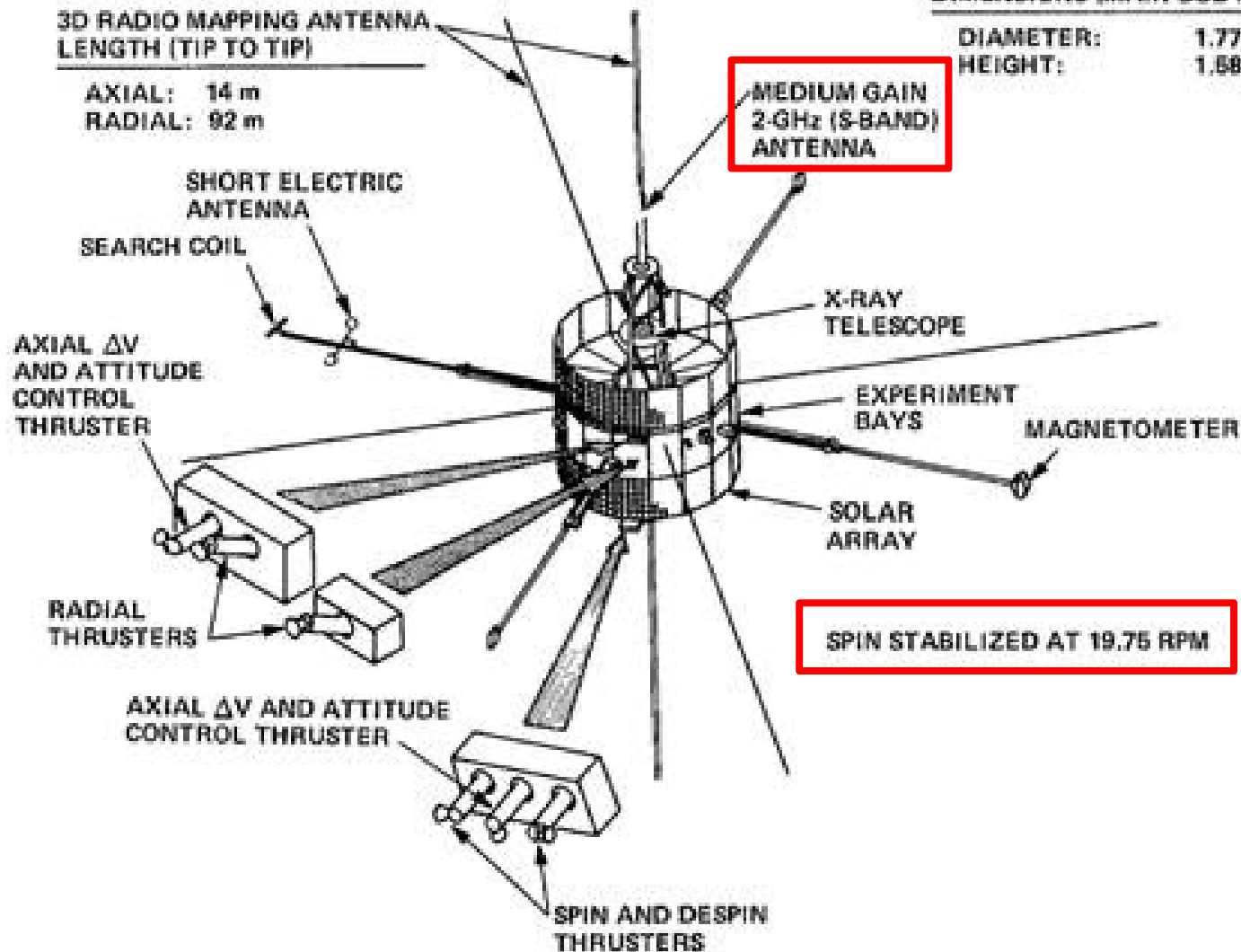
**EXPERIMENTS: 104 kg**

**HYDRAZINE: 89 kg**

**DIMENSIONS (MAIN BODY)**

**DIAMETER: 1.77 m**

**HEIGHT: 1.58 m**



# Science Instruments

**Table 1. ICE Investigations**

Title	Principal Investigator	Affiliation	Experiment Status
Solar Wind Plasma*	Bame	Los Alamos National Lab	Electrons only (Ion Portion Failed)
Plasma Composition*	Ogilvie	GSFC	Operational
Magnetometer*	Smith	JPL	Operational
Plasma Waves*	Scarf	TRW Systems	Operational
Energetic Protons*	Hynds	Imperial College, London	Operational
Radio Waves*	Steinberg	Paris Observatory, Meudon	Operational
X-Rays, Low Energy Electrons	Anderson	UCB	X-Rays and E > 200 keV (Low Energy Electron Portion Failed)
Low Energy Cosmic Rays	Hovestadt	MPI	Partial Failure (Ulezeq)
Medium Energy Cosmic Rays	von Rosenvinge	GSFC	Operational
High Energy Cosmic Rays	Stone	CIT	Partial Failure (Isotope Portion)
High Energy Cosmic Rays	Heckman	UCB/LBL	Partial Failure (Drift Chamber)
Cosmic Ray Electrons	Meyer	University of Chicago	Operational
Gamma Ray Bursts	Teegarden	GSFC	Partial Failure (PHA Memory)

# Old Telemetry Screen

```

ISEE-C:CPU1: 64;ACN:ORB 000;BUS V 28.29;ES CURR 1.34;NE CURR 6.69
OA 0.0; 0.000 RPM; 0.000 SEC;CMD CTR A,B 00; 79;S/C 037/22:24:49 (30261143)
S/C HSK, PAGE 4             RESET CTR A,B 640-639;GMT 074/22:18:08.115 70/03/15
-ATTITUDE AND ORBIT CONTROL SUBSYSTEM-  --- HYDRAZINE PROPULSION SYSTEM ---
- ELECTRONICS A - - ELECTRONICS B -
LOGIC PWR     ON  LOGIC PWR     ON
+28V PWR     ON  +28V PWR     OFF
TSL           0101010 010010
SINIT 01100   OFF  SINIT 10110 10001
SECT WIDTH   360   SECT WIDTH   OFF
FIRINGS      36    FIRINGS      77
RATIO FIRING DIS  RATIO FIRING DIS
THRUST RATI  2     THRUST RATI  114
MANEUVER     TERM  MANEUVER     INIT
MANEUP COMPL NO   MANEUP COMPL YES

PRI HTRS 1/2  LOW   ACCL CTR 1/2  110
SEC HTRS 1/2  OFF   ACCL T 1/2  24.4
ACL PWR 1/2  2.50 T PRI TK HTRS  OFF
PRI TK HTRS100100 SEC TK HTRS  OFF
SEC TK10110 10011 LATCH VALVB  OFF
LATCH VALVA OPEN  LATCH VALVD OPEN
LATCH VALVC CLOS  THERMO CPLF 346.2
THERMO CPL  248.6  TANK PRESS  2.4
TANK PRESS  2.7
  
```

# Loss of Interest

- Original mission drew to a close
- Telemetry disabled
  - Only transponders' carriers remain enabled
- NASA decommissioned ground-based equipment originally used to communicate with spacecraft
  - **Deep Space Network** now incapable of sending commands
- Documentation has been scattered

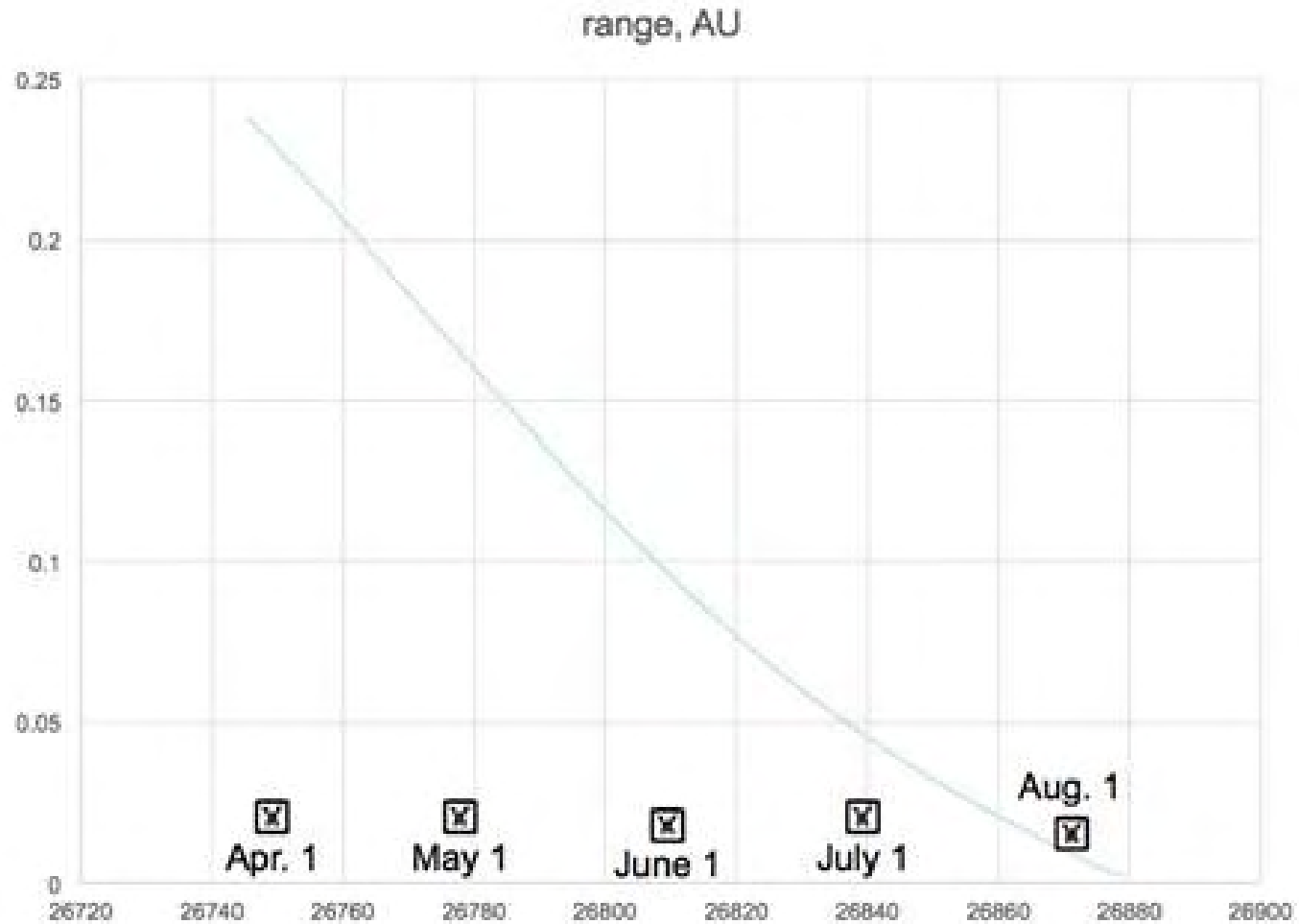


# (New) Mission

- Led by Dennis Wingo (Skycorp Inc) & Keith Cowing (Space College/NASA Watch/SpaceRef)
- Re-capture ISEE-3 (command the spacecraft)
  - Enable telemetry (and decode)
  - Fire thrusters
- Bring into stable high-Earth orbit
- Allow open, public access to remaining operational scientific capabilities
- “NASA Signs Agreement with Citizen Scientists Attempting to Communicate with Old Spacecraft”



# Closing In



Modified Julian Date (days, until Aug. 9 perigee)



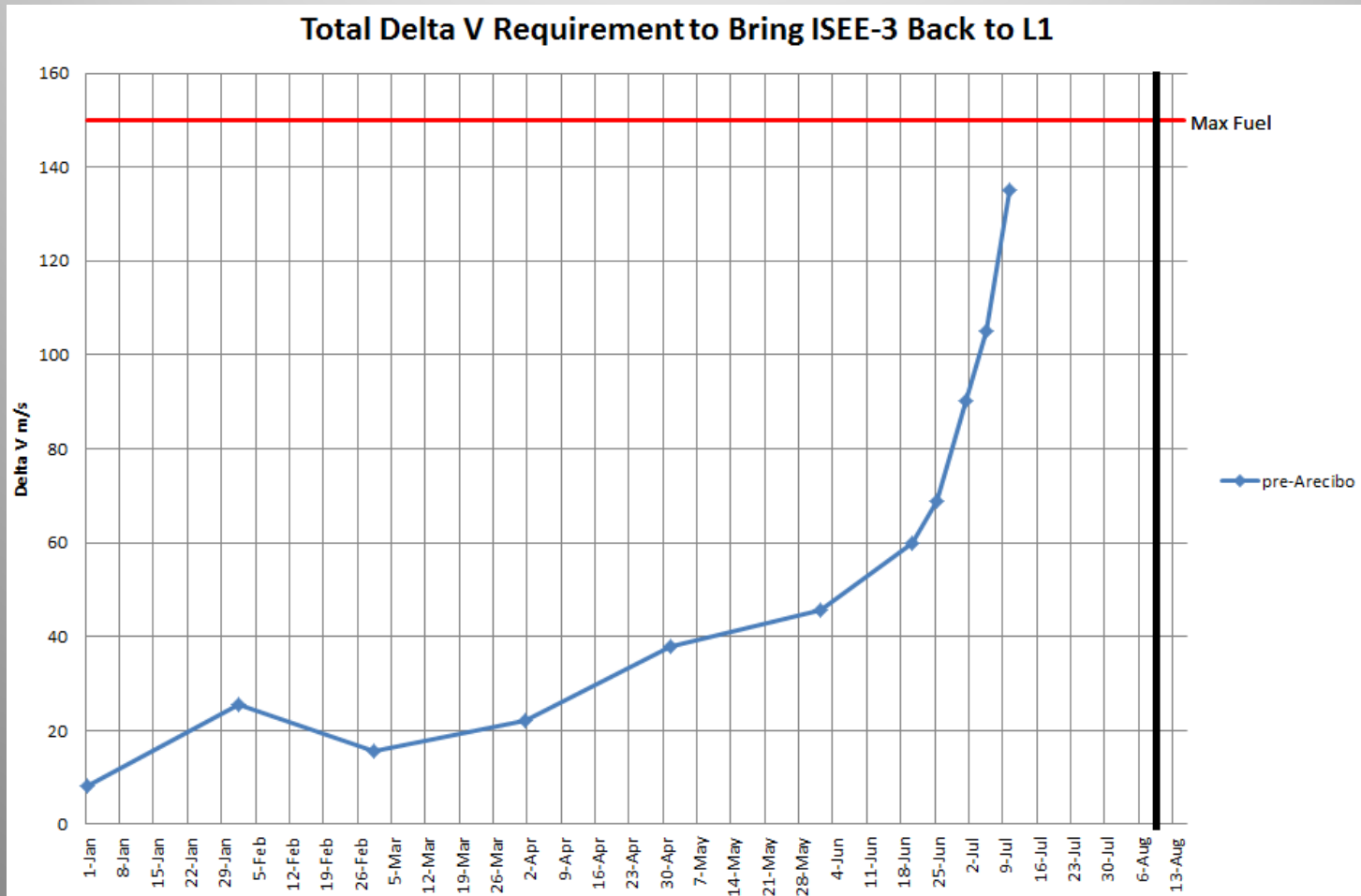
# A Sense of Urgency

<b>Date</b>	<b>Total <math>\Delta V</math>, m/s</b>	<b>% <math>\Delta V</math> capacity</b>	<b>Earth Dist., km</b>	<b>Dist. AU</b>
2013 Aug 1	6.351	4.23	28167878	0.1883
2013 Sep 1	4.720	3.15	28515162	0.1906
2013 Oct 1	4.130	2.75	32859724	0.2197
2013 Nov 1	6.280	4.19	39182694	0.2619
2013 Dec 1	7.352	4.90	44871161	0.2999
2014 Jan 1	8.355	5.57	48294364	0.3228
2014 Feb 1	25.645	17.10	47706709	0.3189
2014 Mar 1	15.691	10.46	43242137	0.2891
2014 Apr. 1	22.260	14.84	34636672	0.2315
2014 May 1	38.029	25.35	24450043	0.1634
2014 Jun 1	45.718	30.48	14415499	0.0964
2014 Jul 1	90.318	60.21	6875760	0.0460
2014 Aug 1	360.000	240.00	1435851	0.0096



# A Sense of Urgency

- Limited fuel remains for  $\Delta V$  manoeuvres



# Success Case

Far x-axis crossings

2014 Dec. 31, Dist. 255 Re

2015 July 1, Dist. 269 Re

2015 Dec. 28, Dist. 267 Re

Near x-axis crossings

2015 Mar. 30, Dist. 178 Re

2015 Sept. 29, Dist. 188 Re

End  
2016  
Jan. 17

To Sun

Lunar  
orbit

Perigee Oct. 26  
Dist. 24.55 Re

S6 Aug. 10, 20:47 UT  
h 117 km, Z +205 km

Earth

Perigee Aug. 9,  
4:57 UT, 27 Re

2013 Nov. 4, 17:00 UT

V 4.69 m/s

-2.94 m/s in V direction

-3.65 m/s normal to  
the heliocentric orbit

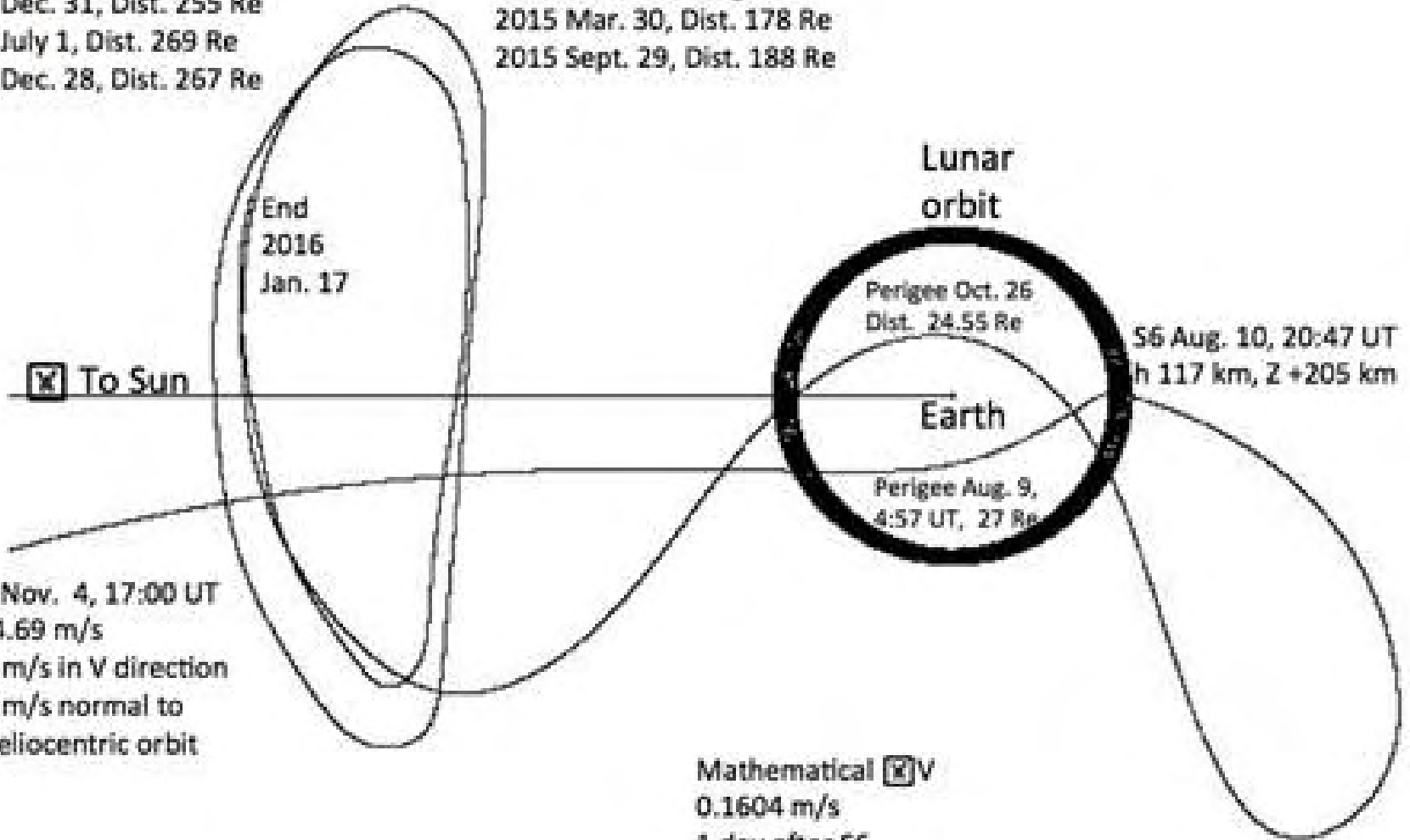
Mathematical  V

0.1604 m/s

1 day after S6

Apogee Sept. 16

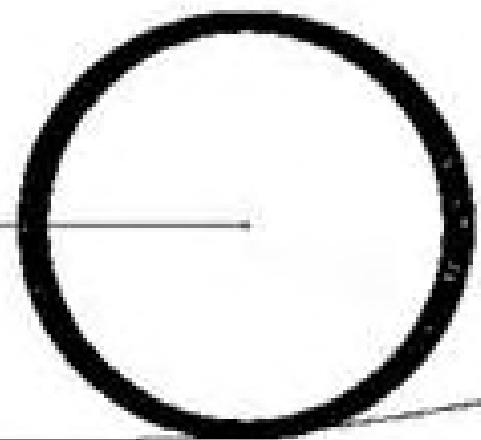
Dist. 230.49 Re





# Failure Case

After this Earth swingby,  
The heliocentric period  
will be 391.44 days;  
ISEE-3/ICE would then  
return to the Earth's  
vicinity after 14  
revolutions in 15 years, in  
August 2029.



Perigee  
2014 Aug. 9, 11:43 UT  
Dist. 63 Re

# Crowdfunding



ROCKETHUB  
The world's crowdfunding machine.

SUCCESS SCHOOL

OUR MOVEMENT

LATEST NEWS

Join | Login

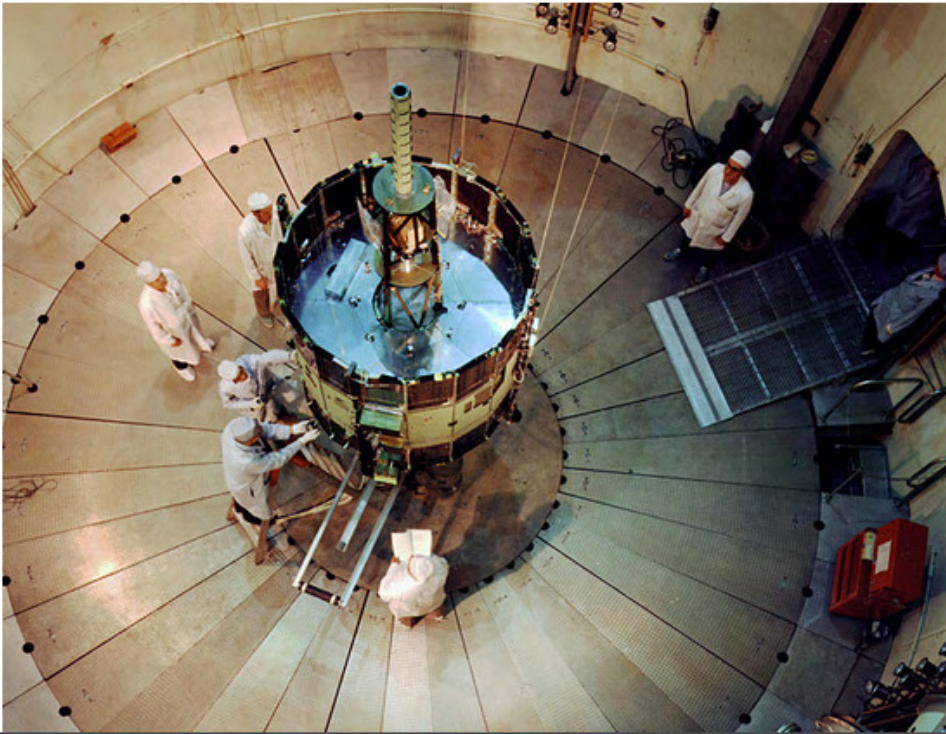
## ISEE-3 Reboot Project by Space College, Skycorp, and SpaceRef

Description

About

Conversations

Activity



### Project Leaders



Dennis Wingo



Keith Cowing



128%

**\$159,602**

Raised towards  
\$125,000 goal

**2238 Funders**

Ended 05/23/14

Complete!

PROJECT COMPLETE

### Funders



[View All](#)

# Communications

- Radio communications required to contact & command the space craft
- Original equipment not available
- Need to quickly create a new modulator & demodulator
- Enter Software Defined Radio...

# Create a new MODEM

## Ettus Research Helps Power ISEE-3 Reboot Effort

By Keith Cowing on May 8, 2014 5:03 PM



[USRP N210](#) units delivered to ISEE-3 Mission Control today. "The USRP N210 provides high-bandwidth, high-dynamic range processing capability. The USRP N210 is intended for demanding communications applications requiring this type of rapid development" [Larger image](#).

Skycorp Incorporated and Ettus Research, a subsidiary of National Instruments, have joined forces for the development of a crucial piece of hardware needed to contact the ISEE-3 spacecraft. Contacting ISEE-3, launched by NASA in 1978, is the focus of the ISEE-3 Reboot Project. ISEE-3 will be returning to Earth orbit in August of this year after having circled the sun for nearly four decades. SpaceRef Interactive, Skycorp, and Space College have joined forces to rescue the spacecraft, put it back into orbit near Earth, and use it for scientific research and STEM education.



# Long-distance Communications

- ~15.5 million km
- Spacecraft has a 5W transmitter
  - Two transponders (up- & down-link)
  - Two antennas (medium & intermediate gain)
- Not-your-average link budget



# Python Real-time Tracker

- Uses predictions from JPL HORIZONS service
- Originally based on ephemeris updated in 2008
  - Derived from trajectory analysis by KinetX Aerospace
- Spacecraft has deviated from predicted path
  - Required manual sky search to find peak carrier signal
- Now updated from Arecibo ‘spider search’



# Python Real-time Tracker

```
UTC   : 2014-05-28 07:50:06.234132  
Local: 2014-05-28 03:50:06.234096 (-4.0)
```

```
Lines: 471/2881 (2410 left)
```

```
Speed (km/s) : -3.4829406  
Speed (m/s)  : -3482.9406368  
Speed (km/hr): -12538.5862925
```

```
Dist (AU) : 0.10369466811595  
Dist (km) : 15512501.553089
```

```
Light time (one-way) : 51.744135 s  
Light time (two-way) : 103.488271 s
```

```
R.A.: 7.7720059526  
Decl: +21.4076608943  
(adjusted for light time)
```

```
Downlink frequencies:
```

```
2.270400000 GHz: 2.270426377 GHz (+26.377449 kHz)  
2.217500000 GHz: 2.217525763 GHz (+25.762858 kHz)
```

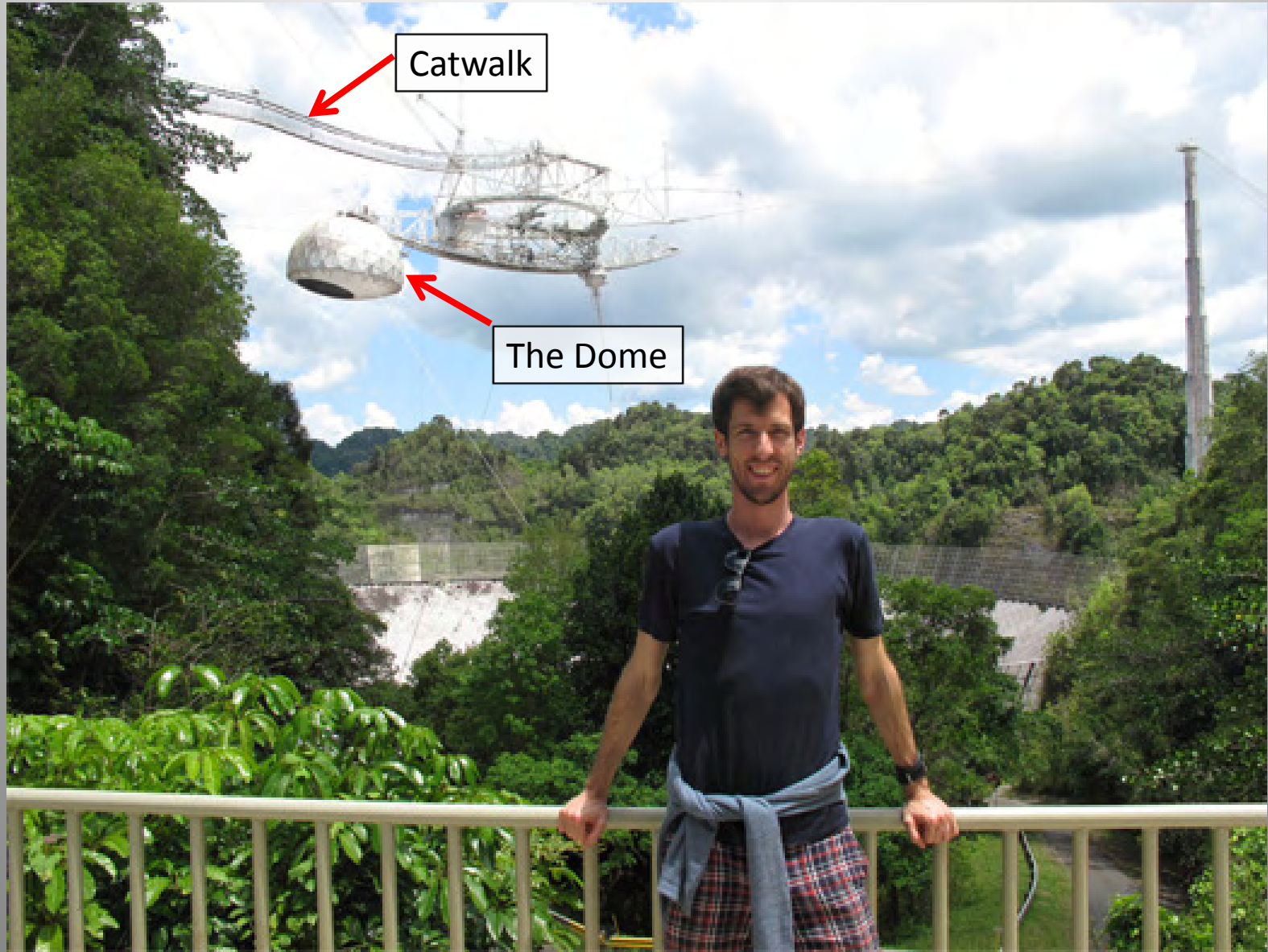
# Long-distance Communications

- Need:
  - a big amplifier
  - a big dish





# Arecibo Radio Observatory

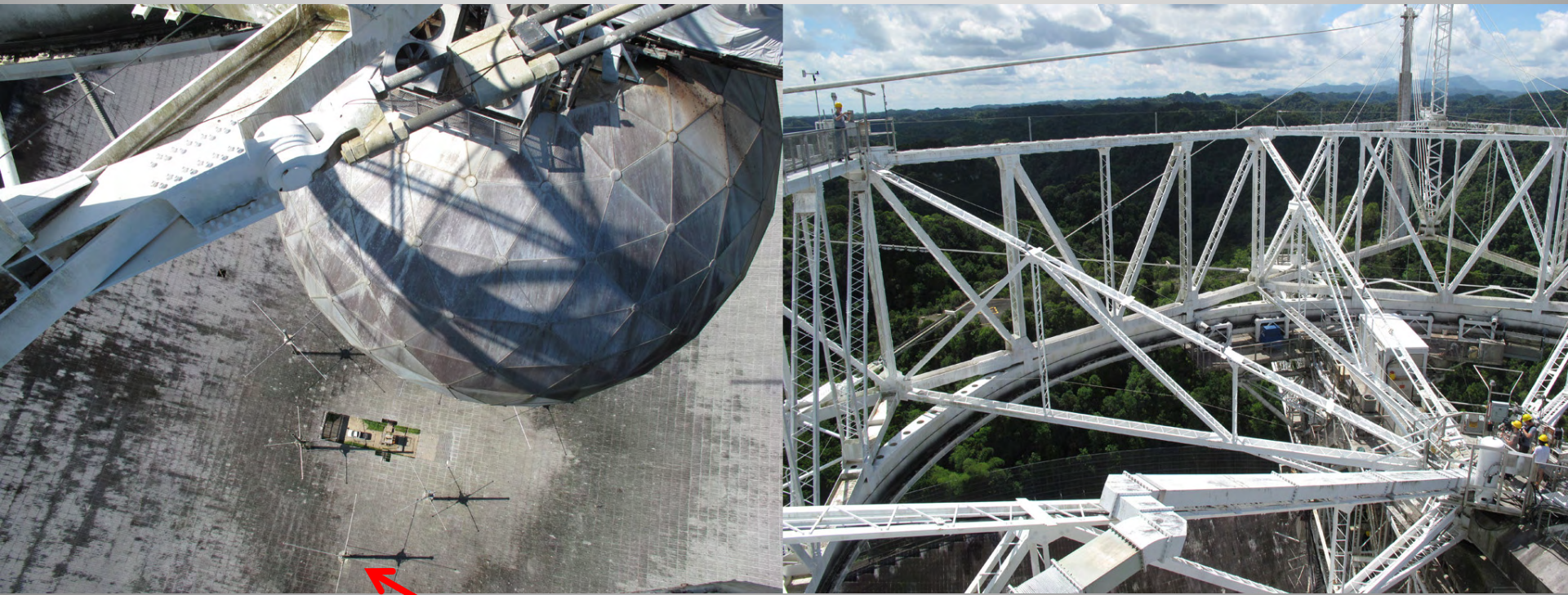




# Arecibo Radio Observatory



# View from above



Ionospheric heaters

# View from above



# On the Structure



Dana Whitlow: microwave RF guru and all-around champ



# Under the Azimuth Arm

Dennis Wingo: our  
fearless leader

Austin Epps:  
engineering brain box

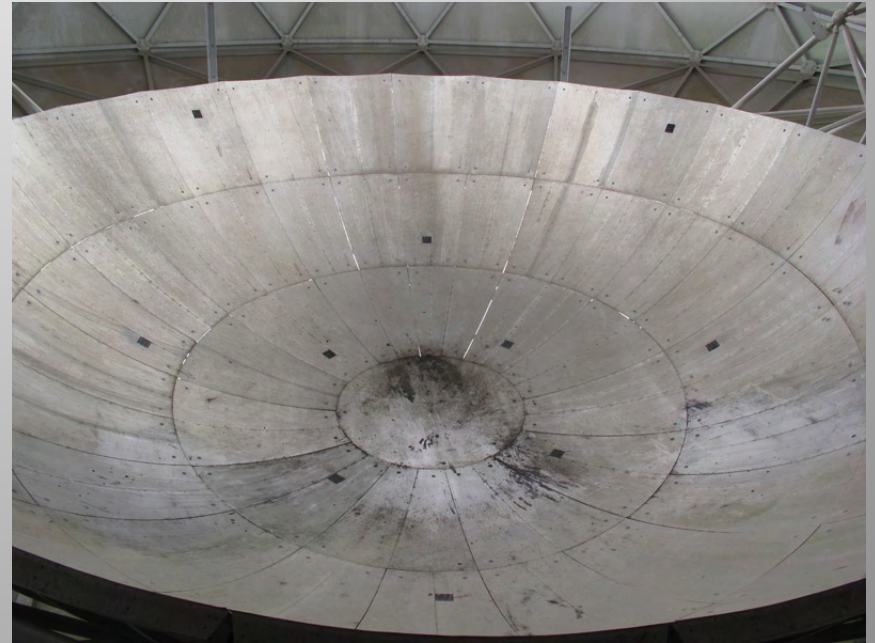
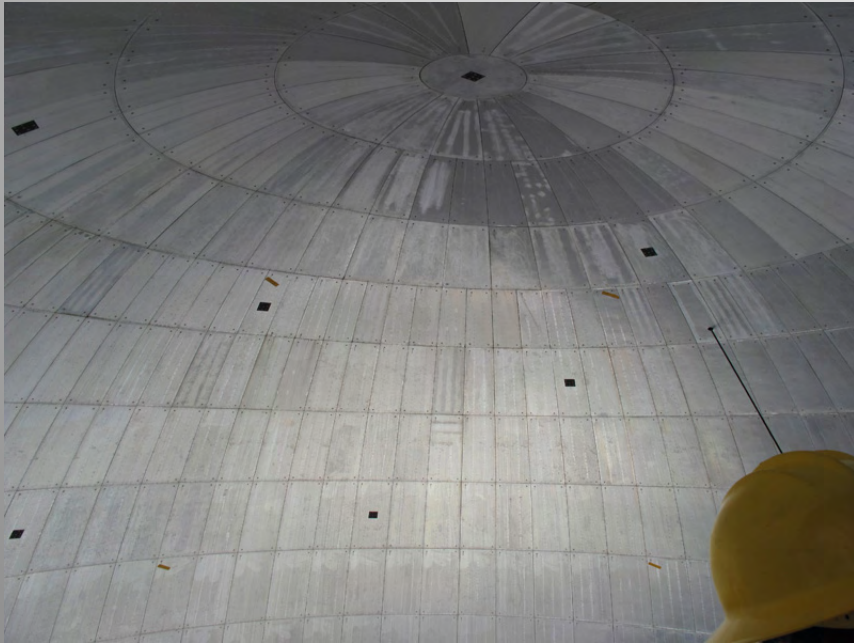


# Cable Car & Underneath the Dome

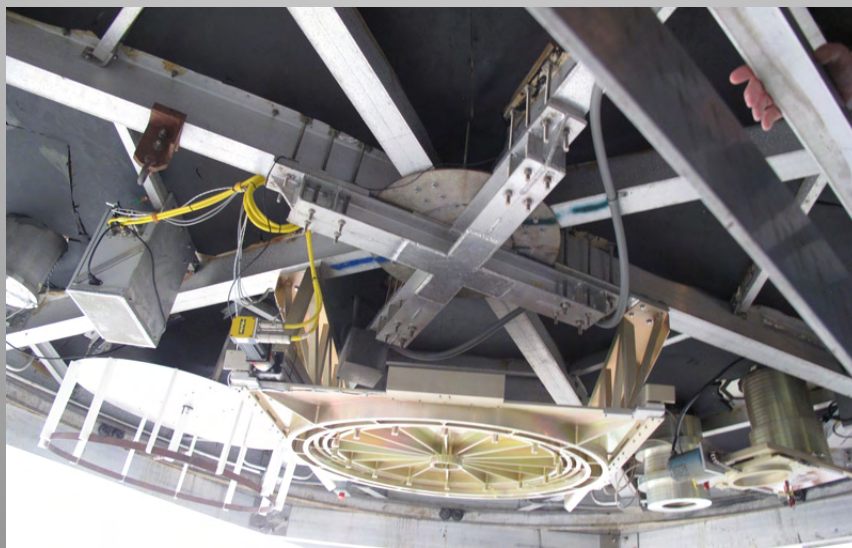


# Dome Reflectors

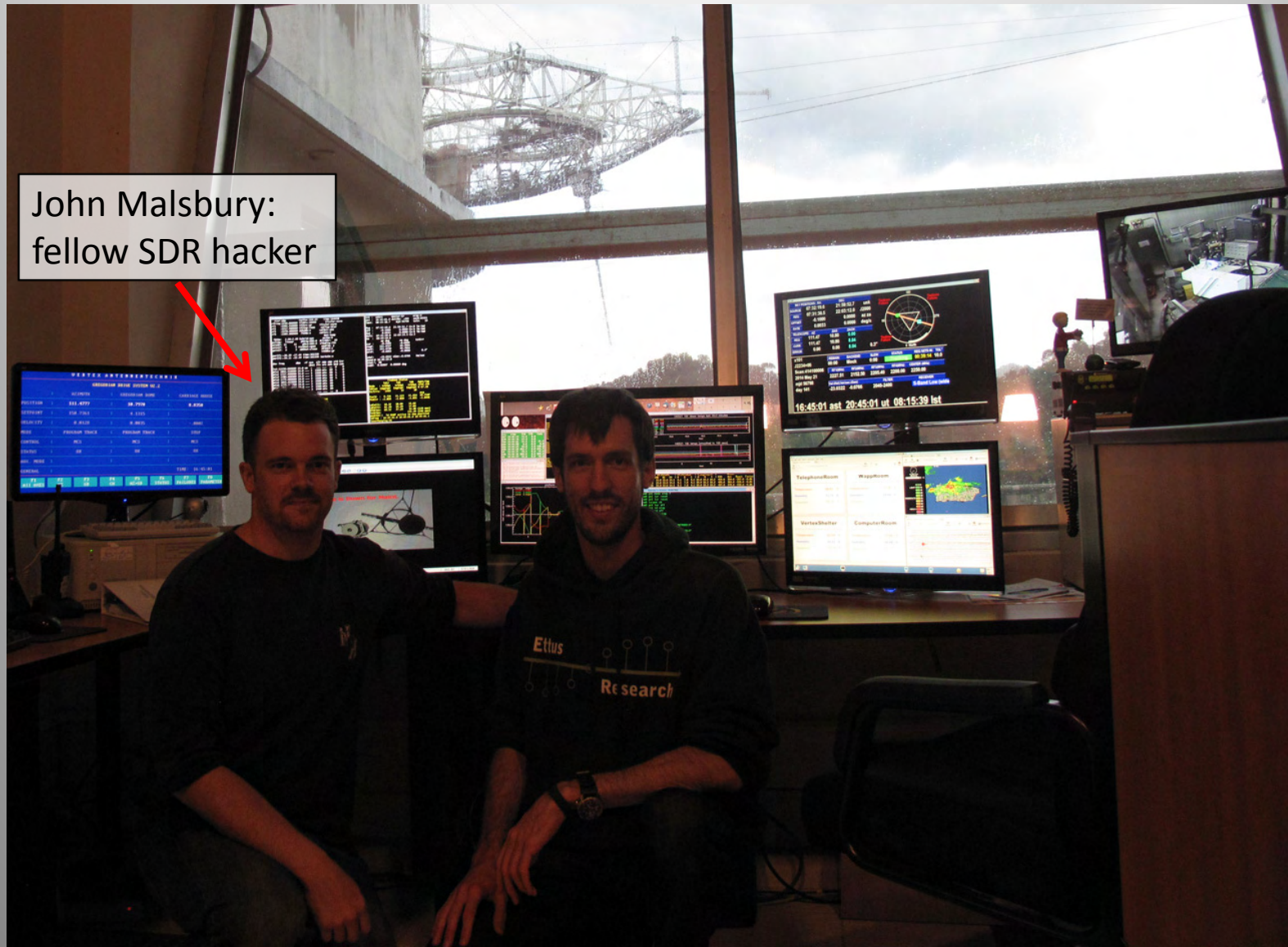
- Secondary & tertiary bring photons to focus



# Feeds with Protective Shutters

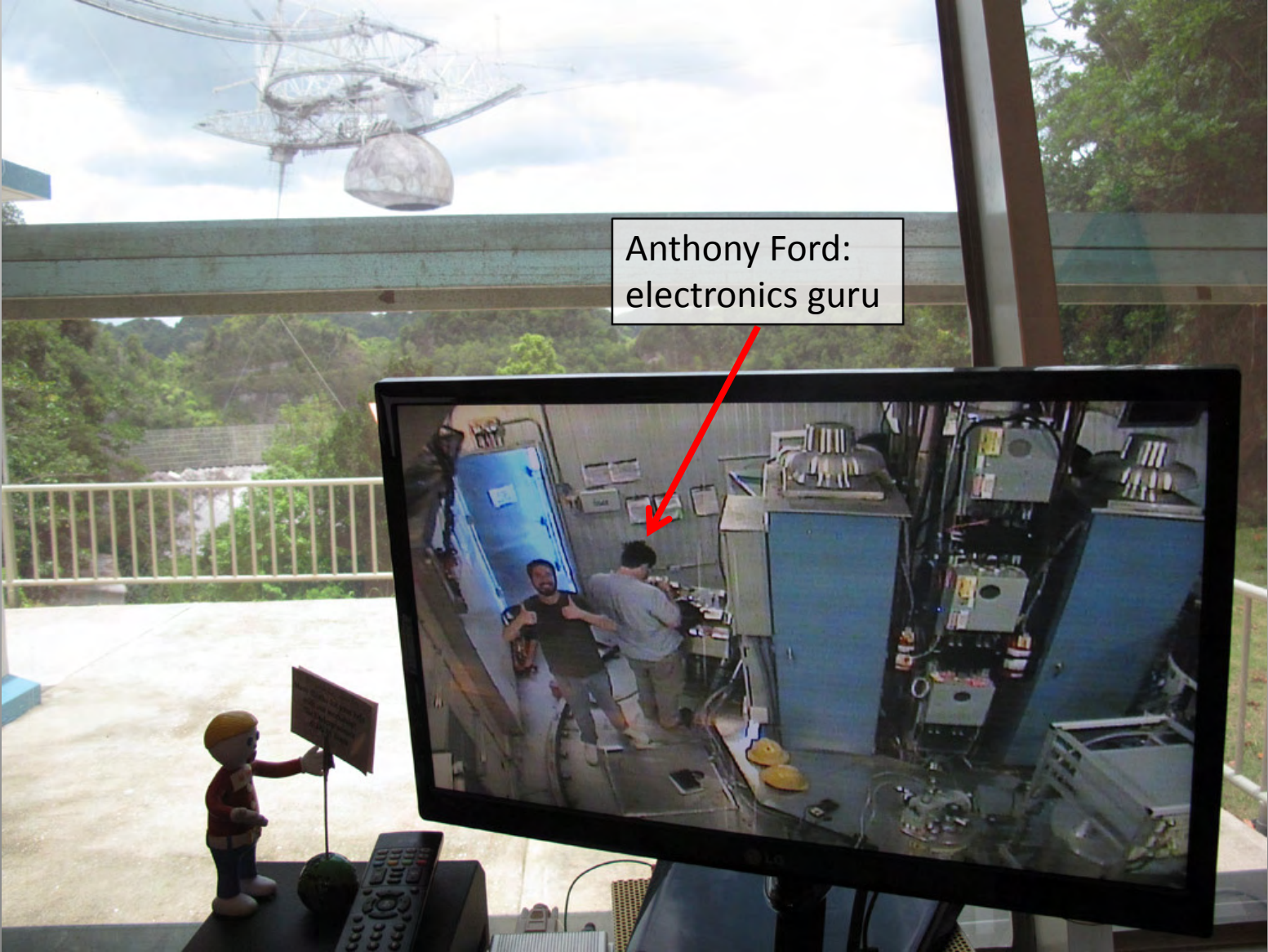


# Telescope Operator's Station





# CCTV on the Receiver Floor



# RFI is a real issue here





# Under the Dish

Victor Negron: mega S-band transmitter wrangler

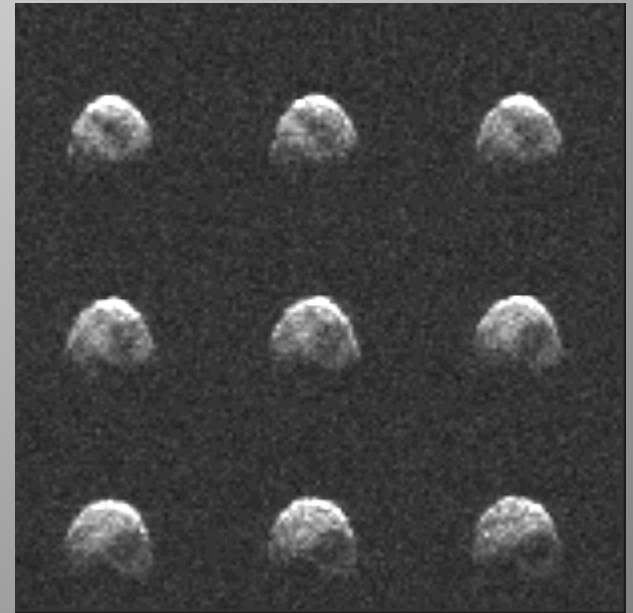






# Arecibo Radio ~~Observatory~~ Transmitter

- Already setup for S-band RADAR
  - Planetary RADAR astronomy (e.g. asteroids)
- Dish gain of  $\sim 73\text{-}75$  dB





# Arecibo Radio ~~Observatory~~ Transmitter

- Already setup for S-band RADAR
  - Planetary RADAR astronomy (e.g. asteroids)
- Dish gain of ~73-75 dB
- Narrow beamwidth (limited steering)
  - Need for accurate ephemeris
  - Signal considerably attenuated in beam sidelobes both for TX & RX (when pointing is not directly on target)

# Arecibo Radio Transceiver

- Use existing S-band wideband (SBW) receiver
  - $T_{\text{sys}} \sim 35\text{-}40\text{ K}$
- Uplink frequency is outside range of existing **dual 1 MW klystron** S-band transmitter
  - Need alternative for TX to contact space probe



# Fun Facts

- S-band RADAR has **20 Terawatt EIRP**
- Unlucky birds that fly through beam:
  - Are boiled
  - Fall onto dish
  - Catch fire
  - Reflect a lot of RF back into the dome  
(bad for receivers)





# New S-band Power Amplifier

- Built by Dirk Fischer (DK2FD) in Germany
  - Shipped overnight
- Consists of four UMTS-inspired amplifiers
  - Outputs combined to provide up to 450W
- Arecibo uses S-band waveguide to propagate signal
- Output of amplifier connected to waveguide transition & uses existing S-band feed



# New S-band Power Amplifier

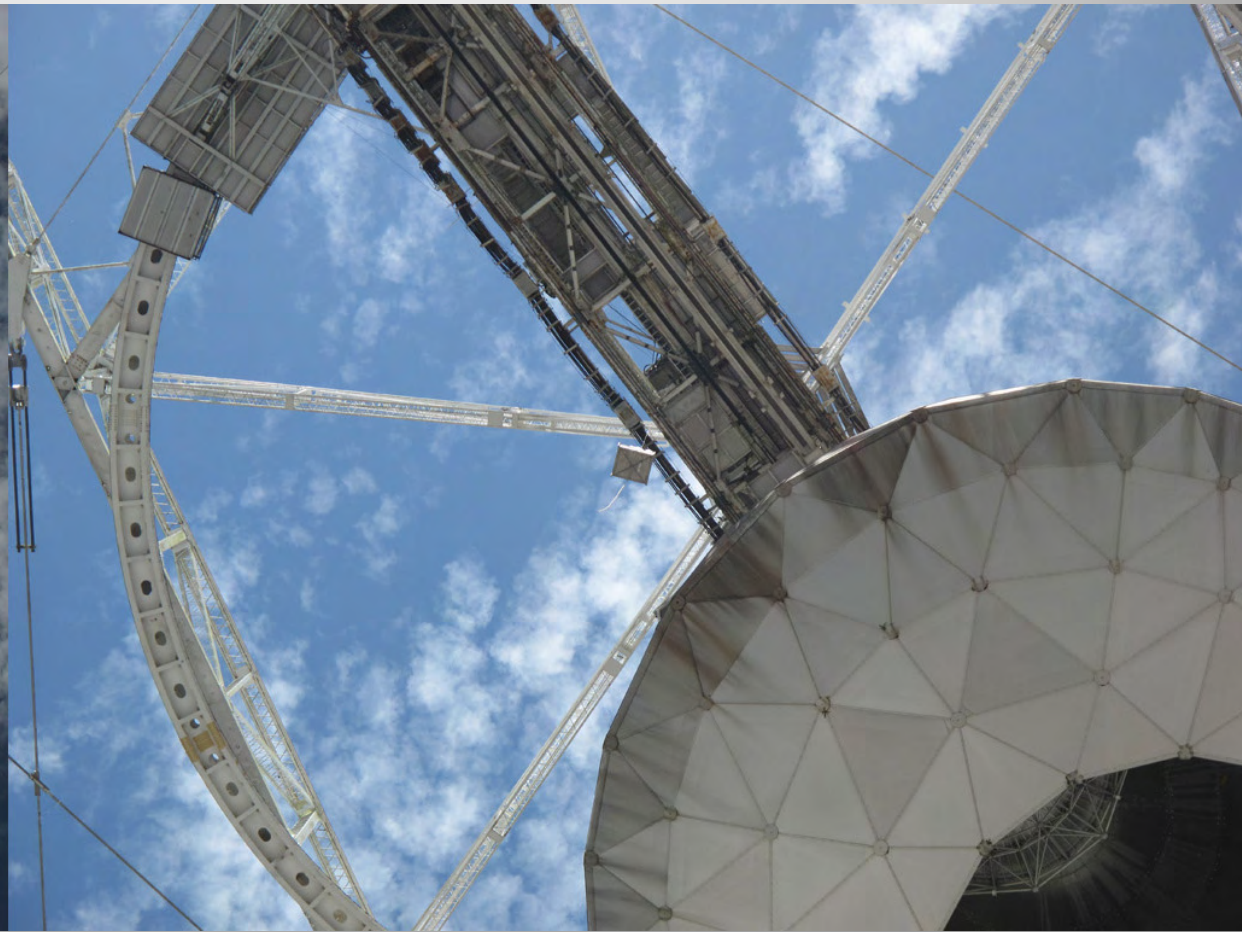


# Hoisting PA to the Dome





# Hoisting PA to the Dome







# Hoisting PA to the Dome





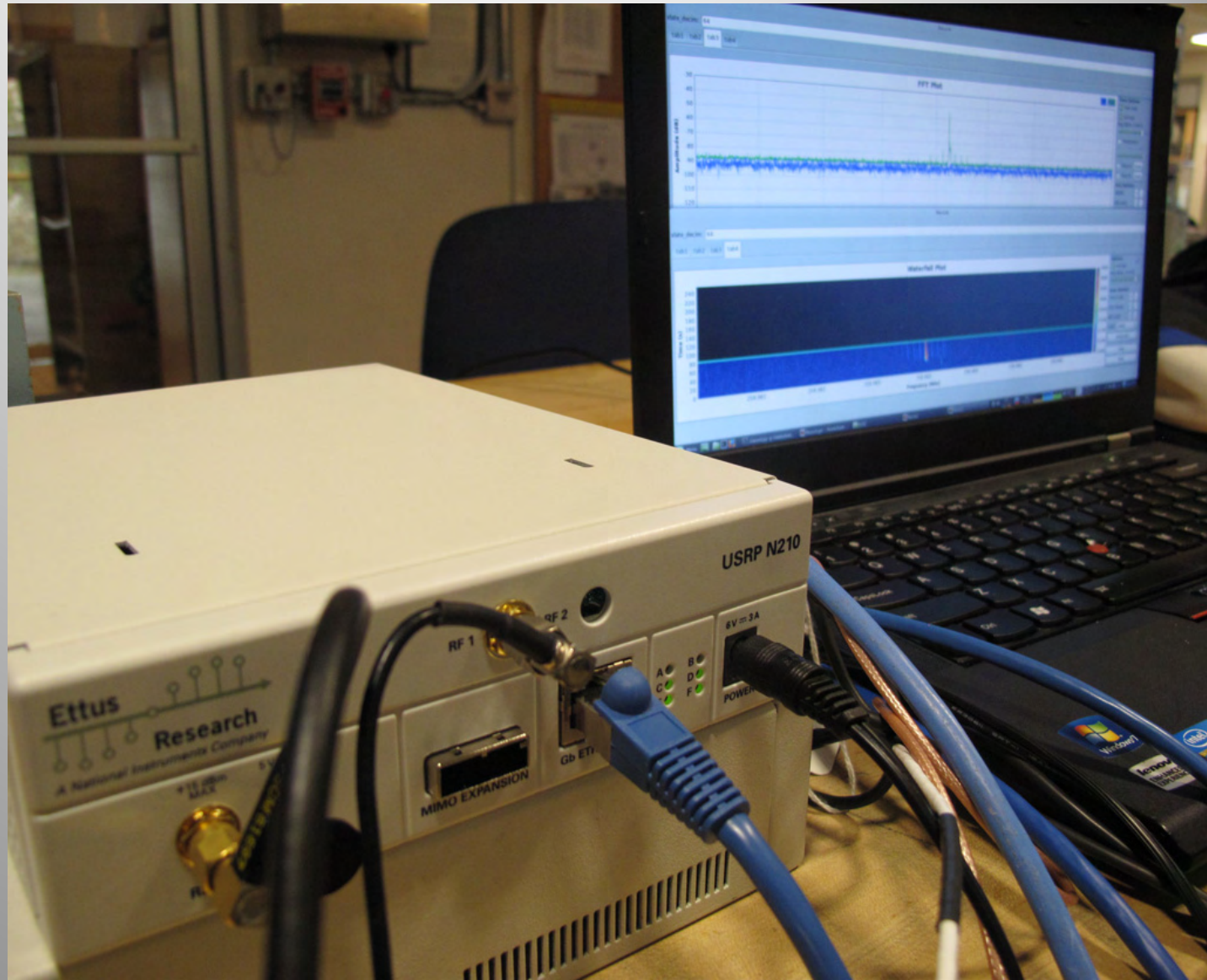
# Observation Time Window

- Total ~2.75 hours each day in the afternoon
  - Space craft passes into arc that Arecibo can track
- Telescope schedule is incredibly busy
- We operate during ‘maintenance’ (unscheduled) periods
- Our actual window varies from day-to-day
- Switching between receiver and transmit takes time (floor rotation, receiver shutter extension)
- Switching between uplink frequencies takes time (PA calibration)



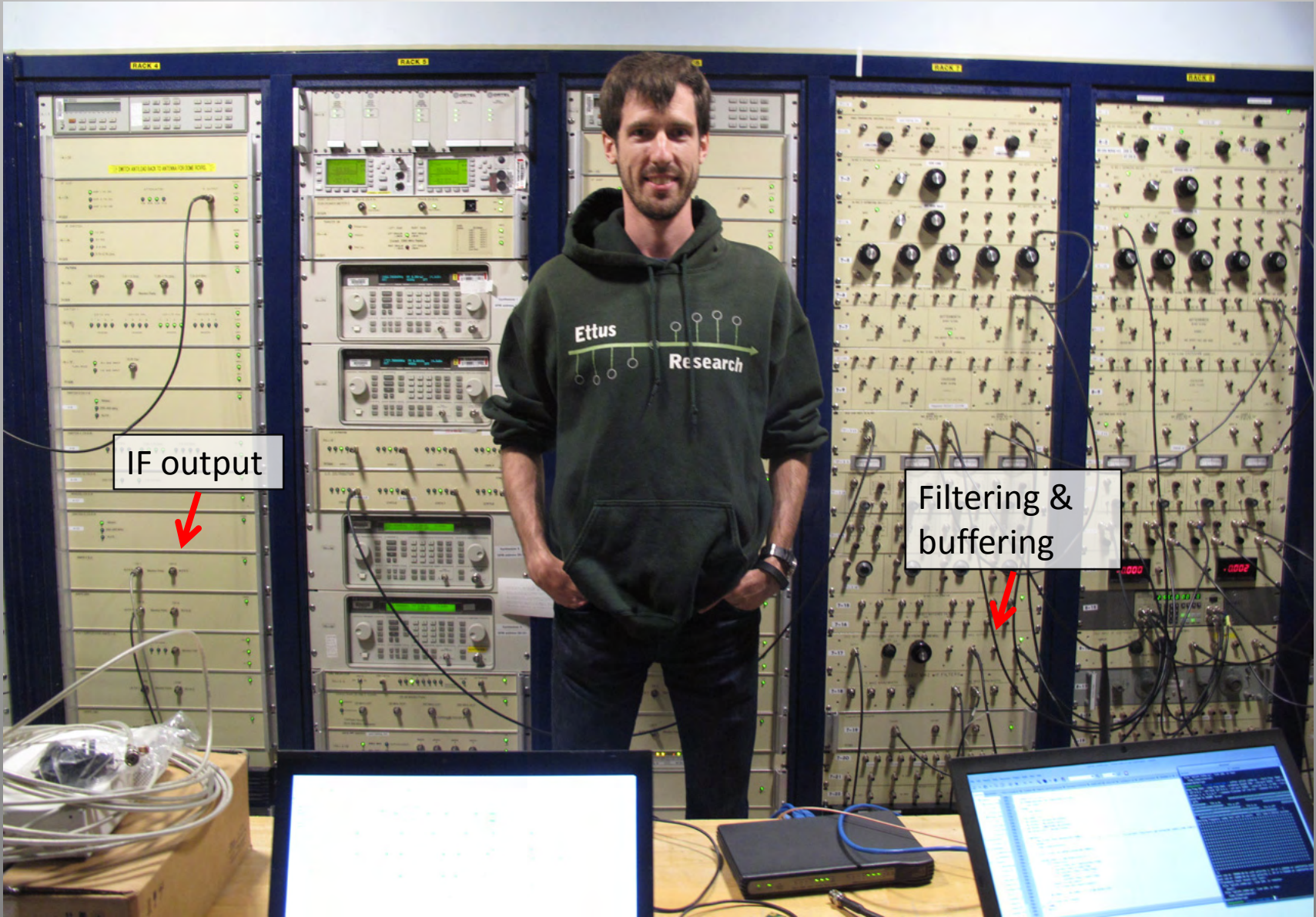
# Receiver

- Two USRPs connected to IF output (260 MHz)
- Can choose between polarisation and downlink frequency





# IF Panel



IF output



Filtering & buffering



DOWNSTAIRS IF-LO SYSTEM  
FUNCTIONAL BLOCK DIAGRAM

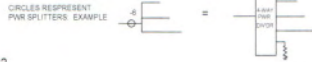
PAGE 1 OF 3: POL. A SIGNAL PATHS  
FIBER & HELIAX INPUTS, POWER  
METER 1, ALFA MONITOR

file: ifdwn1.fct  
rev. APR. 7, 2005 JBH  
rev 9 Feb 2009 DEW  
-last rev 01 Mar 2012 DEW

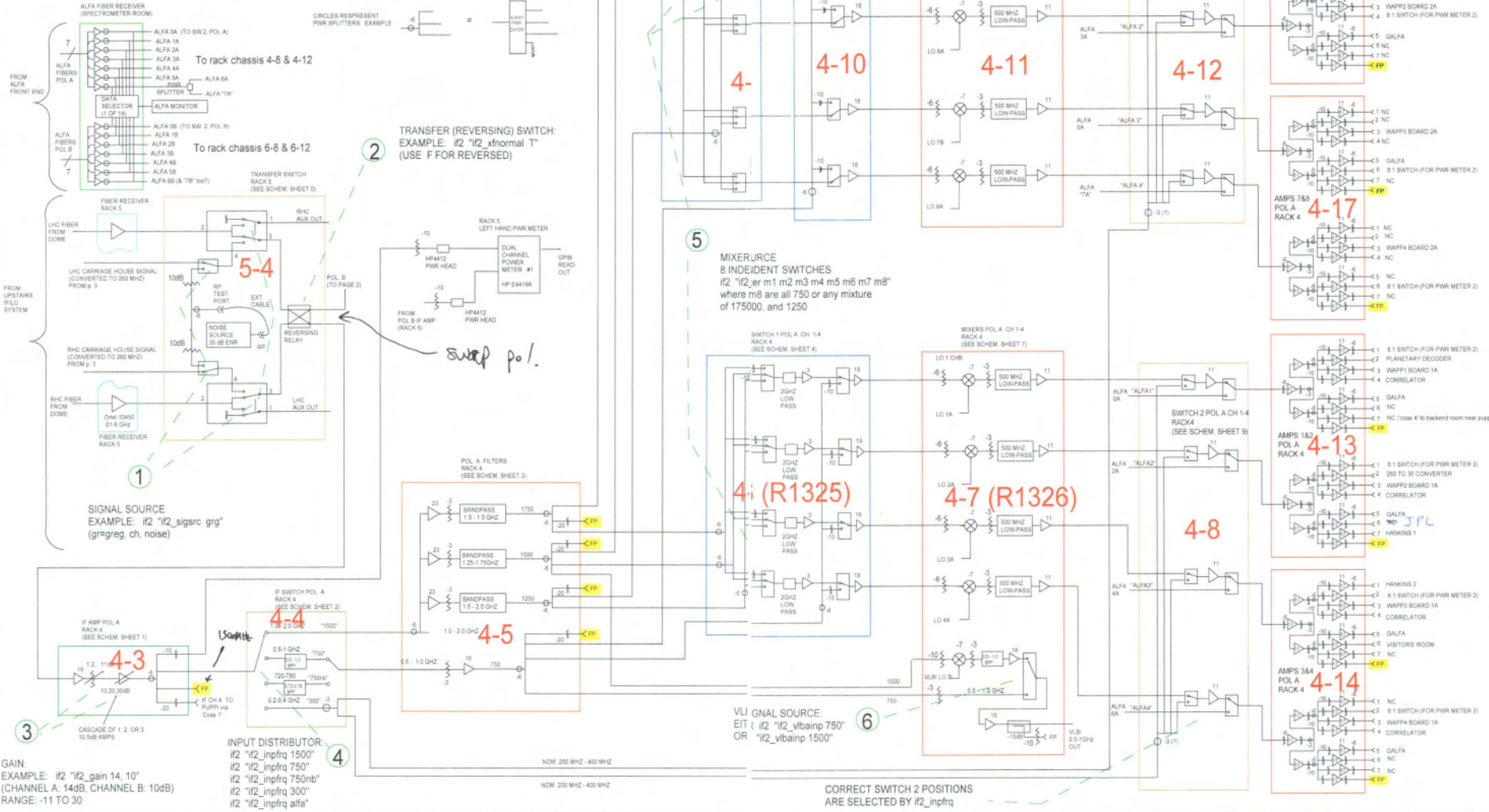
*last rev 11 April 12 DEW*

NOTES

- 1 "FP" = FRONT PANEL
- 2 "RP" = REAR PANEL
- 3 IF AMP IS CASCADE OF 1, 2 OR 3 10dB AMPS
- 4 10.5 dB AMPS COUGAR AC256C, 20dBm OUT @ 1dB COMP
- 5 11dB AMPS COUGAR AC256C, 20dBm OUT @ 1dB COMP
- 6 18dB AMPS COUGAR AC256C, 15.5 dBm OUT @ 1 dB COMP
- 7 23 dB AMPS MCL ZX60-2502M, 18dBm OUT @ 1 dB COMP
- 8 16 dB AMPS MCL ZX60-2514M, 16.5dBm OUT @ 1dB COMP
- 9 13 dB AMPS MCL ZX60-2510M, 15.5dBm OUT @ 1dB COMP
- 10 21 dB AMPS MCL ZFL-5004LN, 17dBm OUT @ 1dB COMP



TRANSFER (REVERSING) SWITCH:  
EXAMPLE: if2 "i12\_ufnormal T"  
(USE F FOR REVERSED)



GAIN  
EXAMPLE: if2 "i12\_gain 14, 10"  
(CHANNEL A: 14dB, CHANNEL B: 10dB)  
RANGE: -11 TO 30

INPUT DISTRIBUTOR  
if2 "i12\_inprq 1500"  
if2 "i12\_inprq 750"  
if2 "i12\_inprq 750nb"  
if2 "i12\_inprq 300"  
if2 "i12\_inprq alfa"

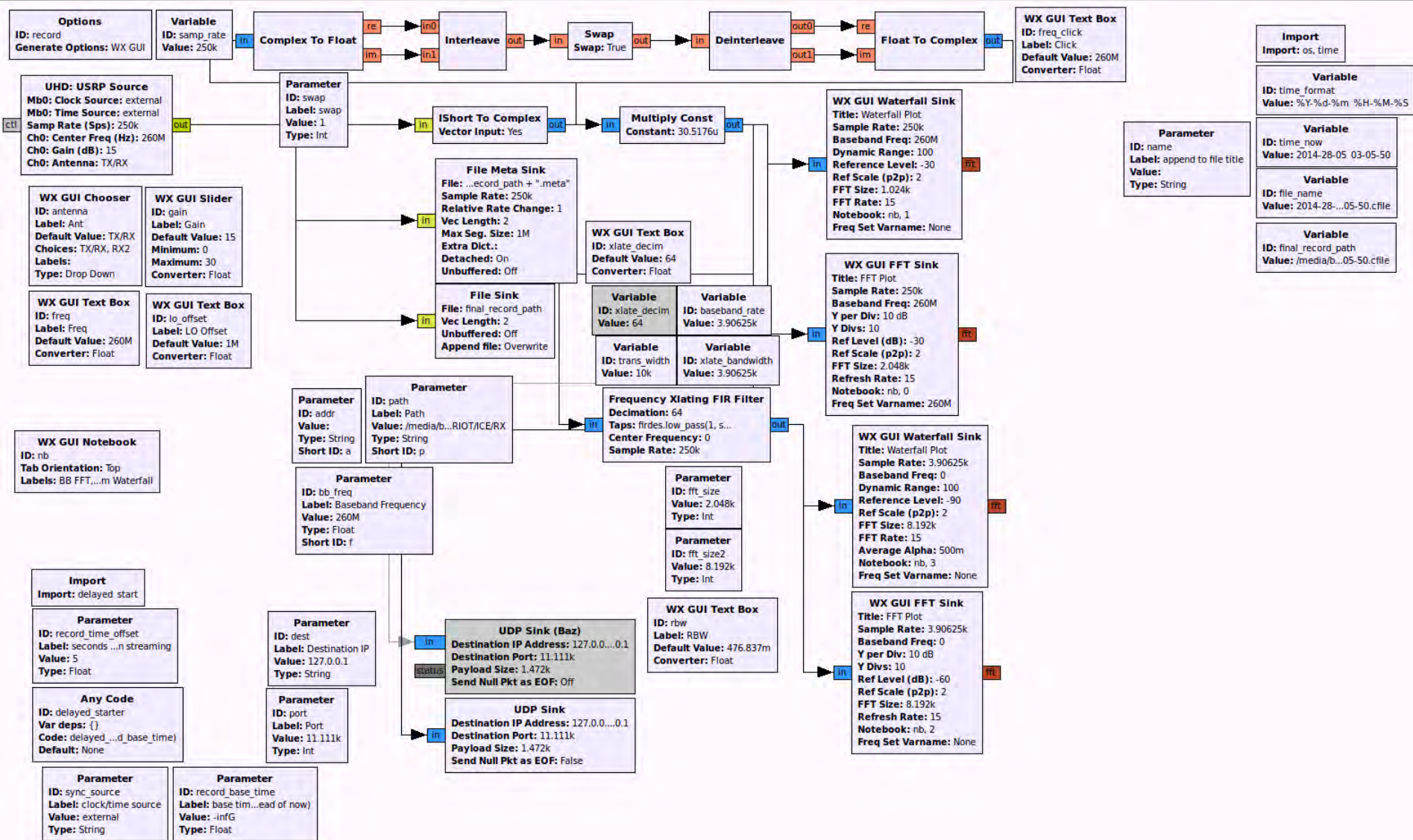
CORRECT SWITCH 2 POSITIONS  
ARE SELECTED BY if2\_inprq

# Synchronisation

- Not your average 10 MHz/1 PPS
- Hydrogen Maser station reference
- Connected to each USRP

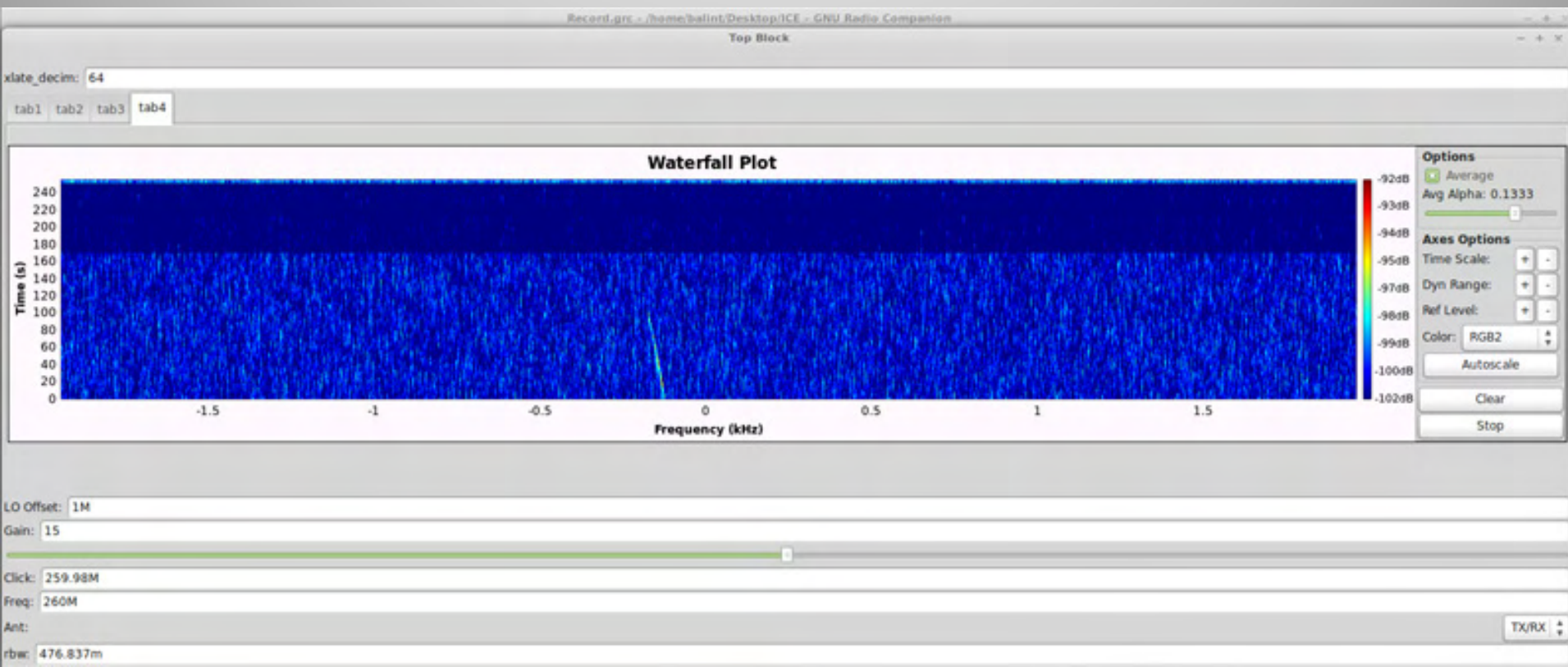


# Display & Recording Flowgraph



# Initial Transponder (Carrier) Detection

- C/N was lower than expected
- Ephemeris is stale (out-of-date)





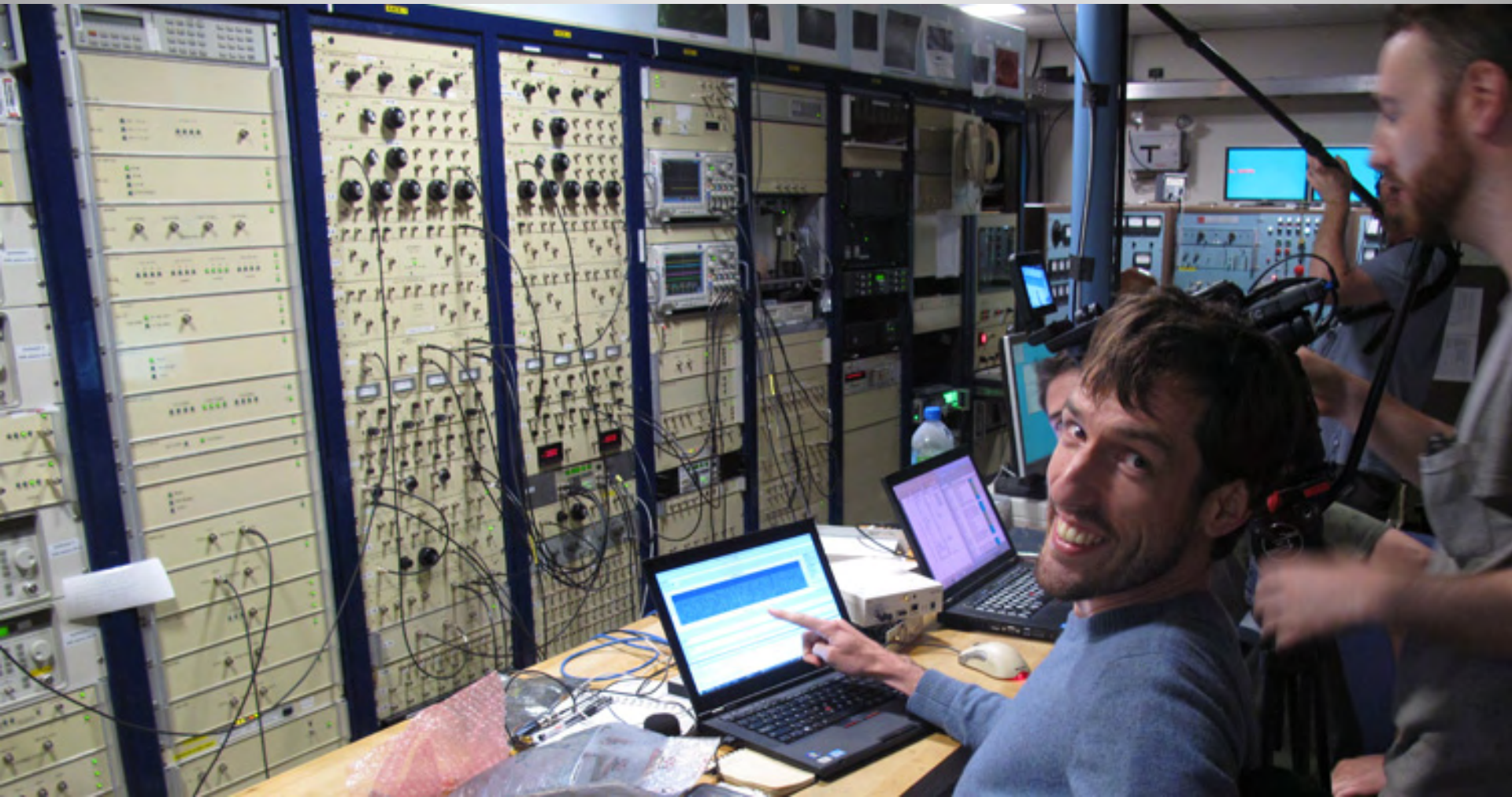




# Stale Ephemeris

- Spacecraft has deviated from predicted path
  - Signal is being attenuated in beam's sidelobes
  - Requires manual sky search to find peak carrier signal
  - Done by one of the local gurus here (Phil Perillat):
    - Watching carrier peak move while tweaking pointing (quick)
    - Doing an automatic search around predicted point to maximise peak (longer)

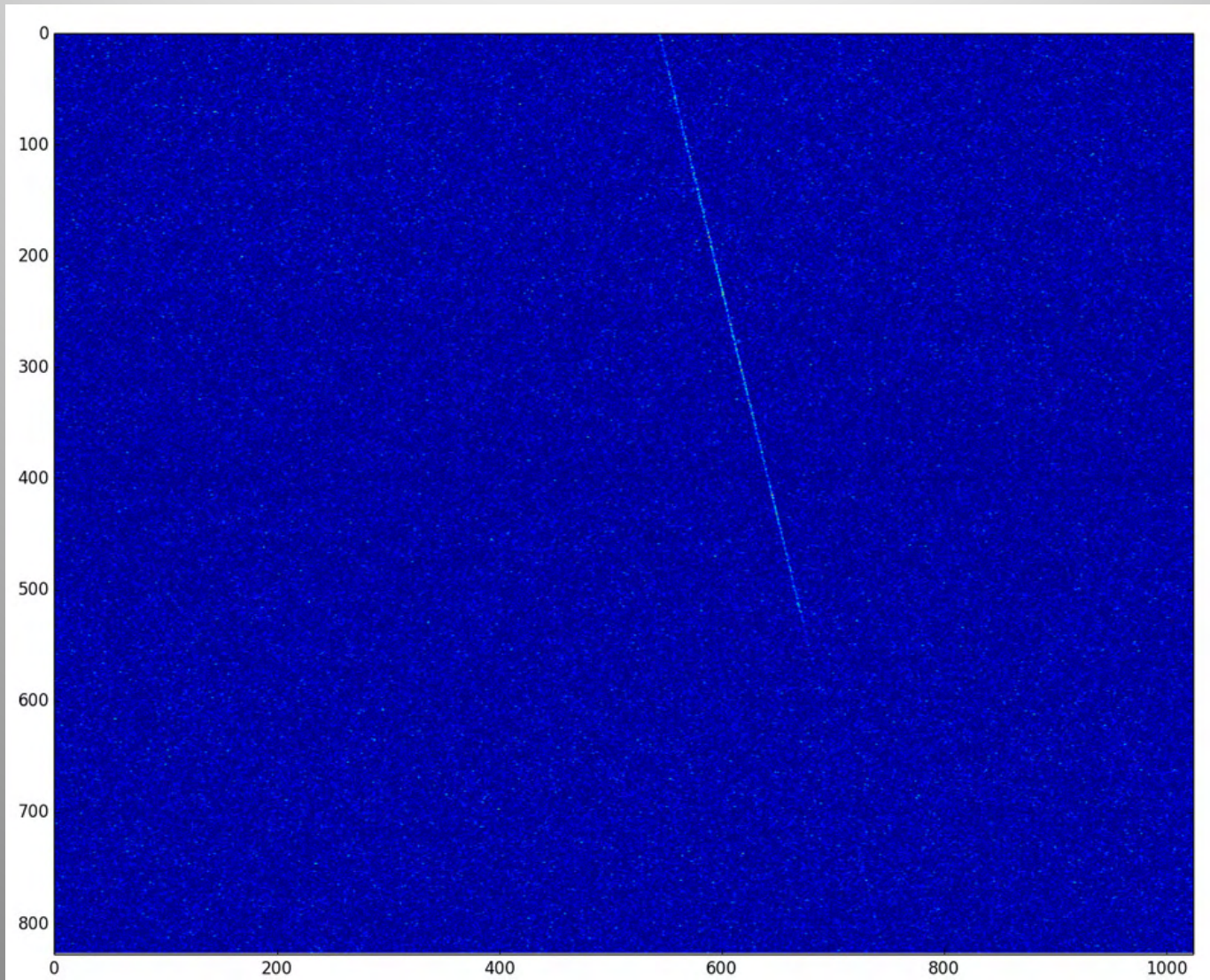
Still a good start...



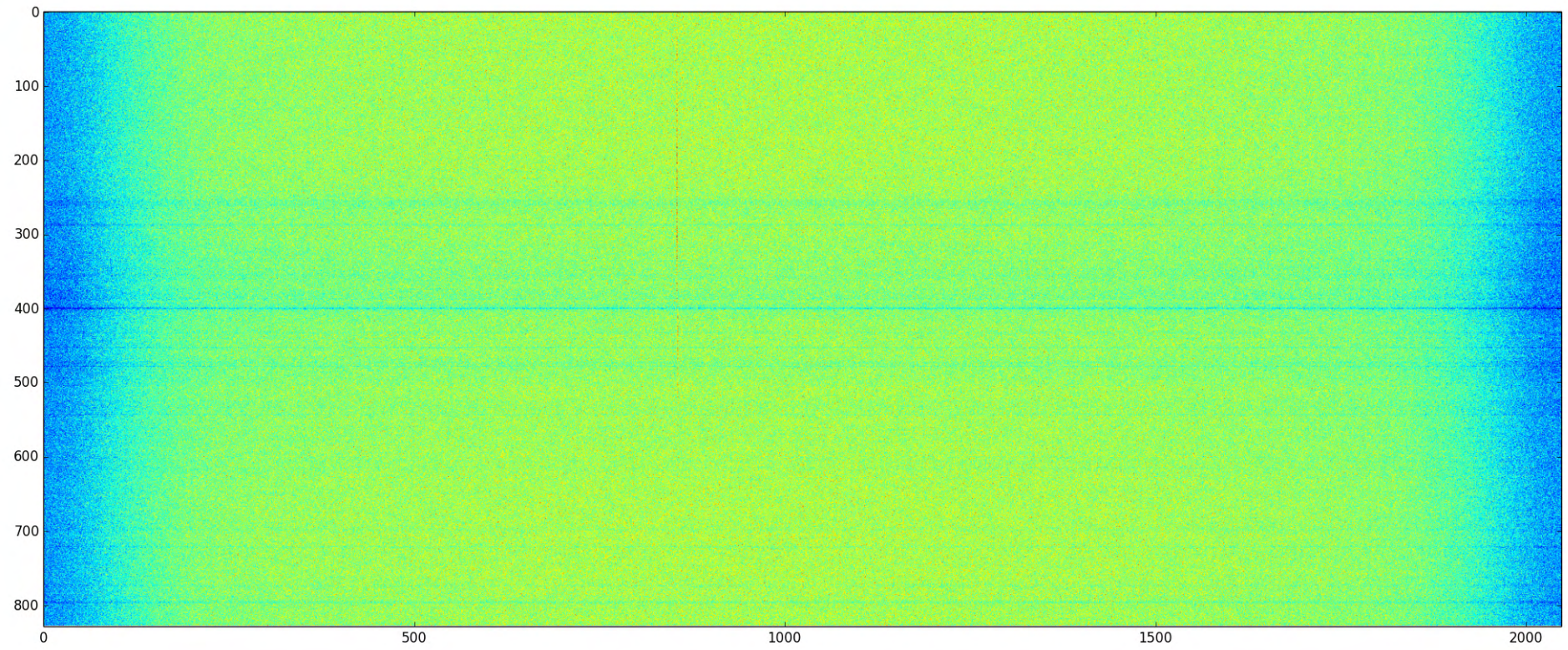
Still a good start...



# numpy & matplotlib

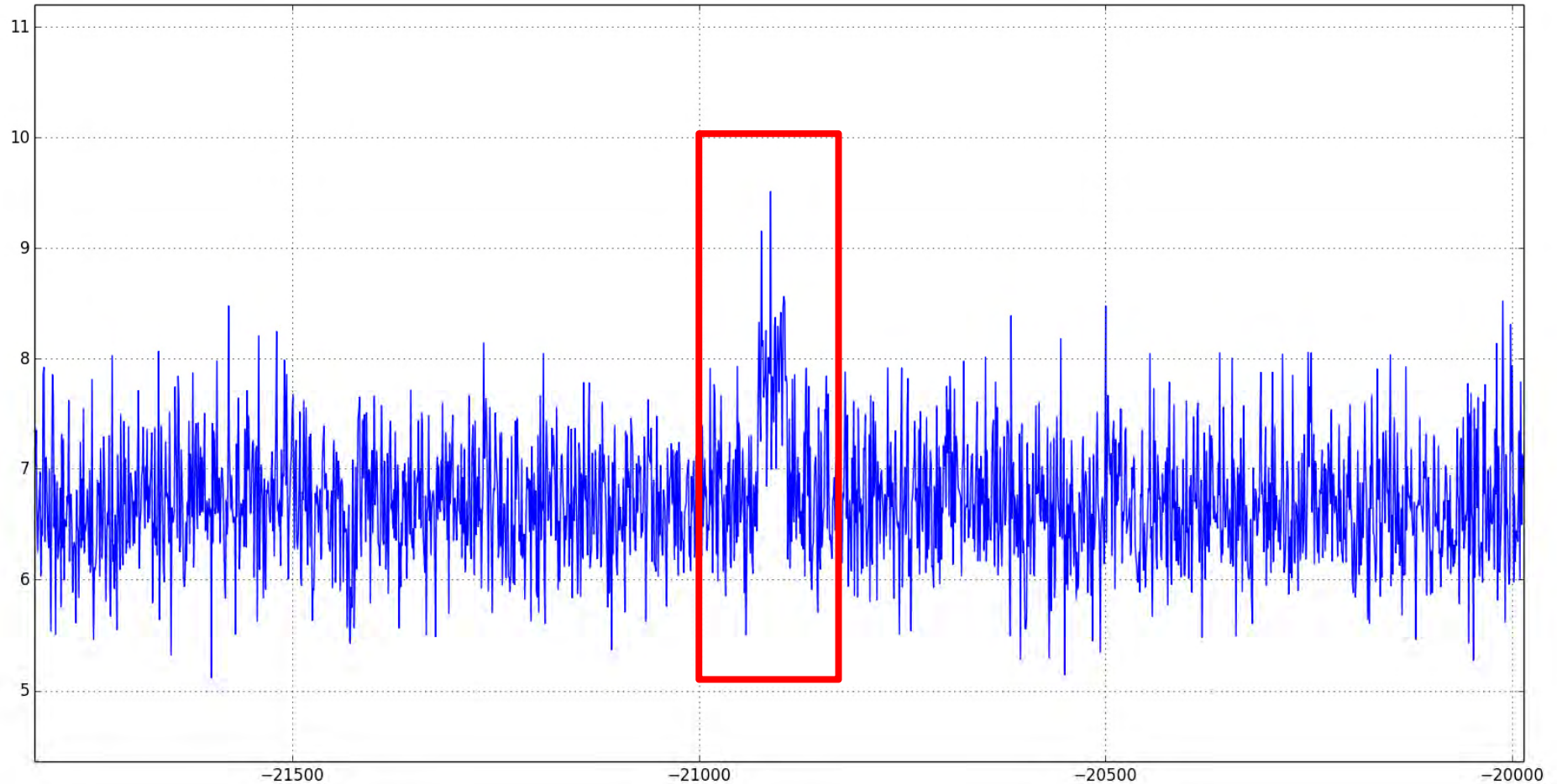


# numpy & matplotlib



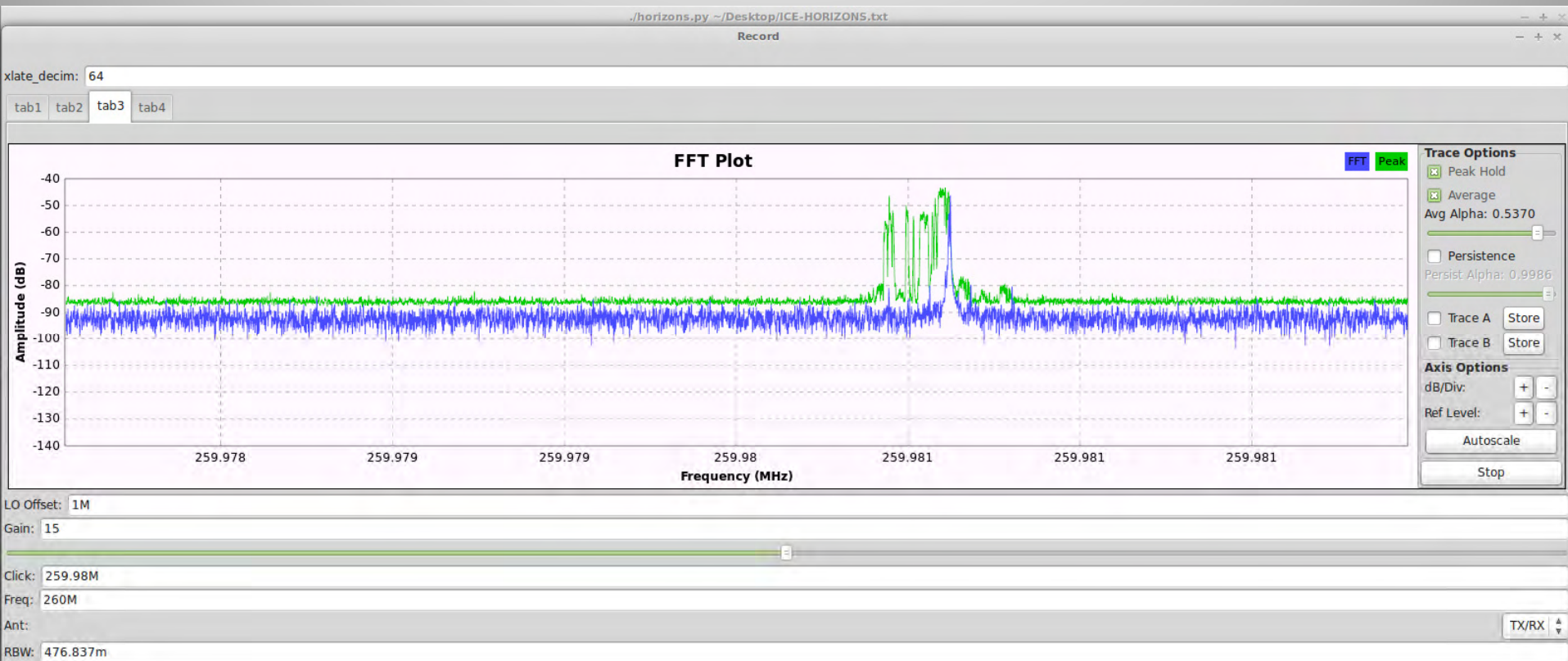


# Weak Signal $\rightarrow$ Low RBW



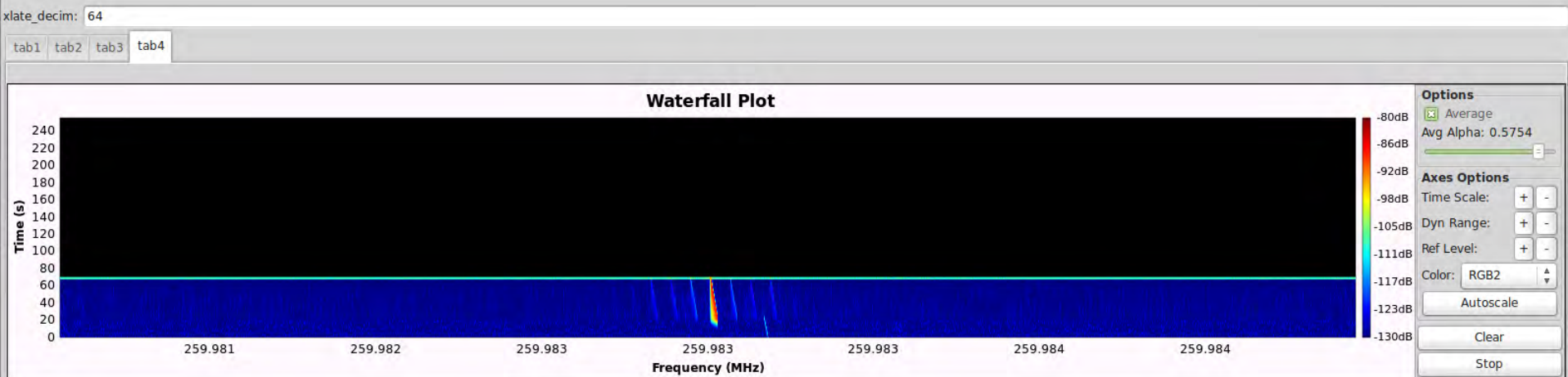
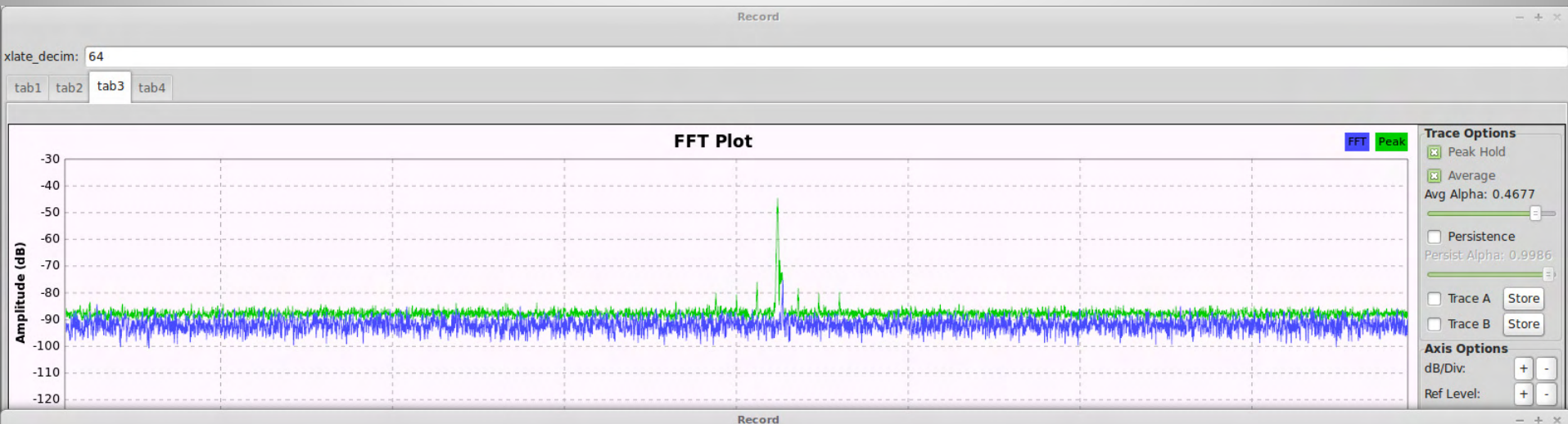
# After Improving Pointing

- ~45 dB C/N
- Moving peak below due to Doppler shift

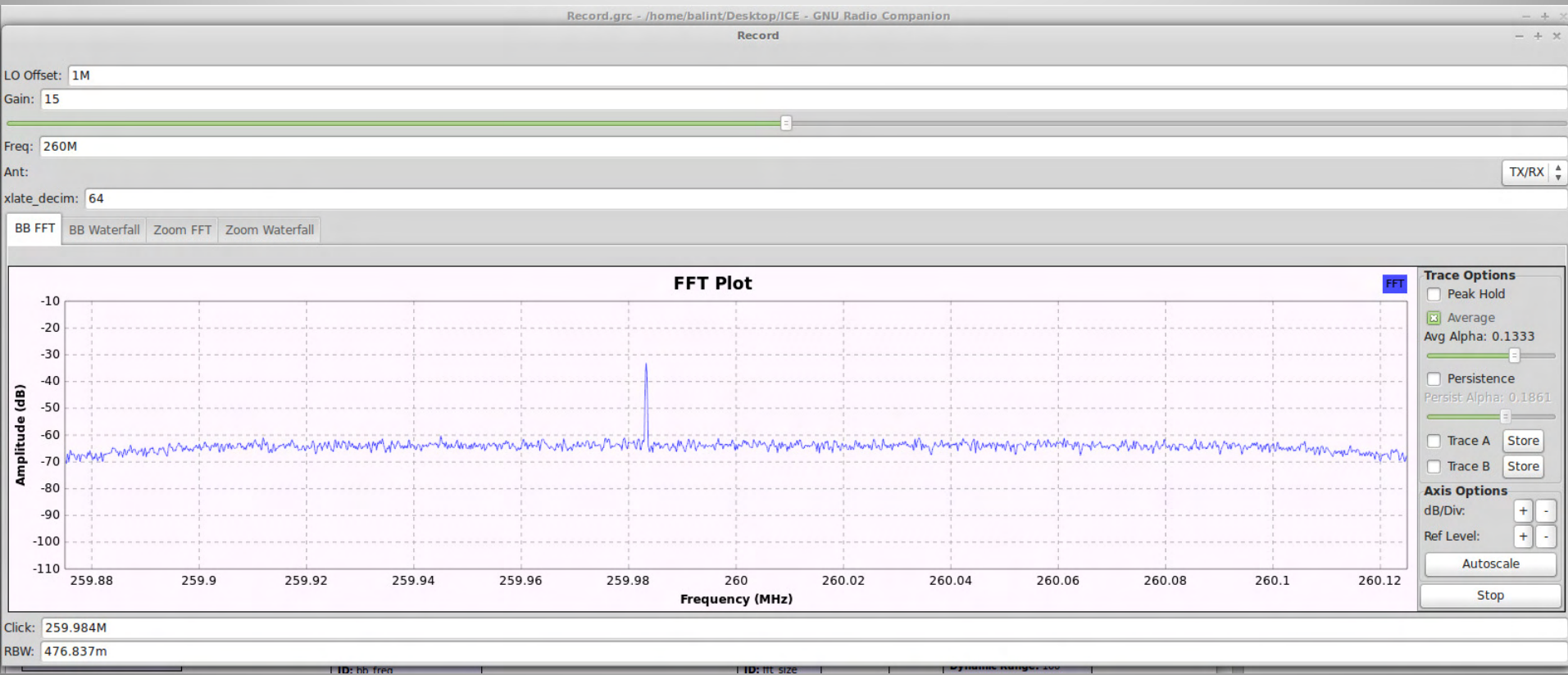




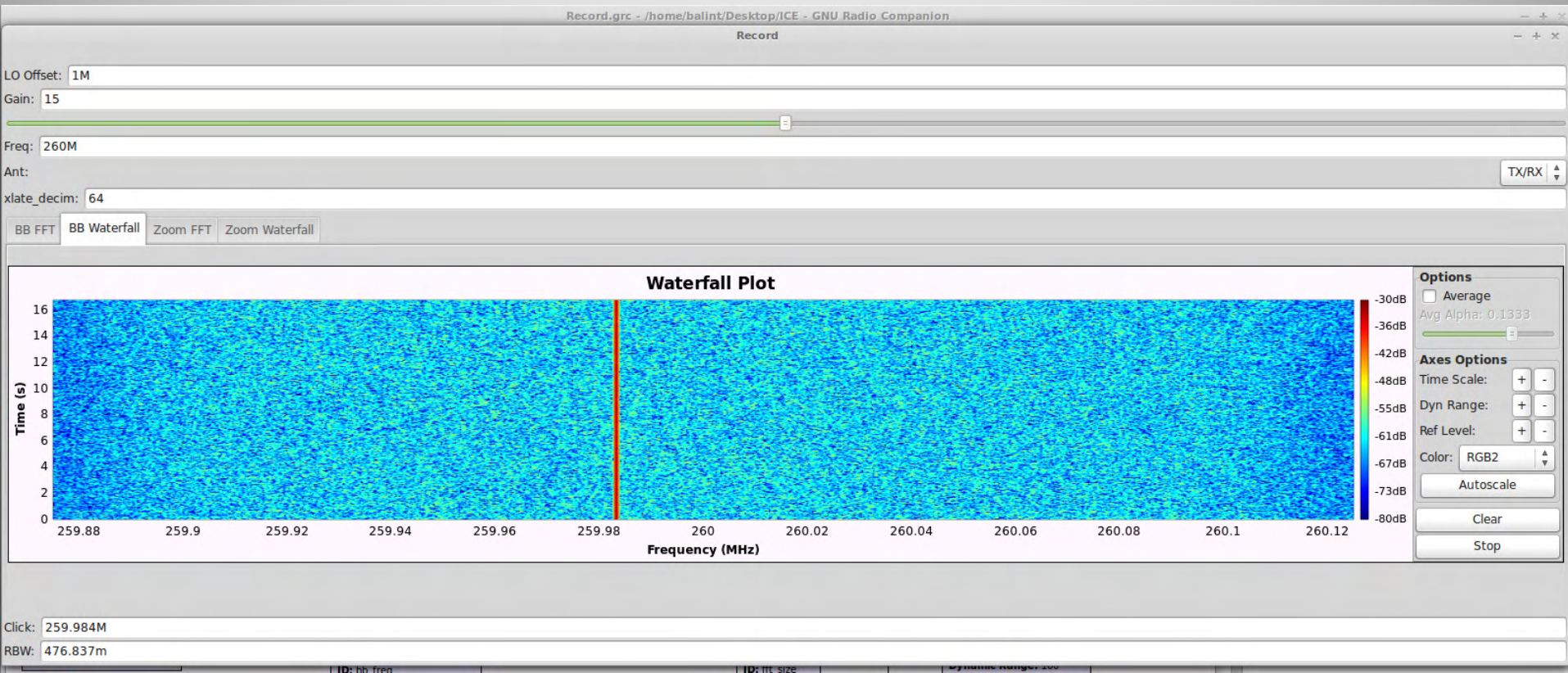
# Dual Receive



# Live Sampled Baseband

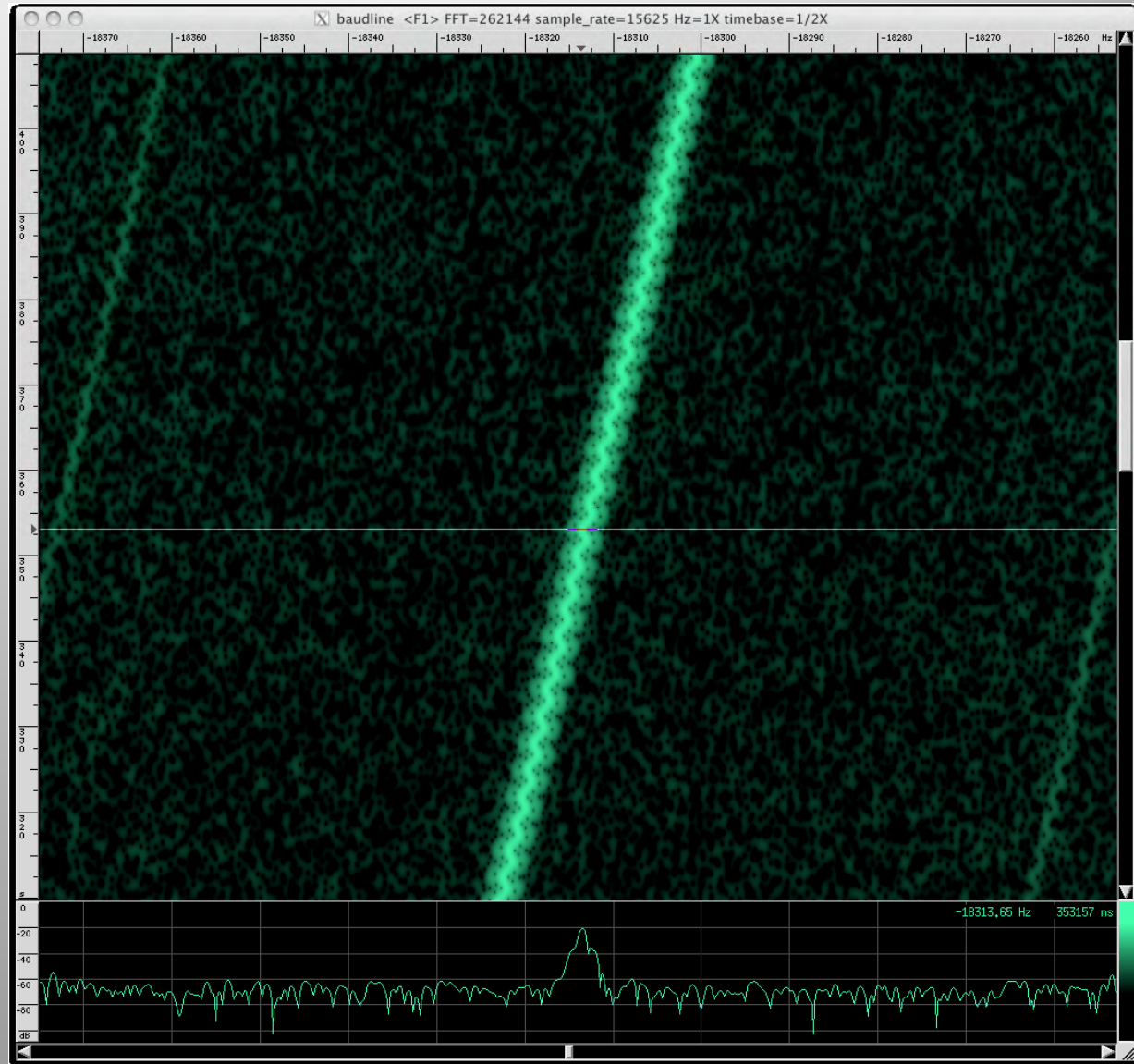


# Live Sampled Baseband



# Spin

- As space probe is spinning, so is antenna
- This can be seen at an extremely small RBW
- Image courtesy of Erik Olson (creator of 'baudline')



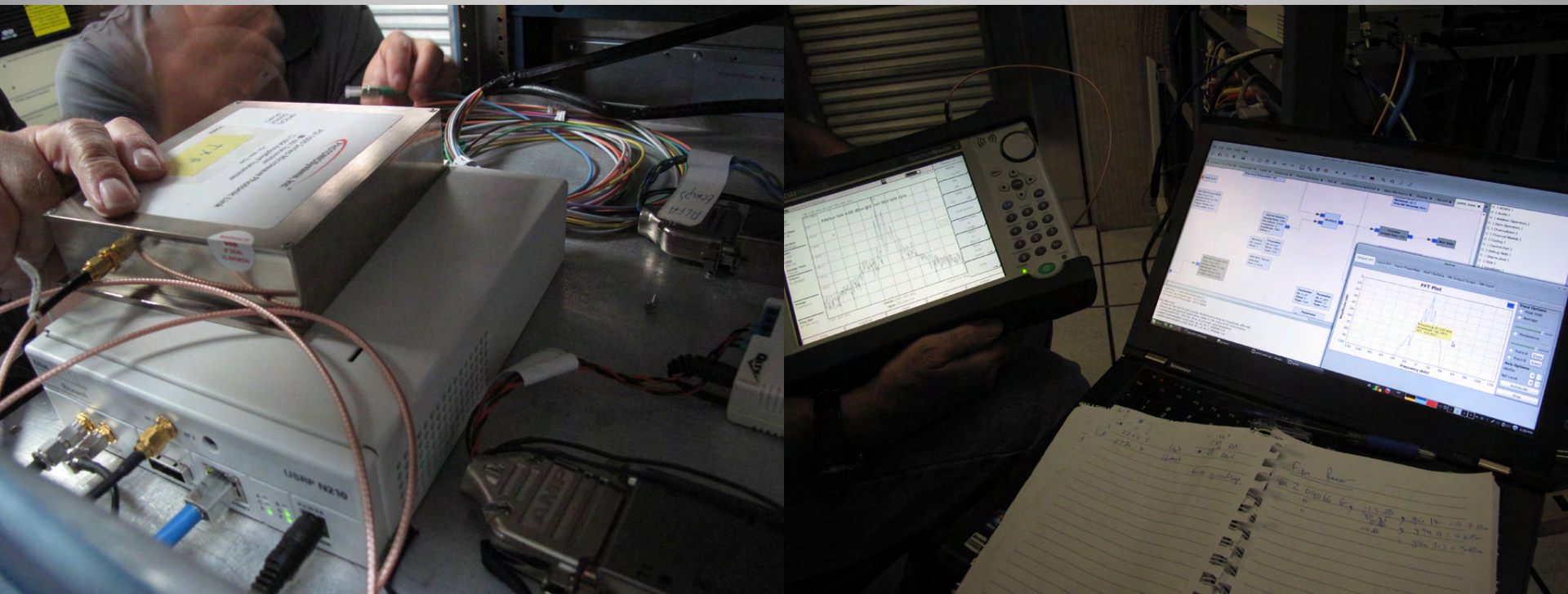
# Media

- Wide-spread coverage
- #ISEE3, @ISEE3Reboot, @EttusResearch, @spenchnet
- This particular photo has appeared on The Register, CBS News...



# Preparing for Uplink

- Dedicated TX USRP
- SBX daughterboard outputs RF into fiber link
  - RF-over-fiber sent to dome (to pre-amps & PA)

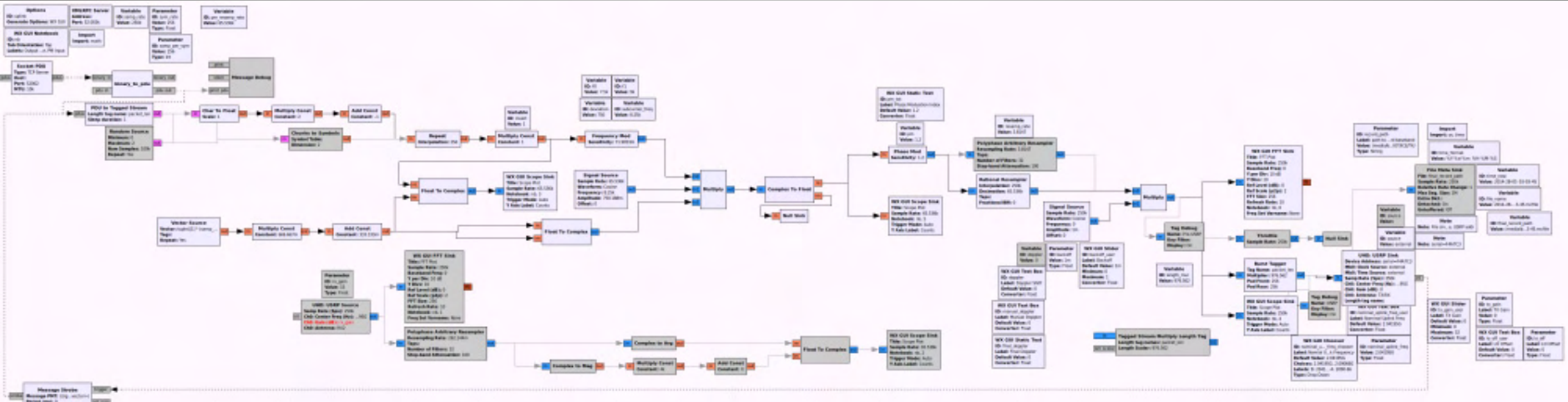


# Preparing for Uplink



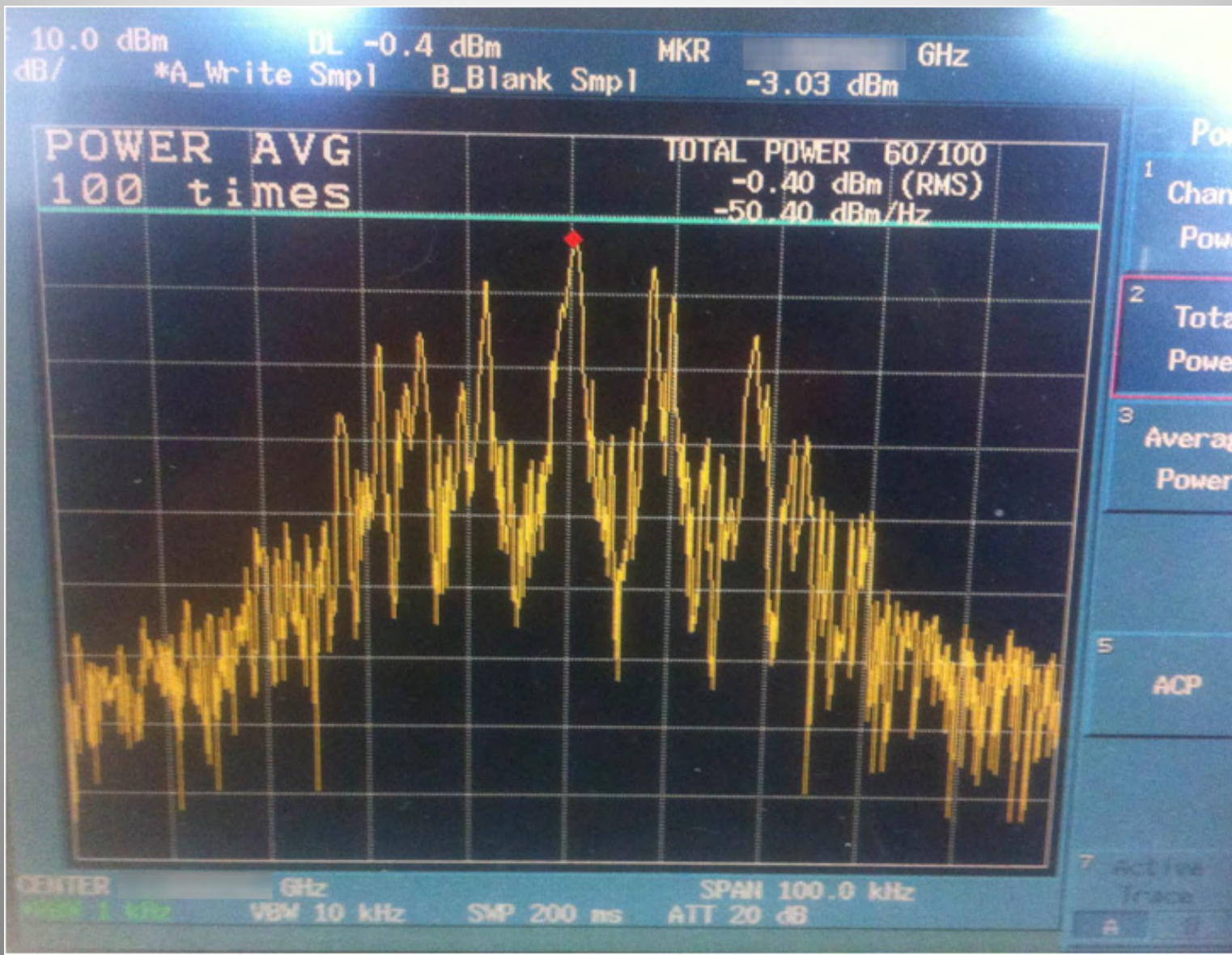
# Uplink GNU Radio App

- Created with John Malsbury (@\_jmalsbury)
- Command bit strings sent via Socket PDU
- Parameters via XML-RPC





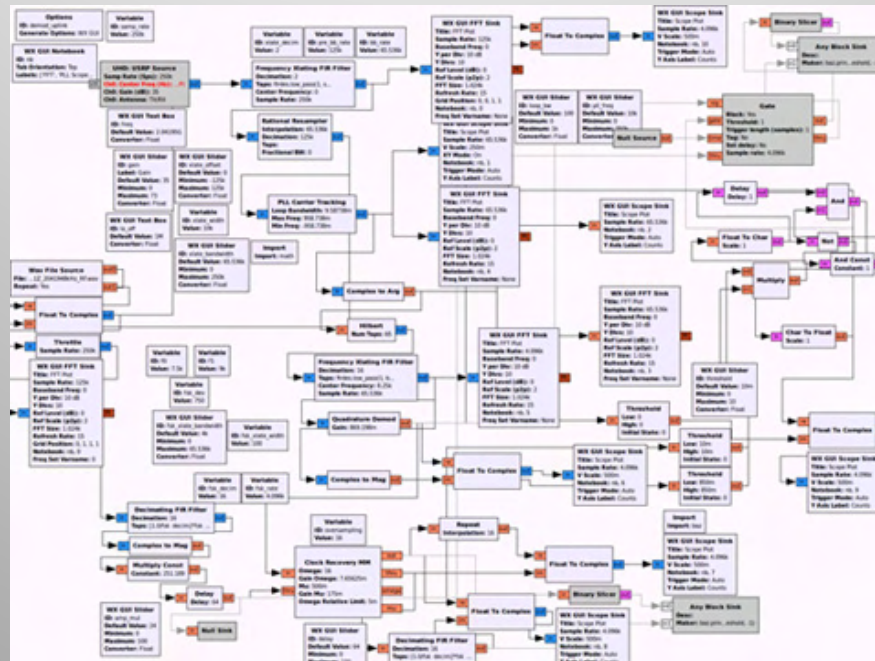
# Previous Test at Ettus HQ (NISV)



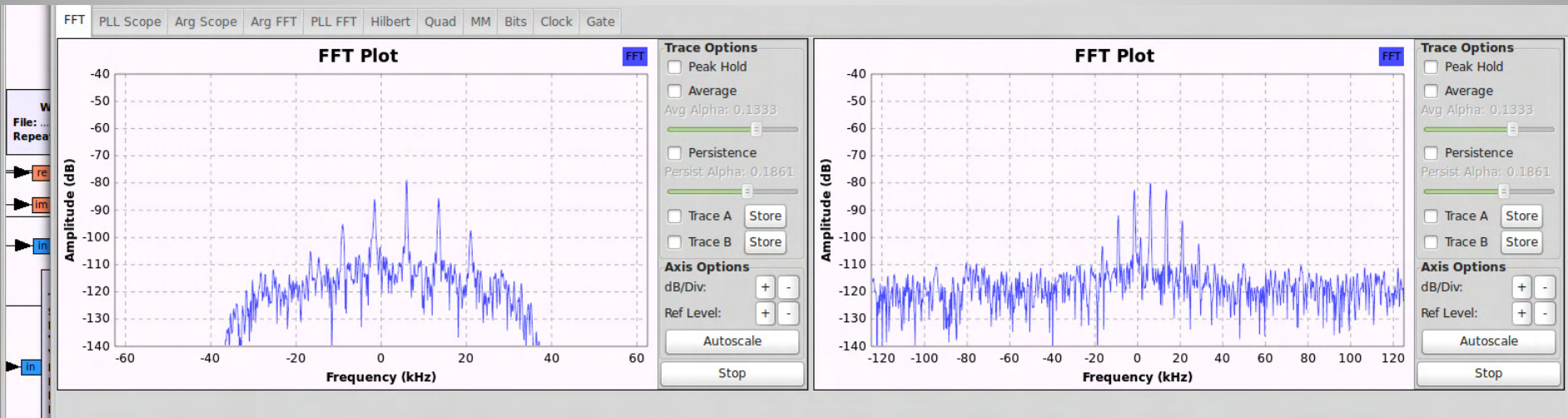
# Fingers Crossed

- Some ambiguity remains
  - Not all modulation/protocol details completely described in documents
  - Run through possible permutations of parameters to ensure all combinations
    - Double-checking that commands cannot misinterpreted (e.g. fire thrusters instead of enabling telemetry)
- Created a test uplink demodulator
  - Identical bits decoded
  - Assuming same mistake is not on both sides

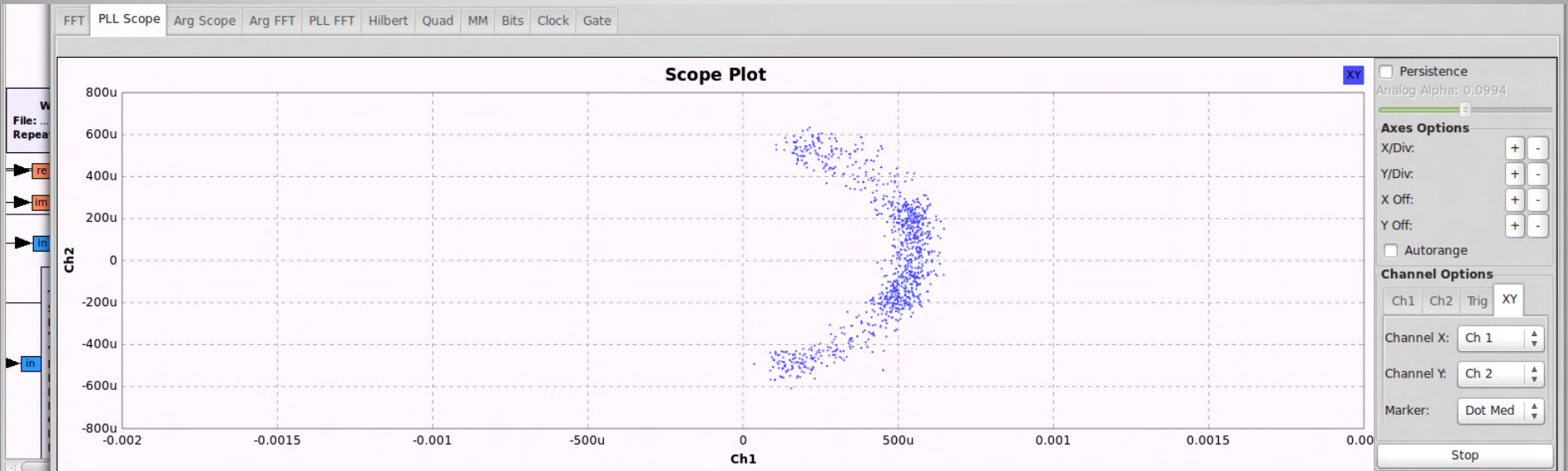
# Verifying Transmitted (Test) Data



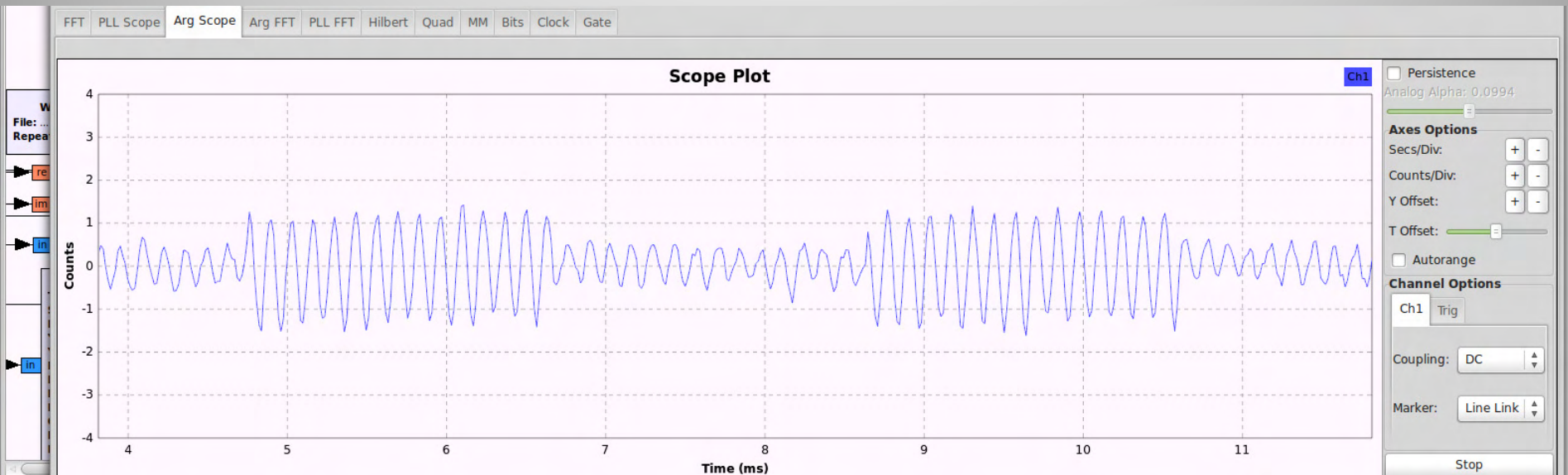
# Received Test Uplink Signal



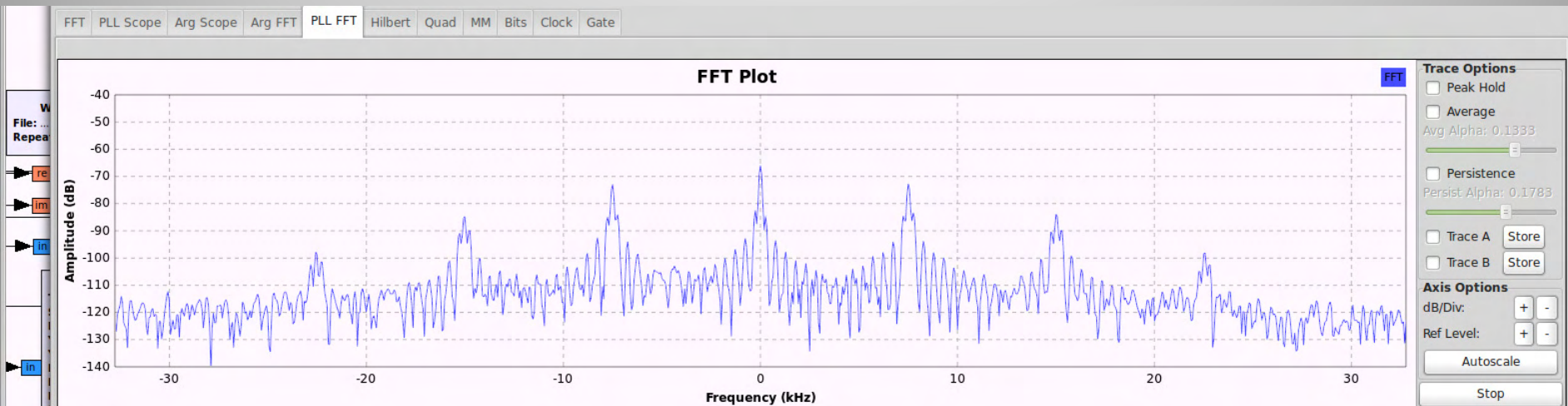
# IQ Scope Plot



# Phase Demodulated Signal



# Phase Demodulated Signal

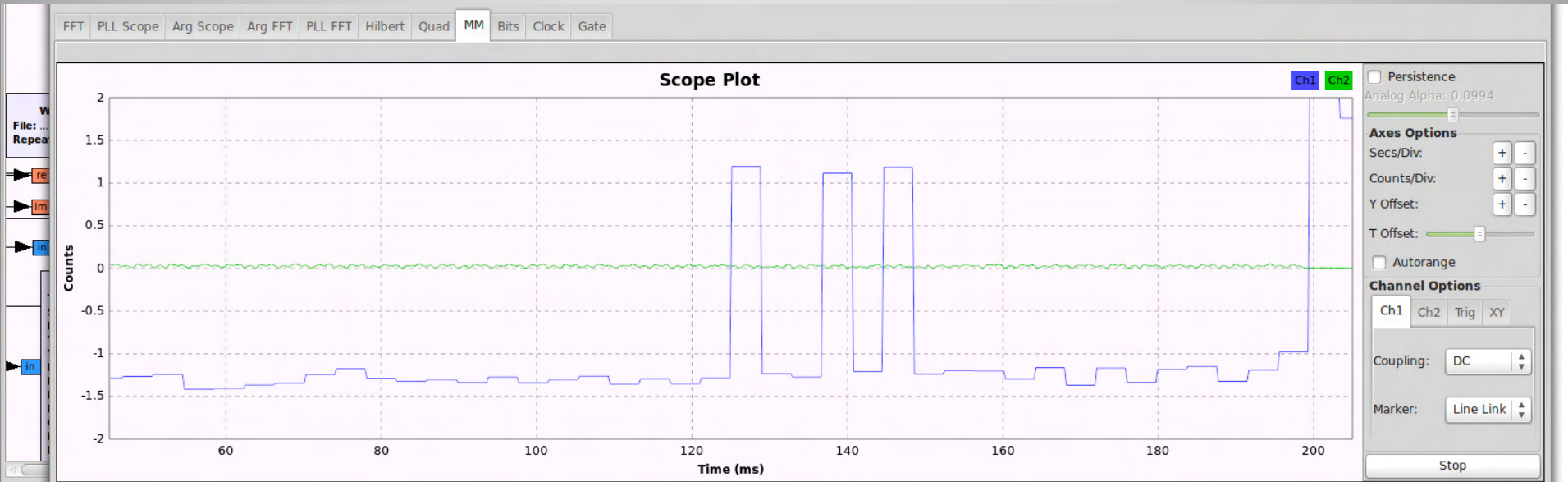


# Self-clocking Signal





# Receiver Clock Recovery



# Self-locked Sampling



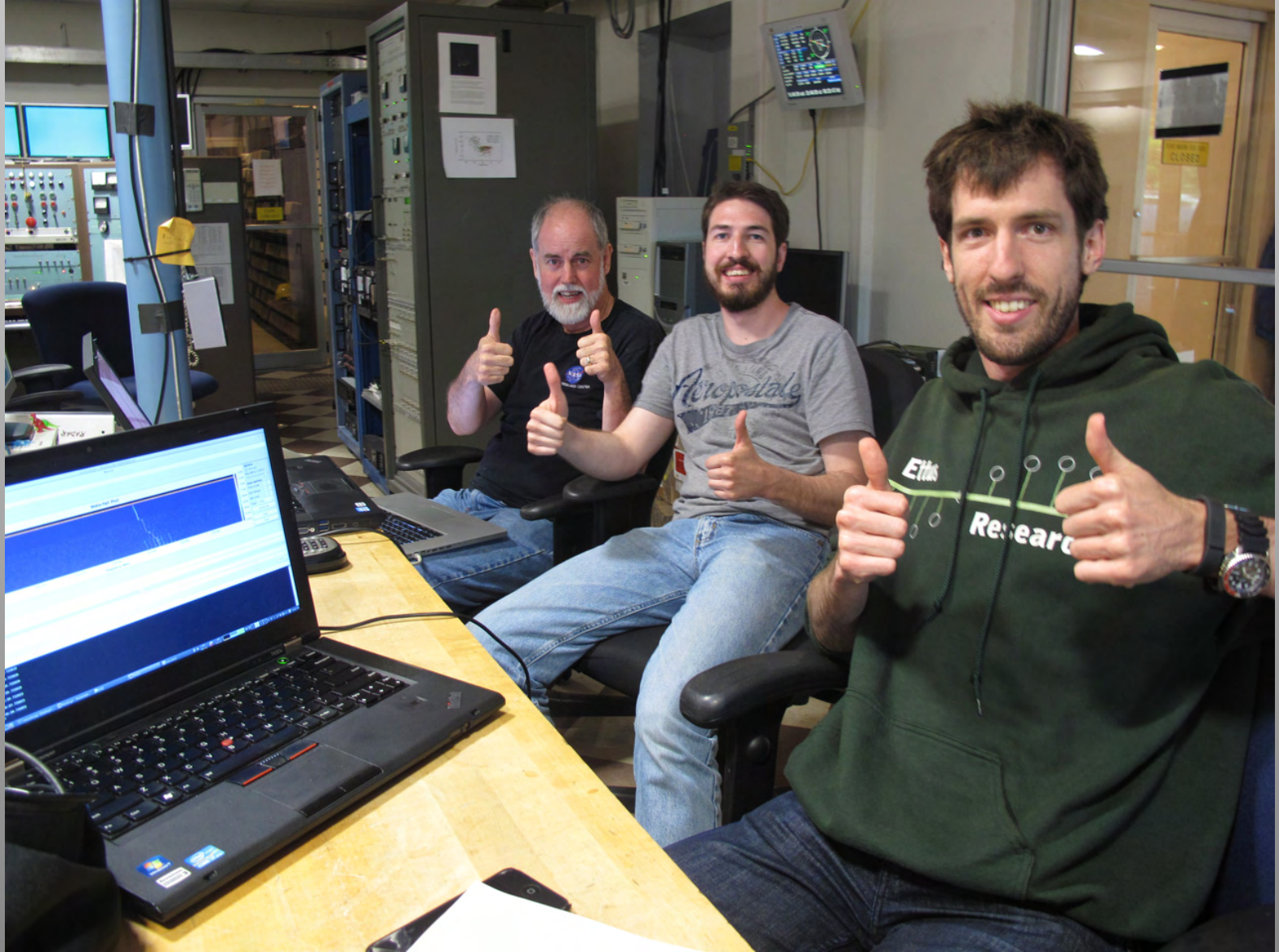


# Telemetry

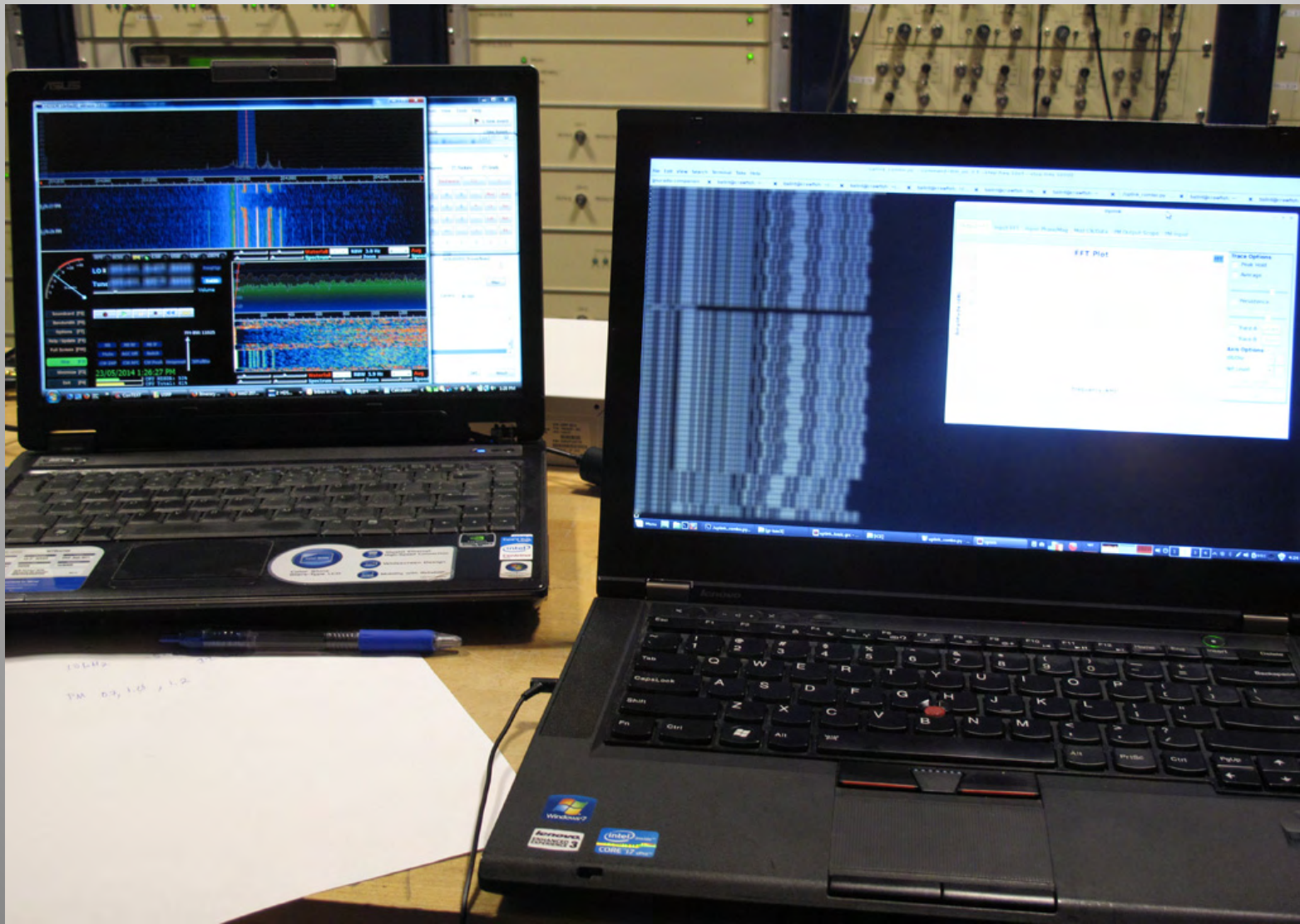
- Telemetry can be downlinked at a variety of different bit rates
- Convolutionally encoded
  - $K=24$
  - $\frac{1}{2}$  rate
  - Quick-Look-In code
- “Nonsystematic Convolutional Codes for Sequential Decoding in Space Applications” – Massey & Costello



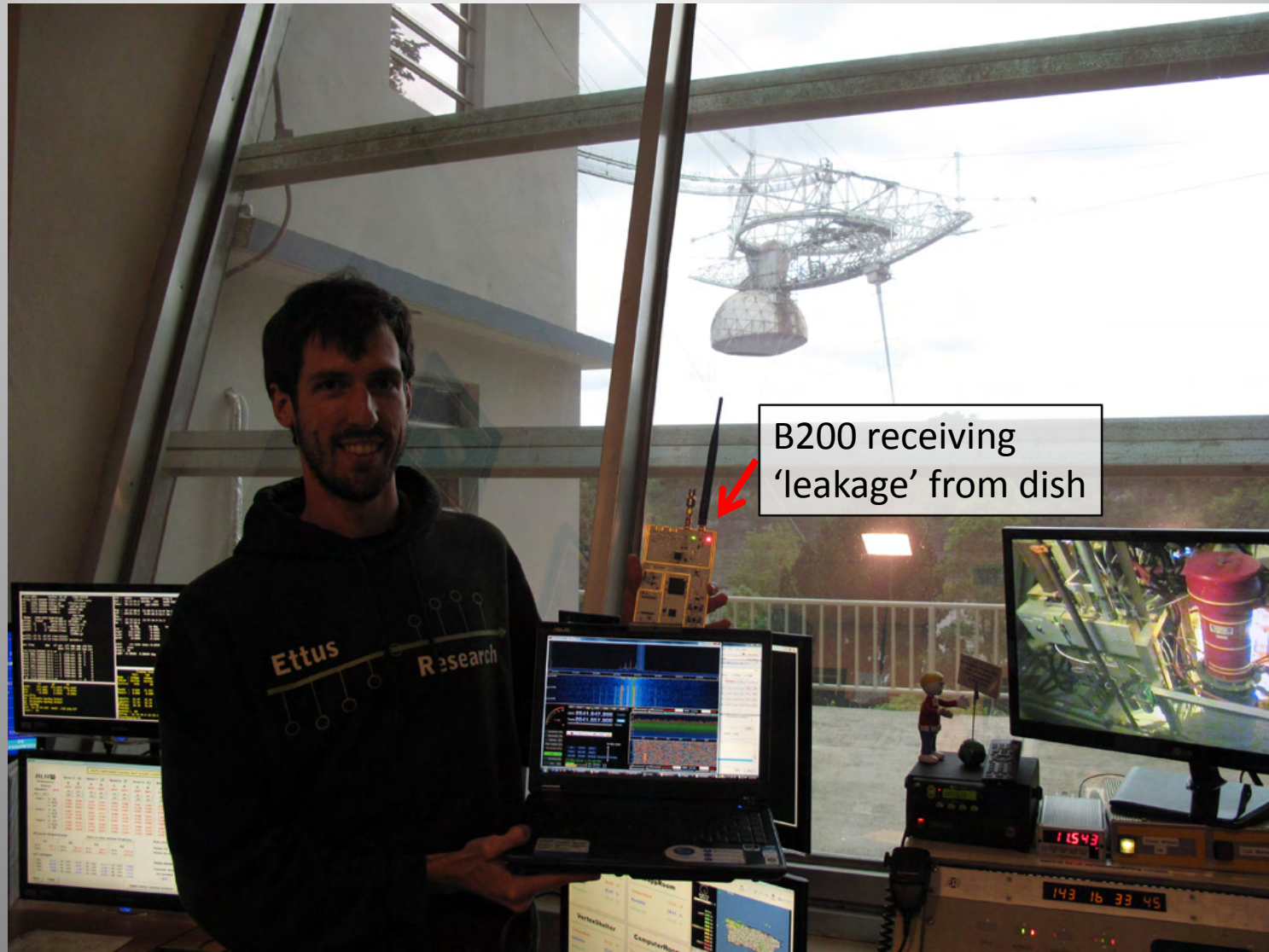
# Fingers Crossed



# Transmission to Enable Telemetry



# Verifying Transmitted Signal



# We Have Telemetry!

Turned out that all interpretations & assumptions were valid, so it wasn't necessary to run all the permutations



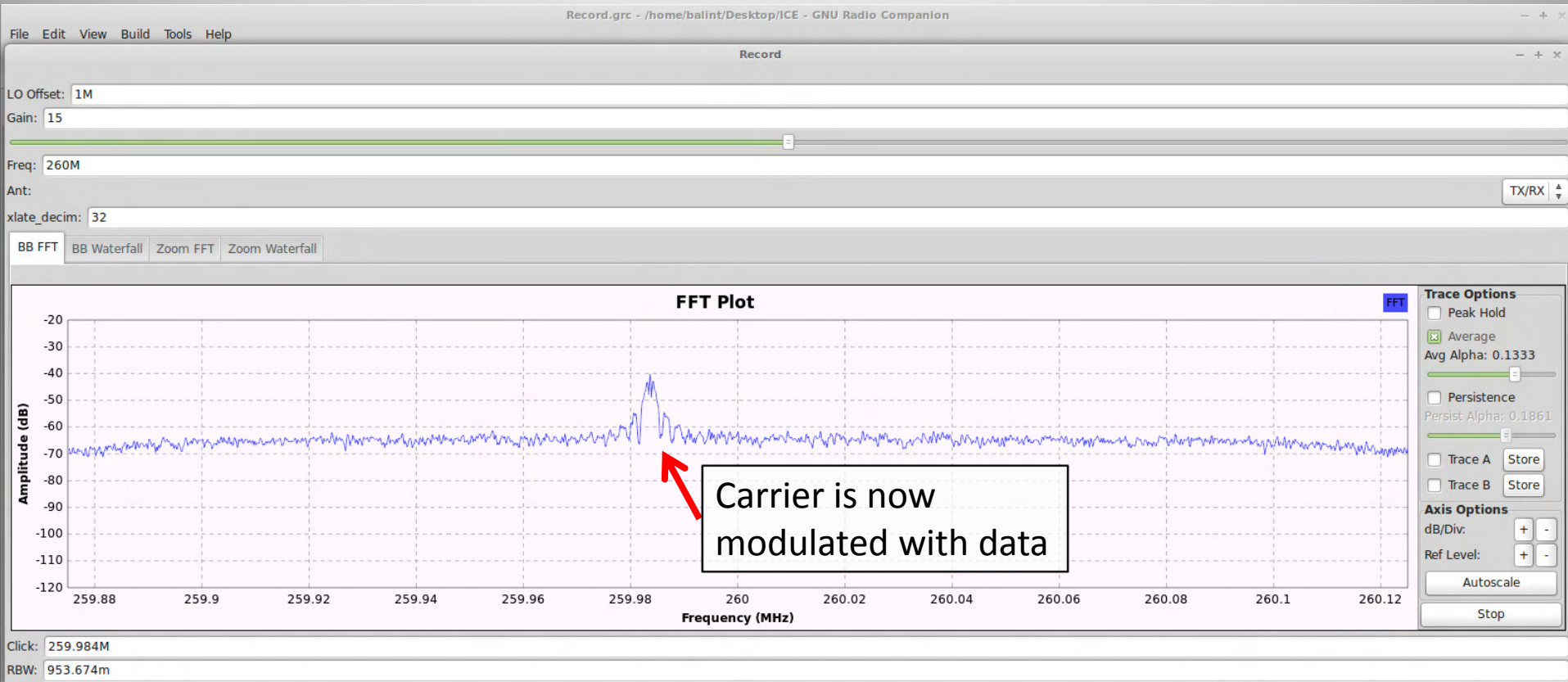
# Celebration

Phil Perillat: lives and breathes the telescope

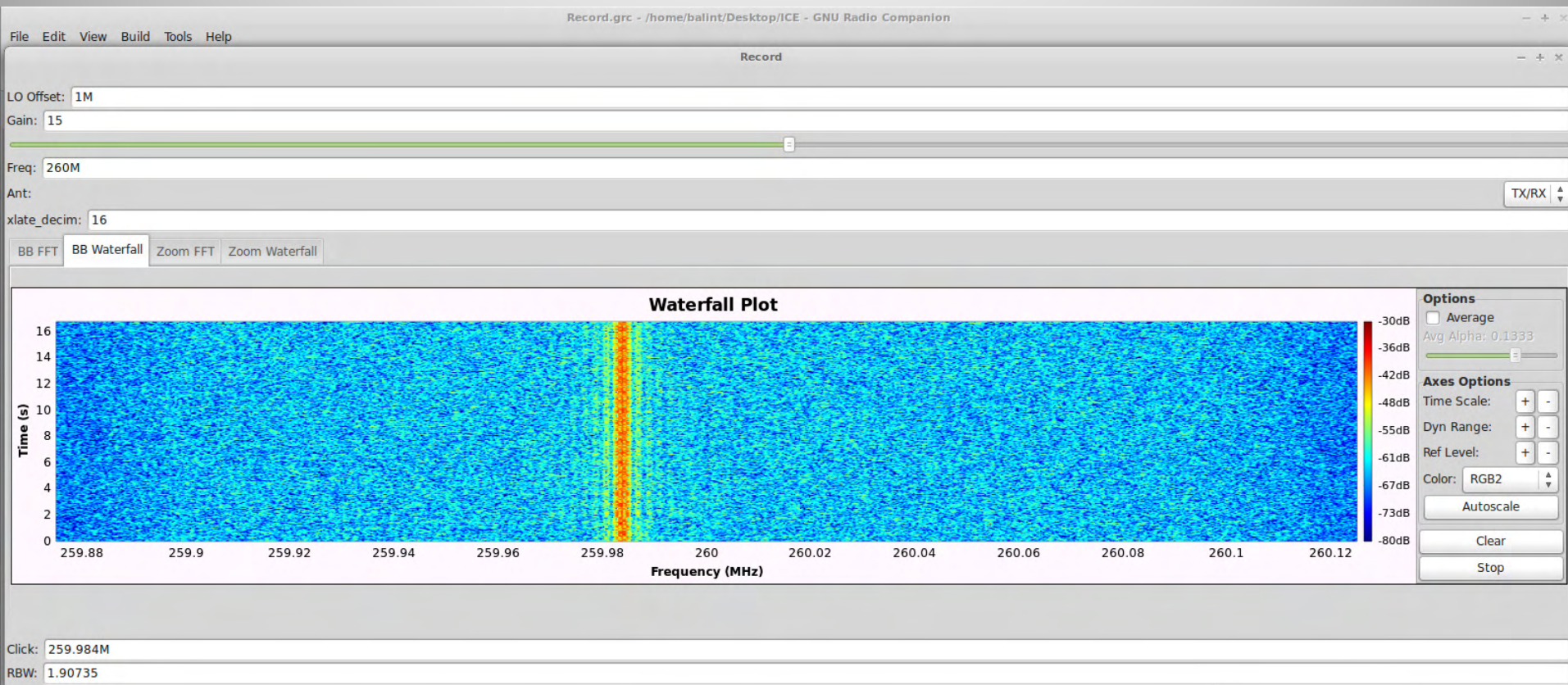




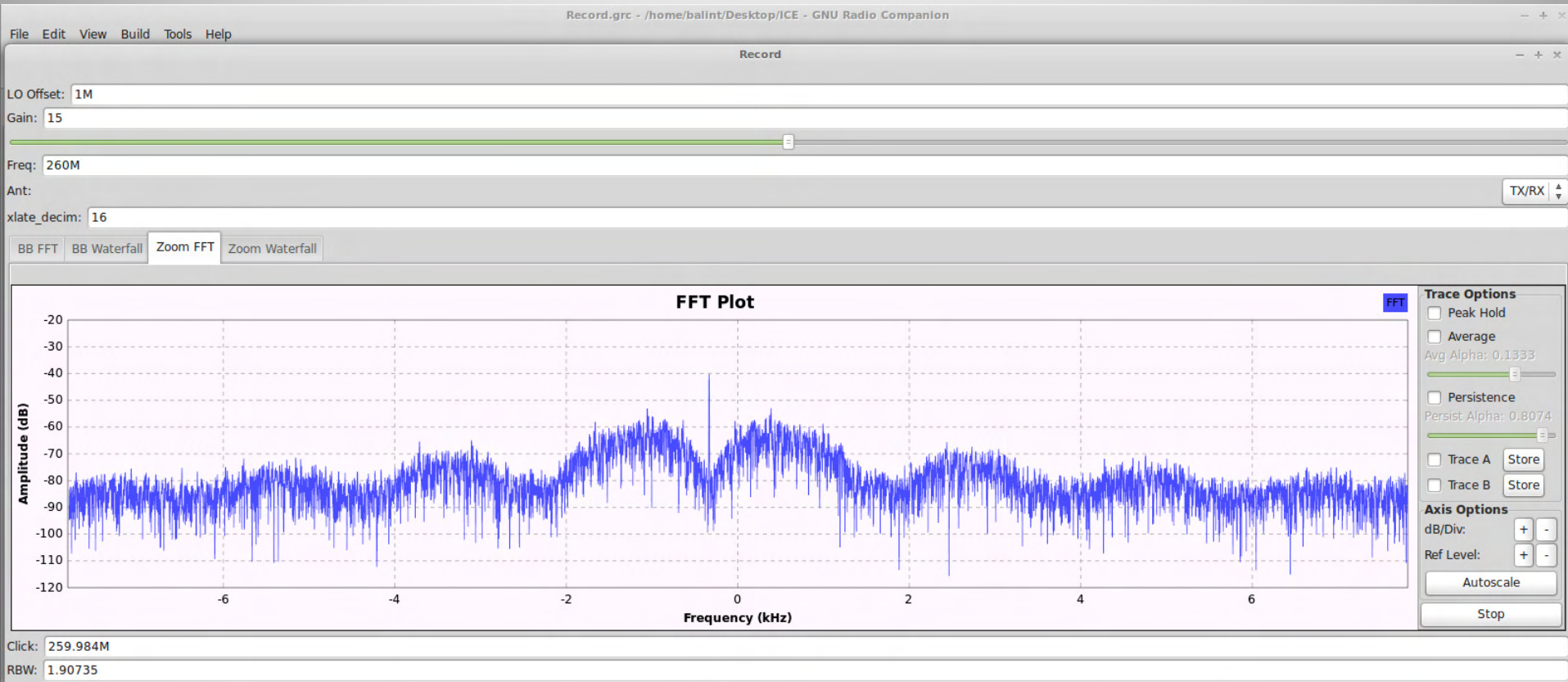
# Live Sampled Baseband



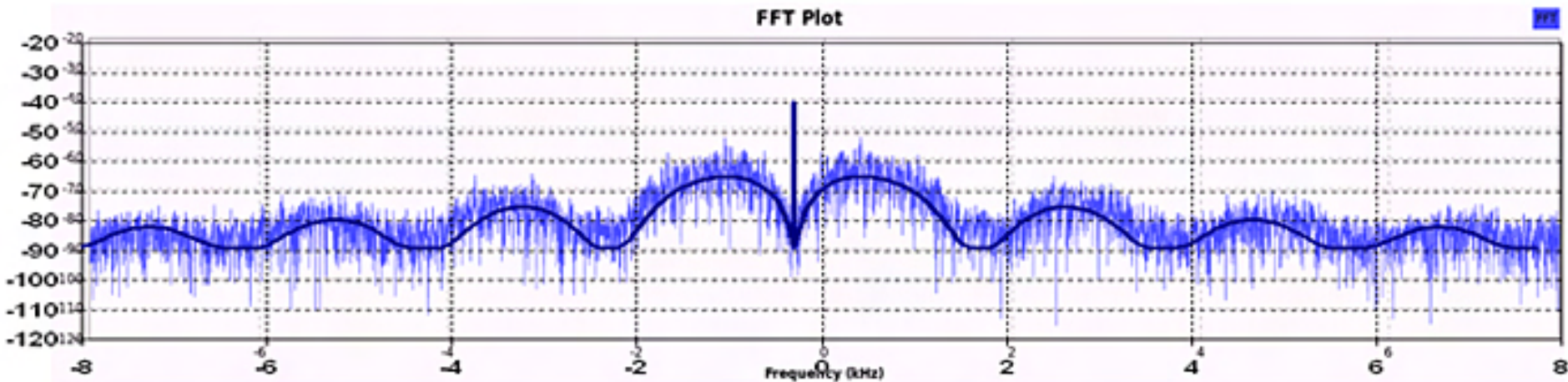
# Live Sampled Baseband



# Zoomed Baseband



# Compare to Simulation



assumptions:

512 bps, R=1/2 convolutional code  
biphase PSK, 60 deg mod index

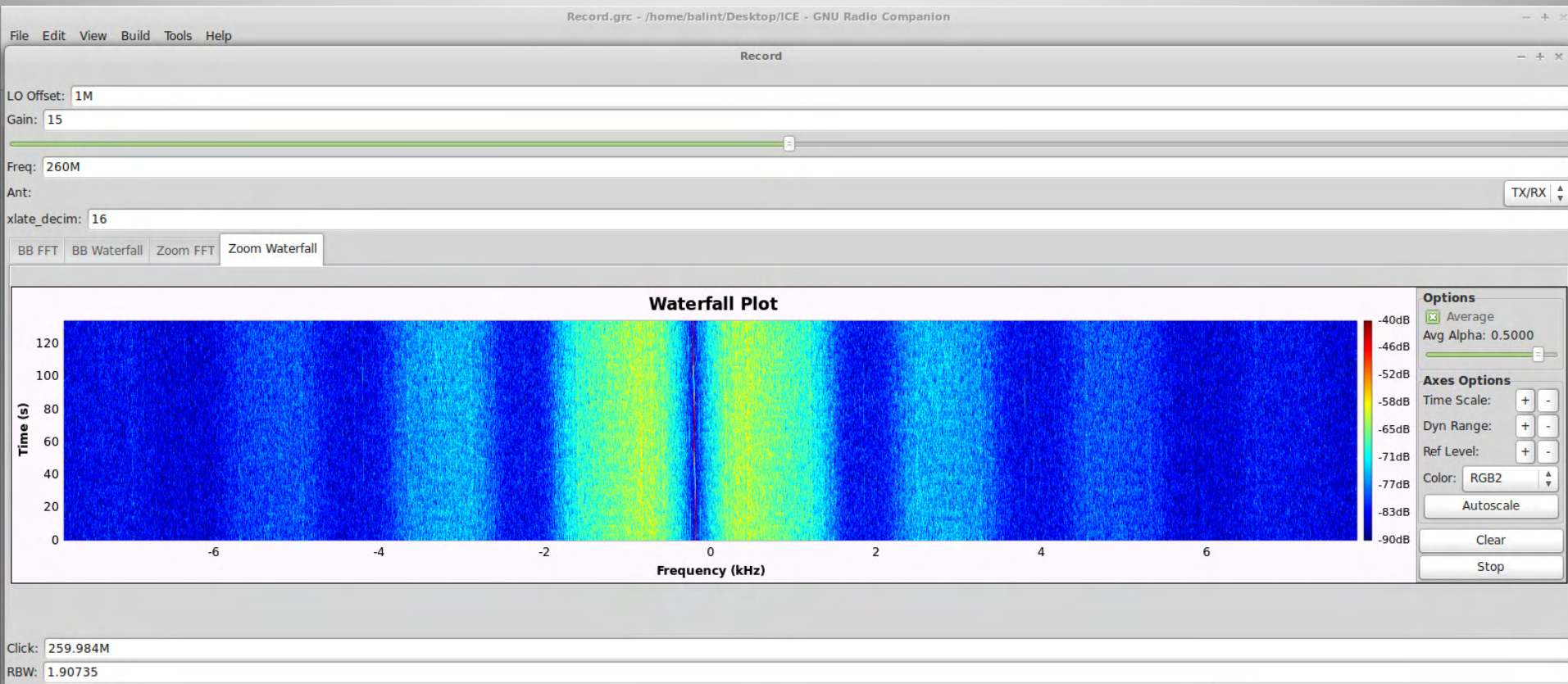
free parameters:

noise power, total signal power, frequency offset

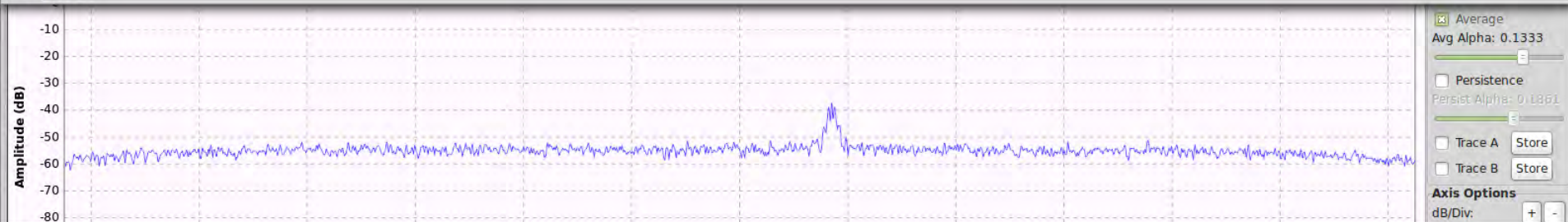
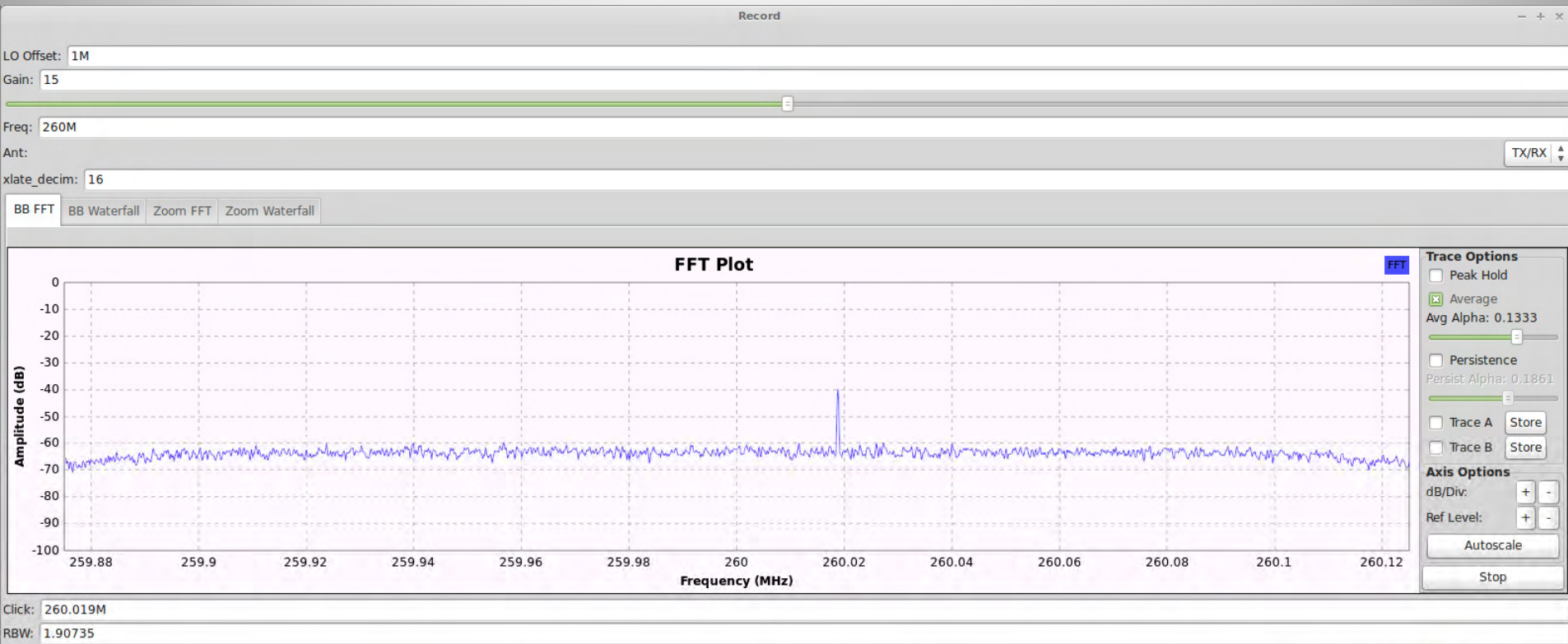
measured spectrum by ISEE-3 Reboot Project

simulated spectrum by A. Vollhardt (DH2VA), AMSAT-DL

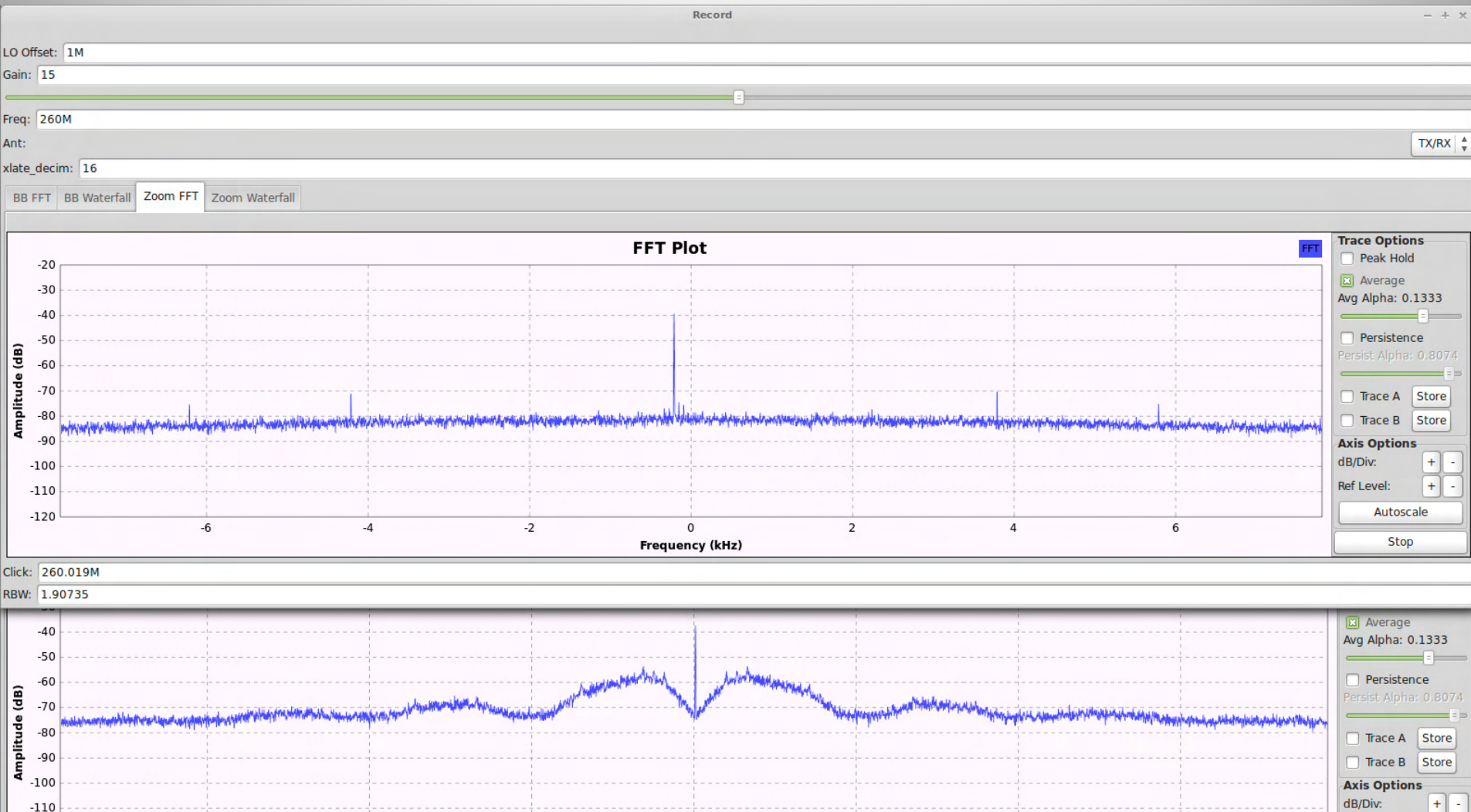
# Zoomed Baseband



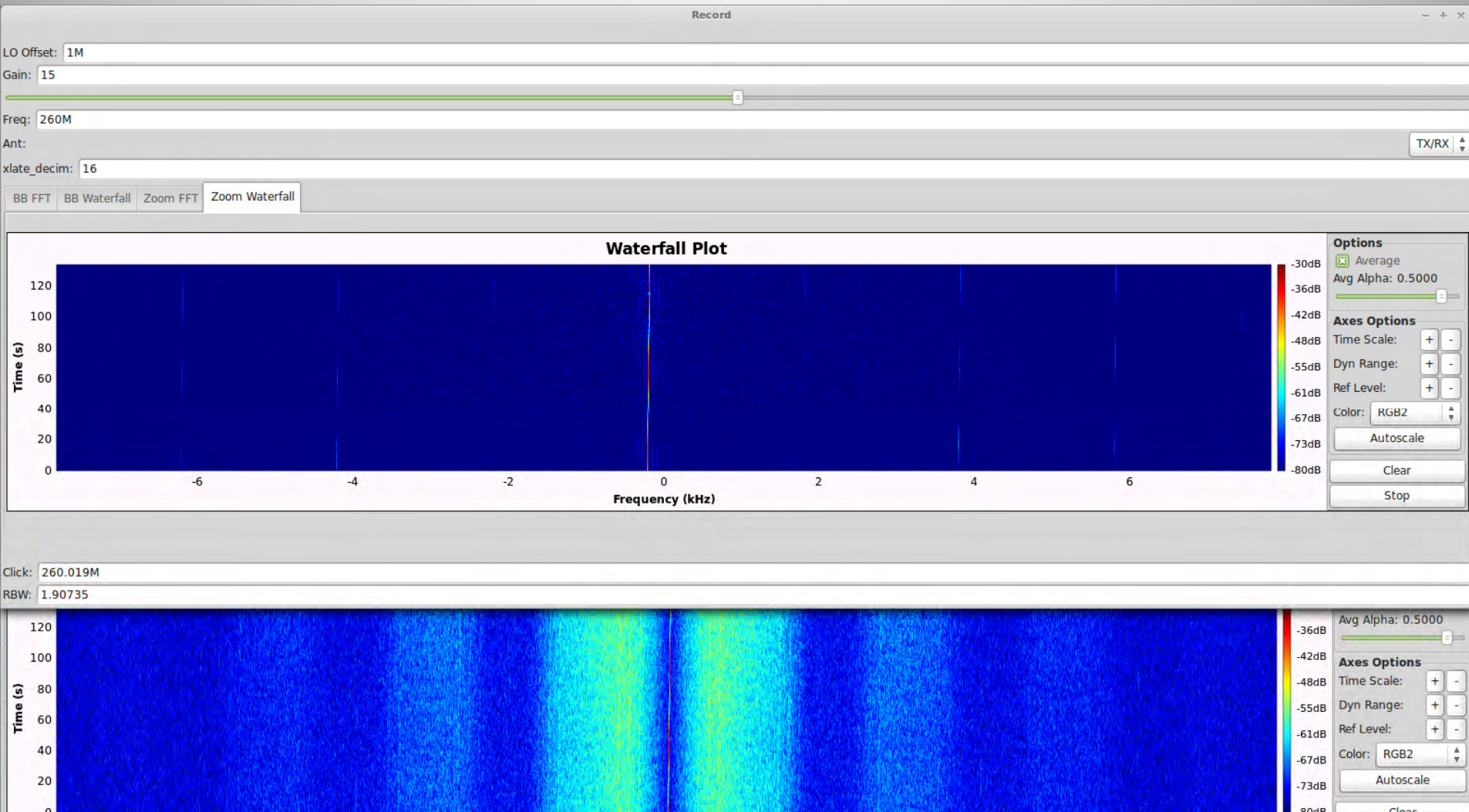
# Only carrier on B, TLM on A



# Zoomed: Only carrier on B, TLM on A



# Only carrier on B, TLM on A







# Commanding

- Transponder B did not respond to commands
  - Perhaps we had a marginal link to the space probe based on antenna, receiver and distance
- Transponder *A* *did* respond
  - TLM was successfully re-enabled
- Then commanded TLM to be enabled on B, uplinked via A

# TLM on A and B

The image displays two side-by-side screenshots of the GNU Radio Companion (GRC) interface, showing the 'Record' block's FFT plot. The left screenshot shows a peak at 260.017M, and the right screenshot shows a peak at 260.019M. Both plots show Amplitude (dB) vs Frequency (kHz).

**Left Screenshot (Record):**

- LO Offset: 1M
- Gain: 15
- Freq: 260M
- Ant: TX/RX
- xlate\_decim: 16
- Trace Options: Peak Hold, Average (Avg Alpha: 0.1333), Persistence (Persist Alpha: 0.8074), Trace A, Trace B
- Axis Options: dB/Div, Ref Level, Autoscale, Stop
- Click: 260.017M
- RBW: 1.90735

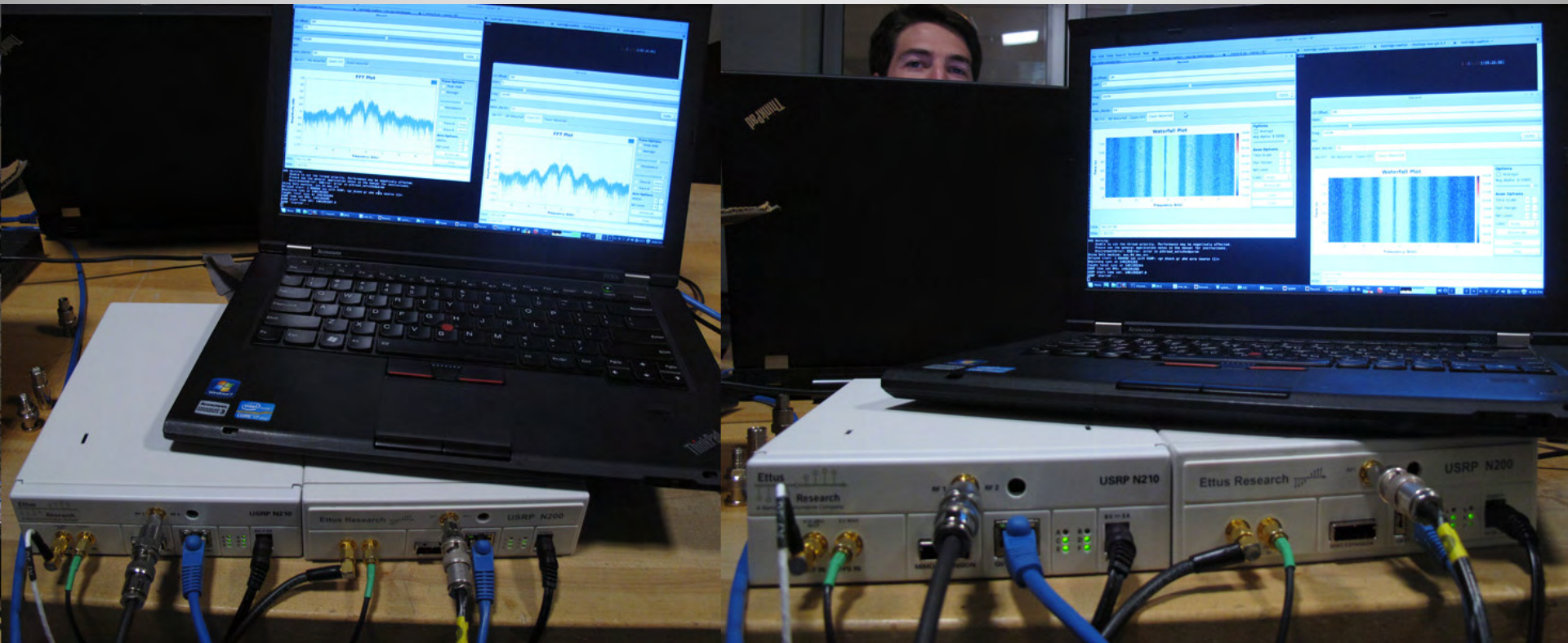
**Right Screenshot (Record):**

- LO Offset: 1M
- Gain: 5
- Freq: 260M
- Ant: TX/RX
- xlate\_decim: 16
- Trace Options: Peak Hold, Average (Avg Alpha: 0.1333), Persistence (Persist Alpha: 0.8074), Trace A, Trace B
- Axis Options: dB/Div, Ref Level, Autoscale, Stop
- Click: 260.019M
- RBW: 1.90735

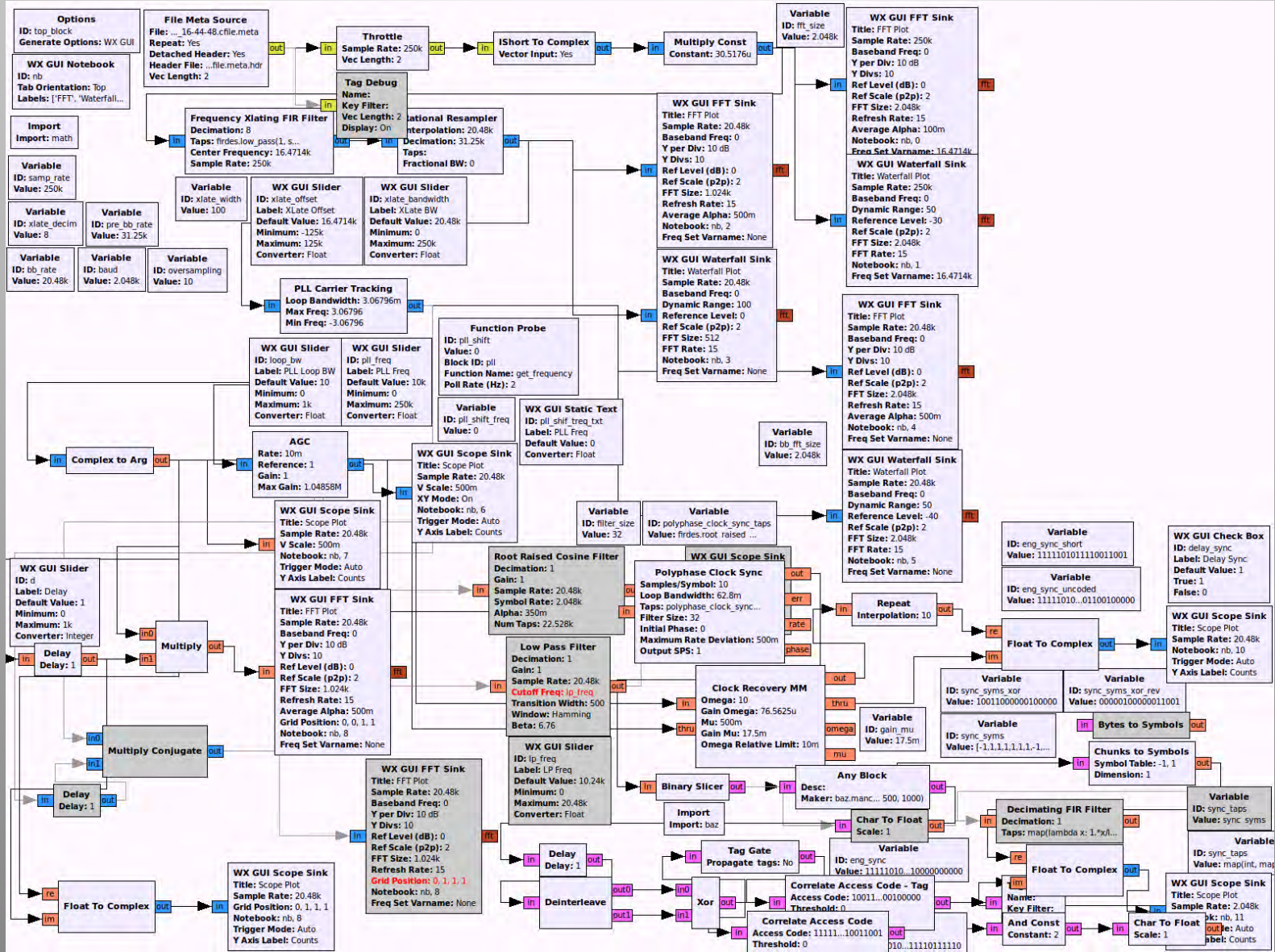
**Terminal Output (Bottom Left):**

```
UHD Warning:
Unable to set the thread priority. Performance may be negatively affected.
Please see the general application notes in the manual for instructions.
EnvironmentError: OSError: error in pthread_setschedparam
Using Volk machine: avx_64_mmx_orc
Delayed start: 2.000000 sec with USRP: <gr_block gr_uhd usrp source (1)>
Beginning sync at 1401393263
Caught local sync at 1401393264
USRP time set PPS: 1401393265
USRP start time set: 1401393267.0
USRP 'started'...
```

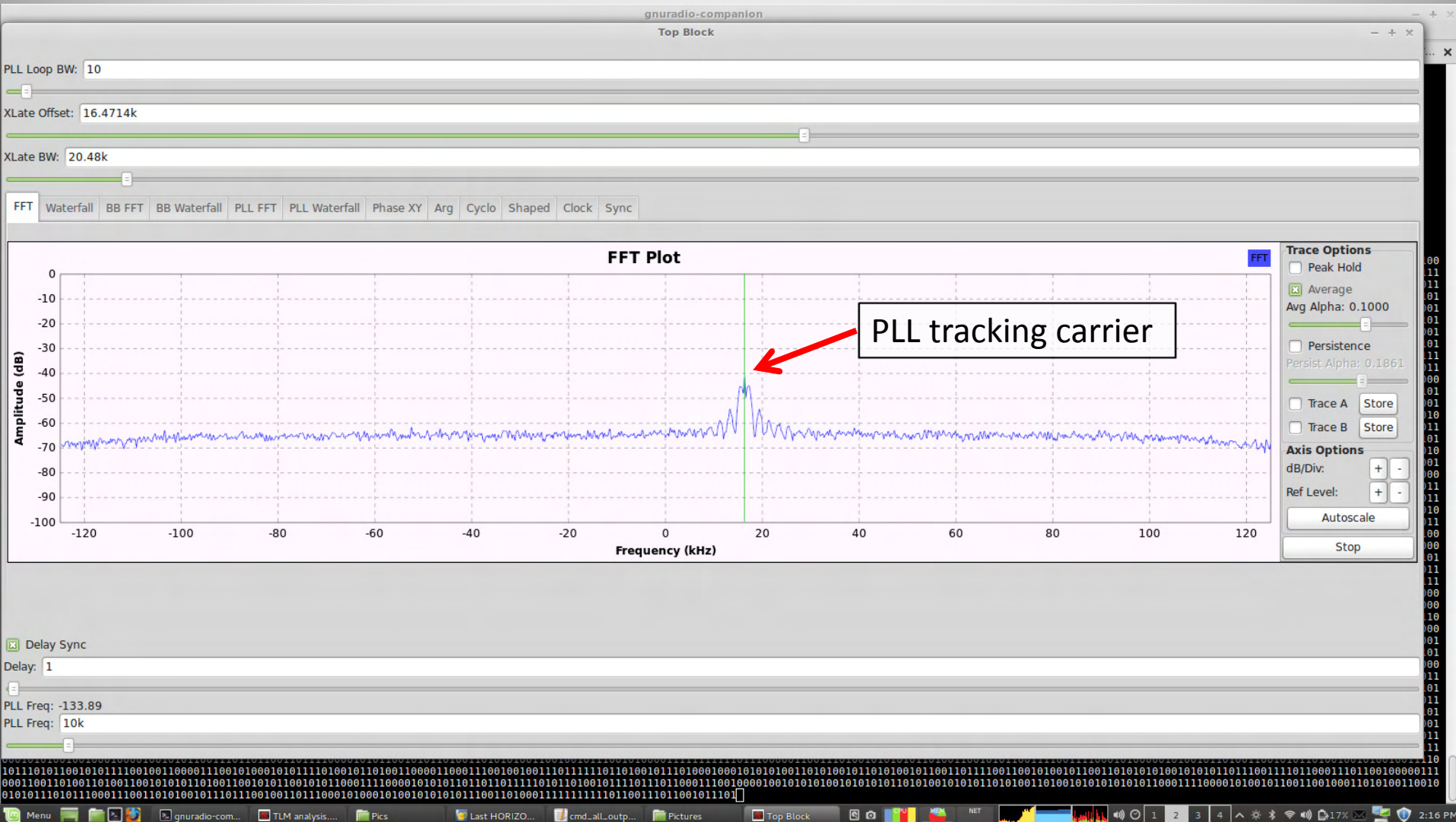
# Recording Telemetry



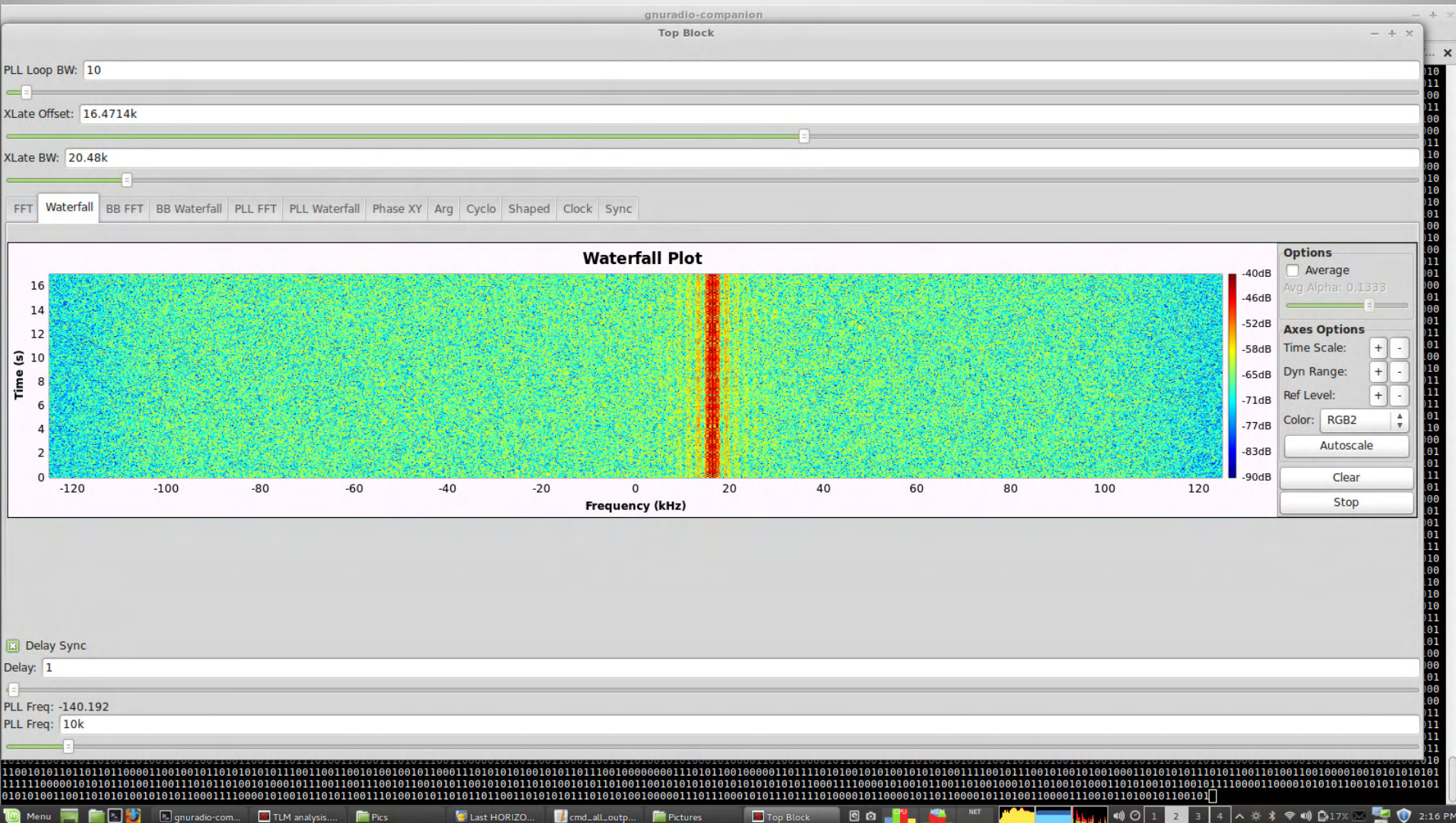
# Analysing Telemetry



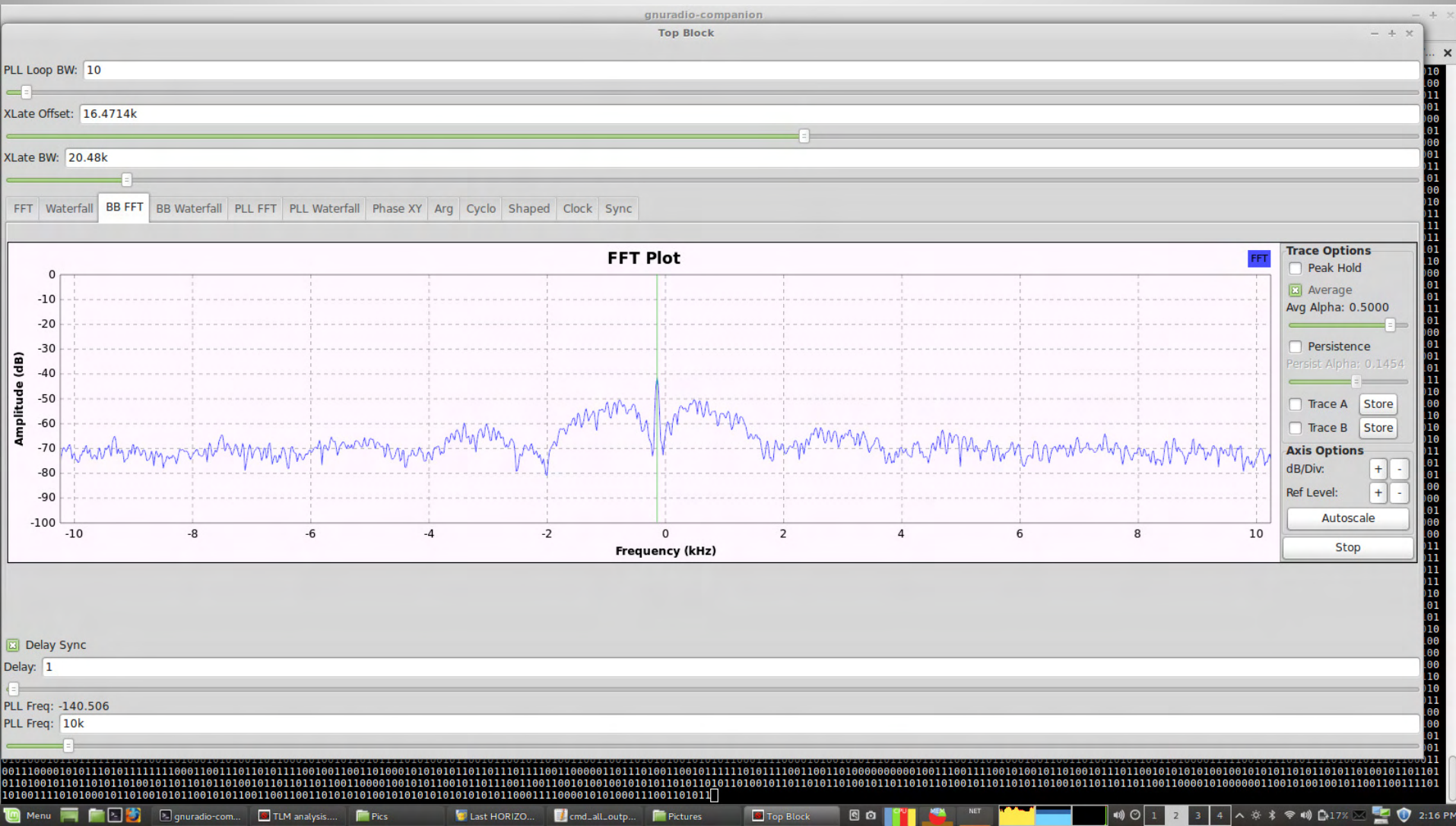
# Raw Captured Baseband



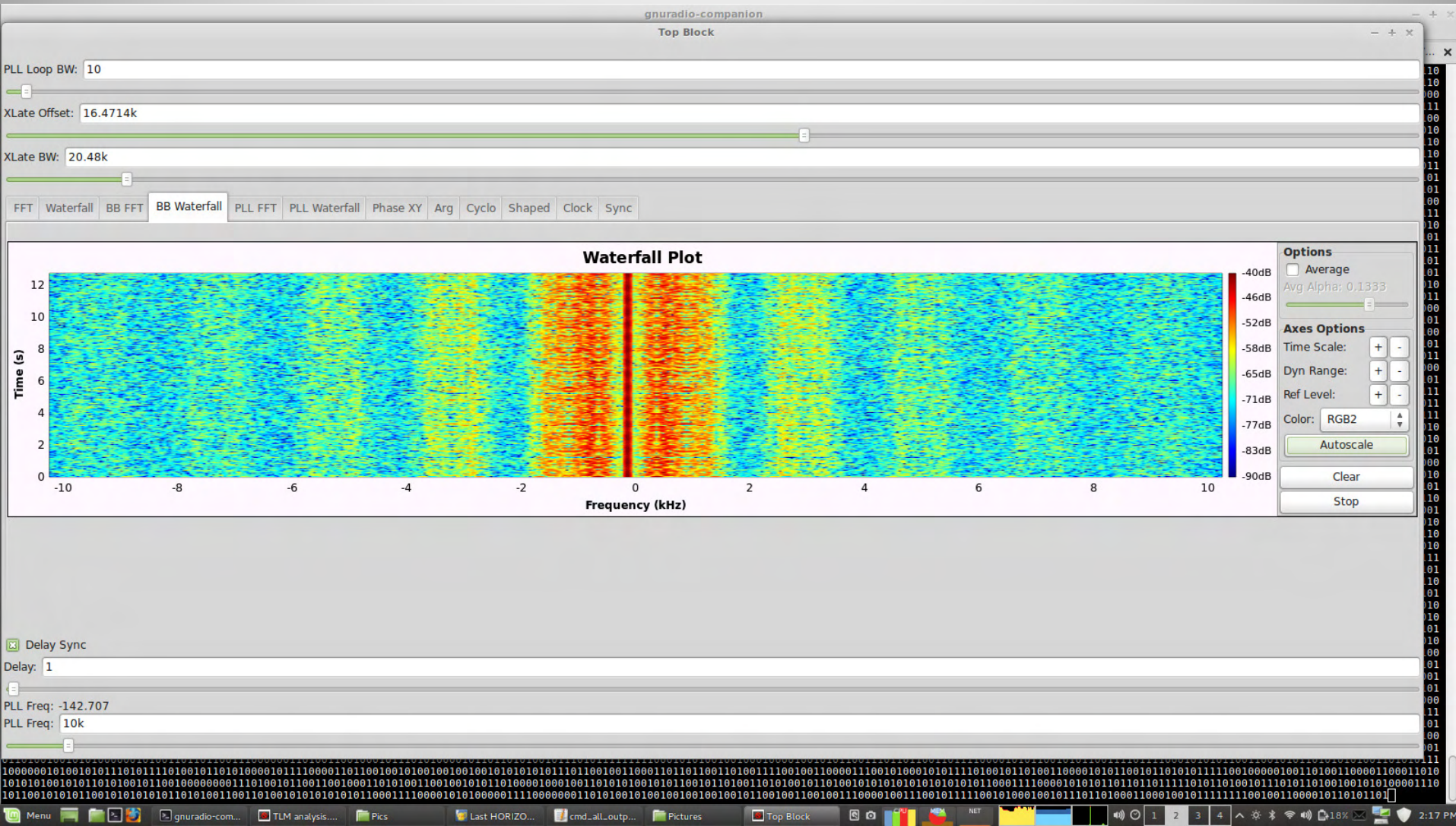
# Raw Captured Baseband



# Zoomed Baseband

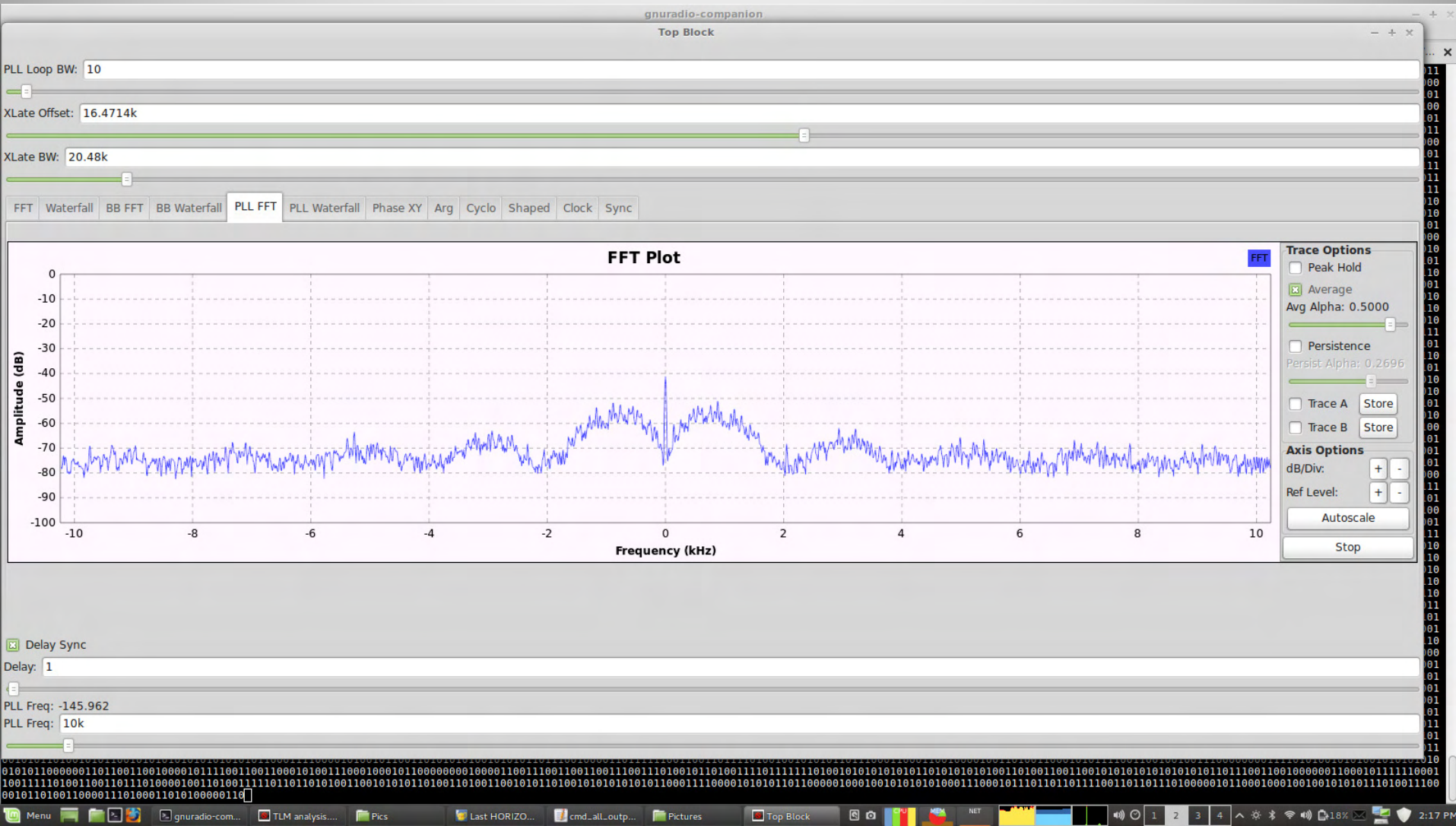


# Zoomed Baseband

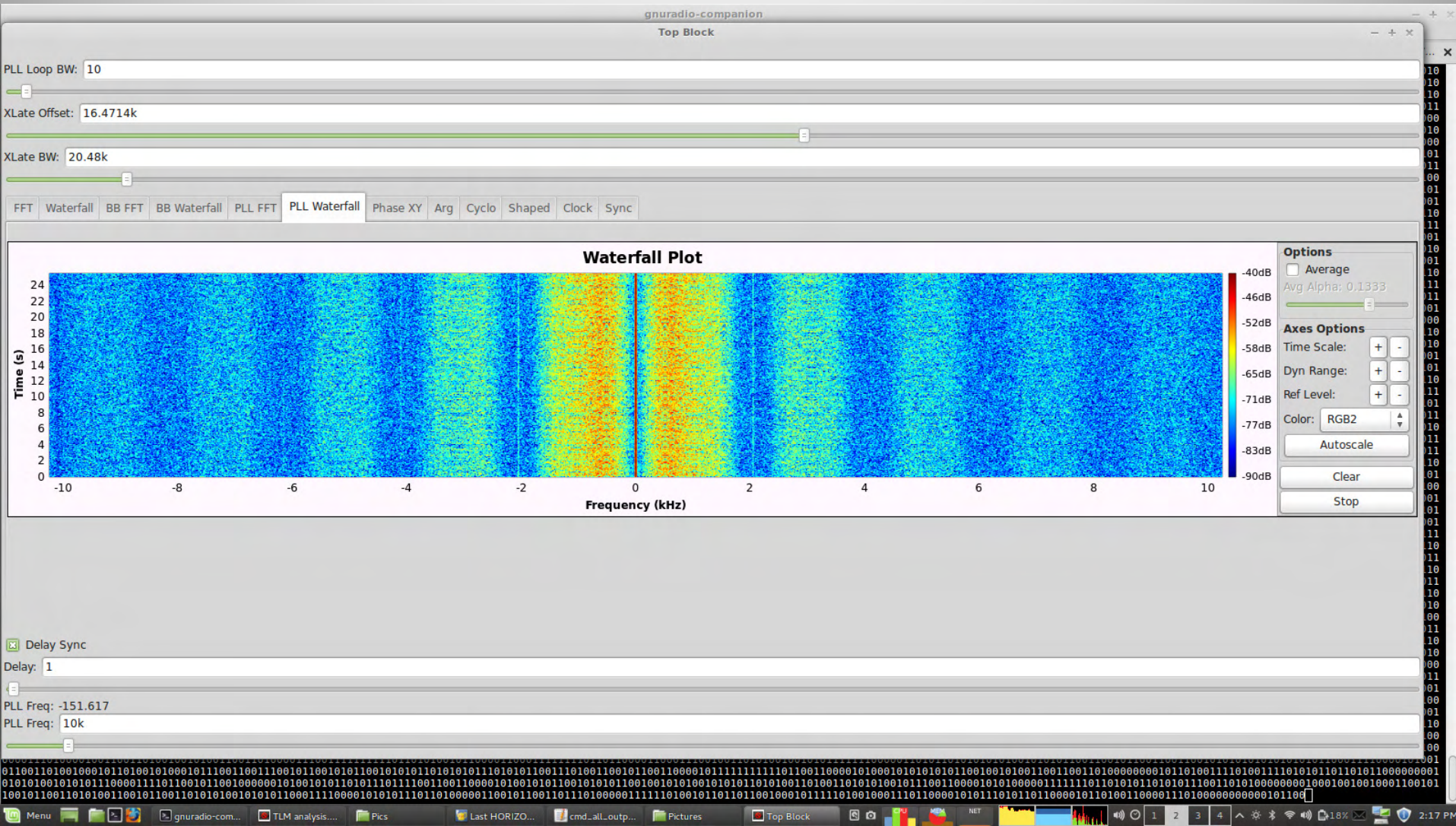




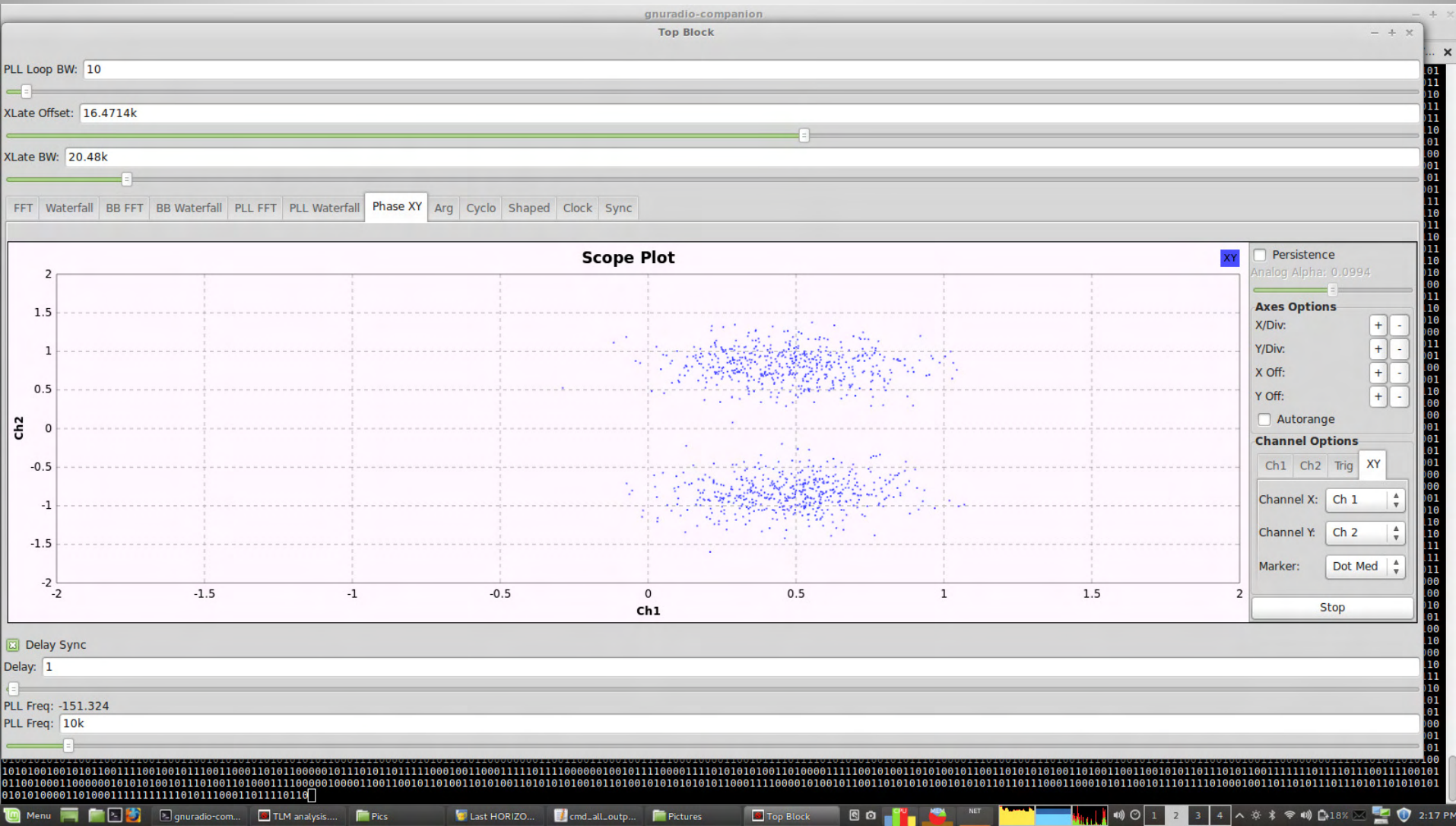
# PLL Lock



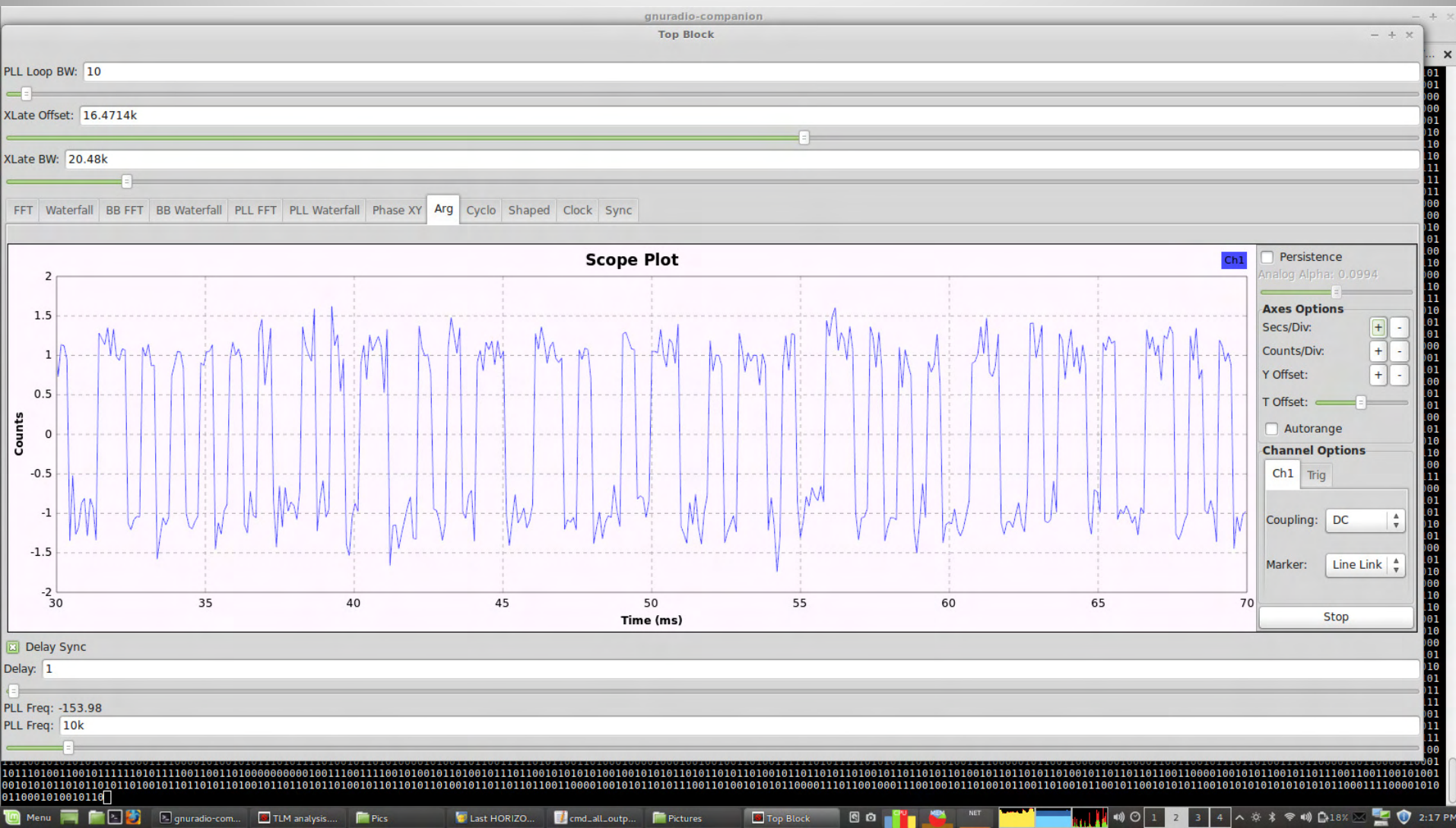
# PLL Lock



# IQ Scope Plot



# Raw Phase Demodulation



# Cyclostationary Analysis

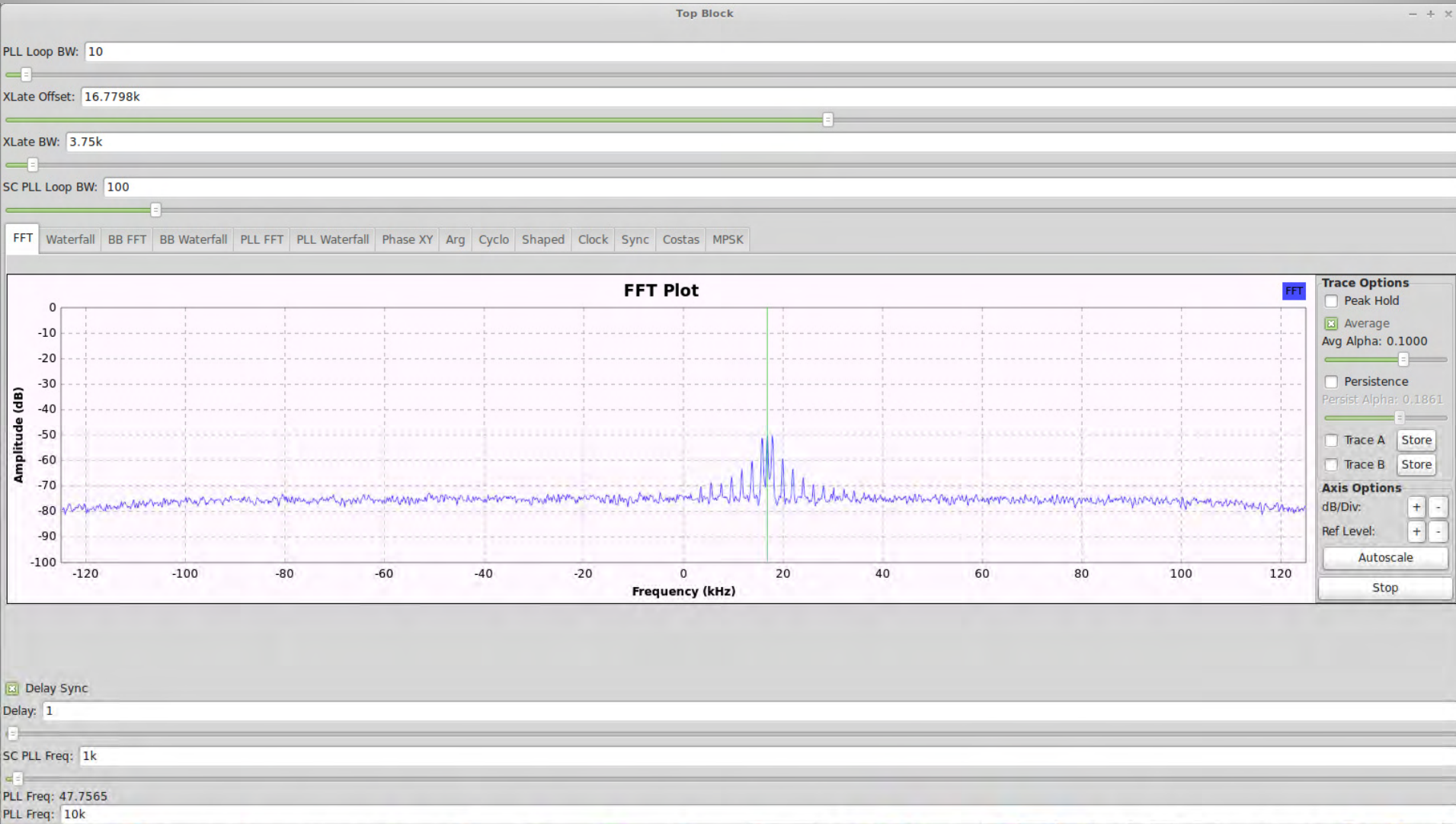




# Sync Correlation Peak

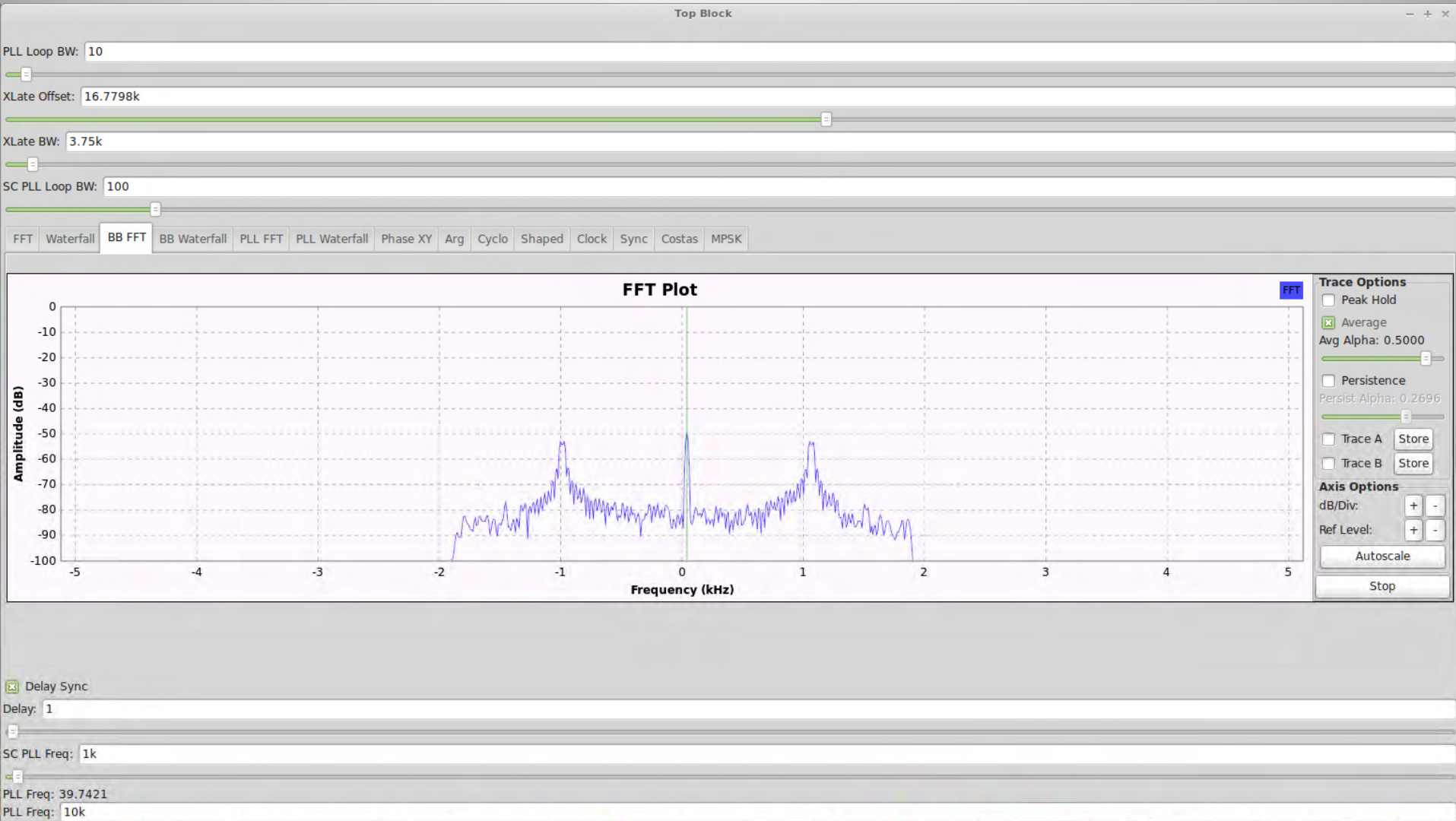


# Low-rate w/ 1024 Hz sub-carrier

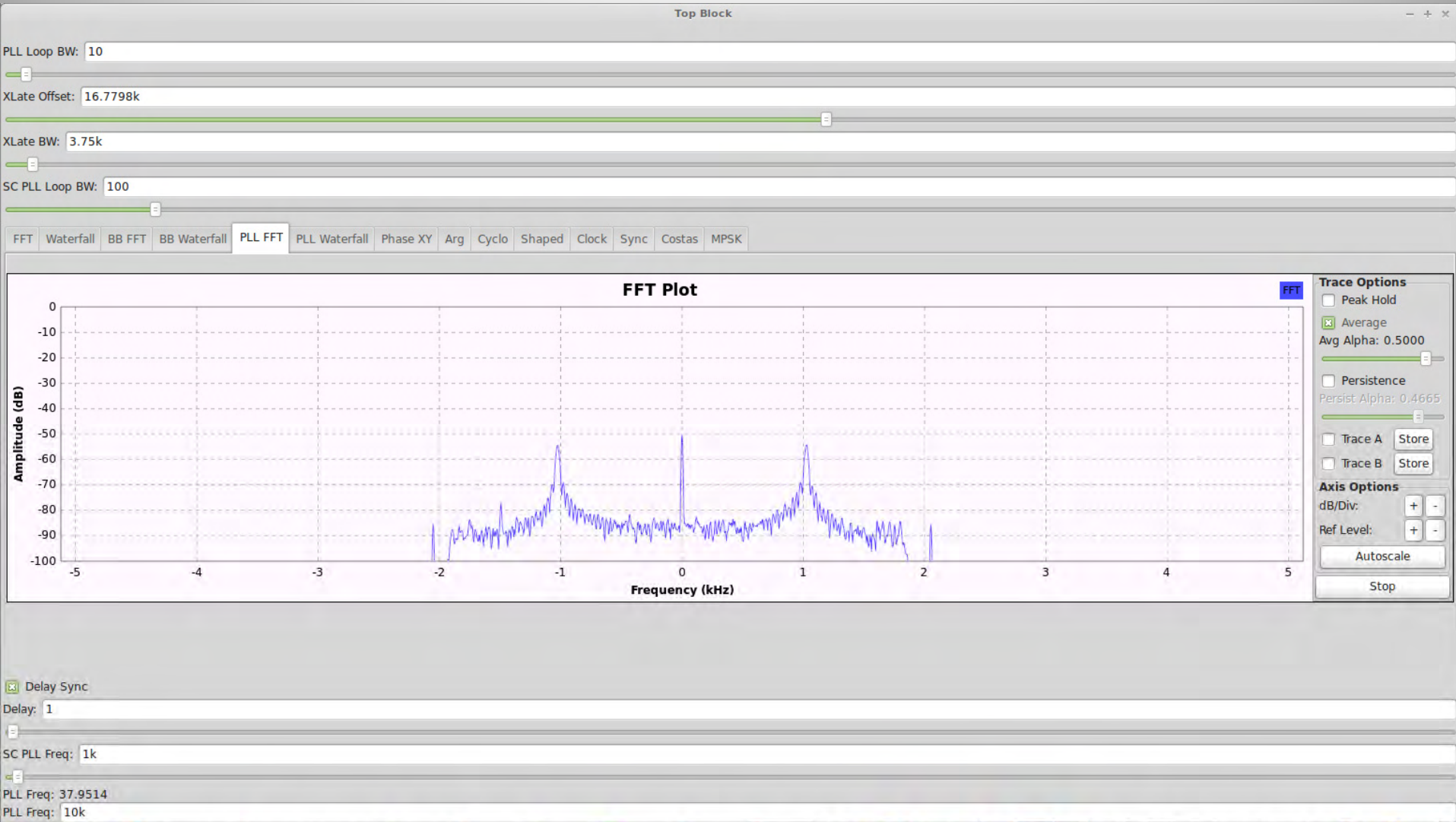




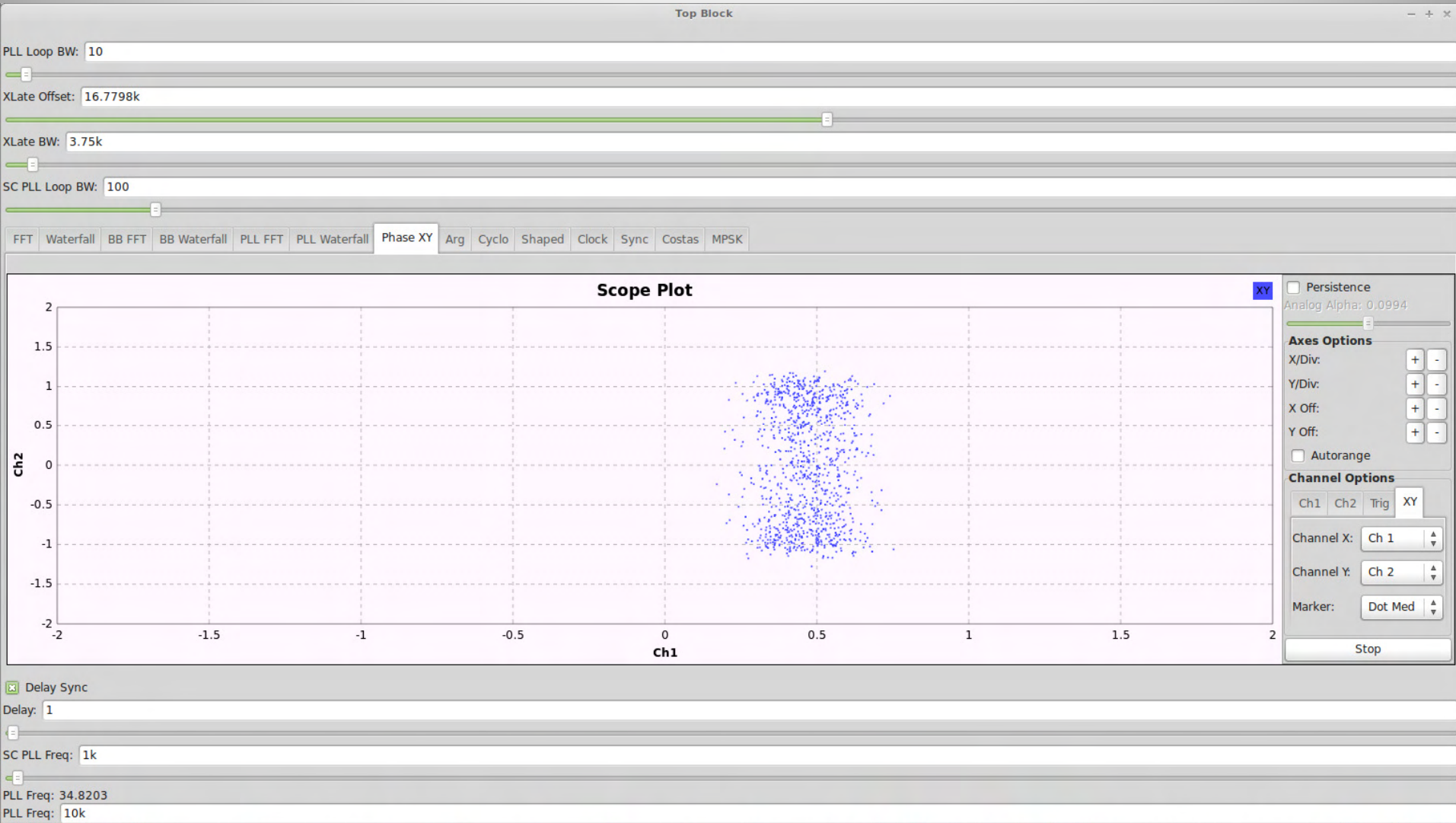
# Zoomed Baseband



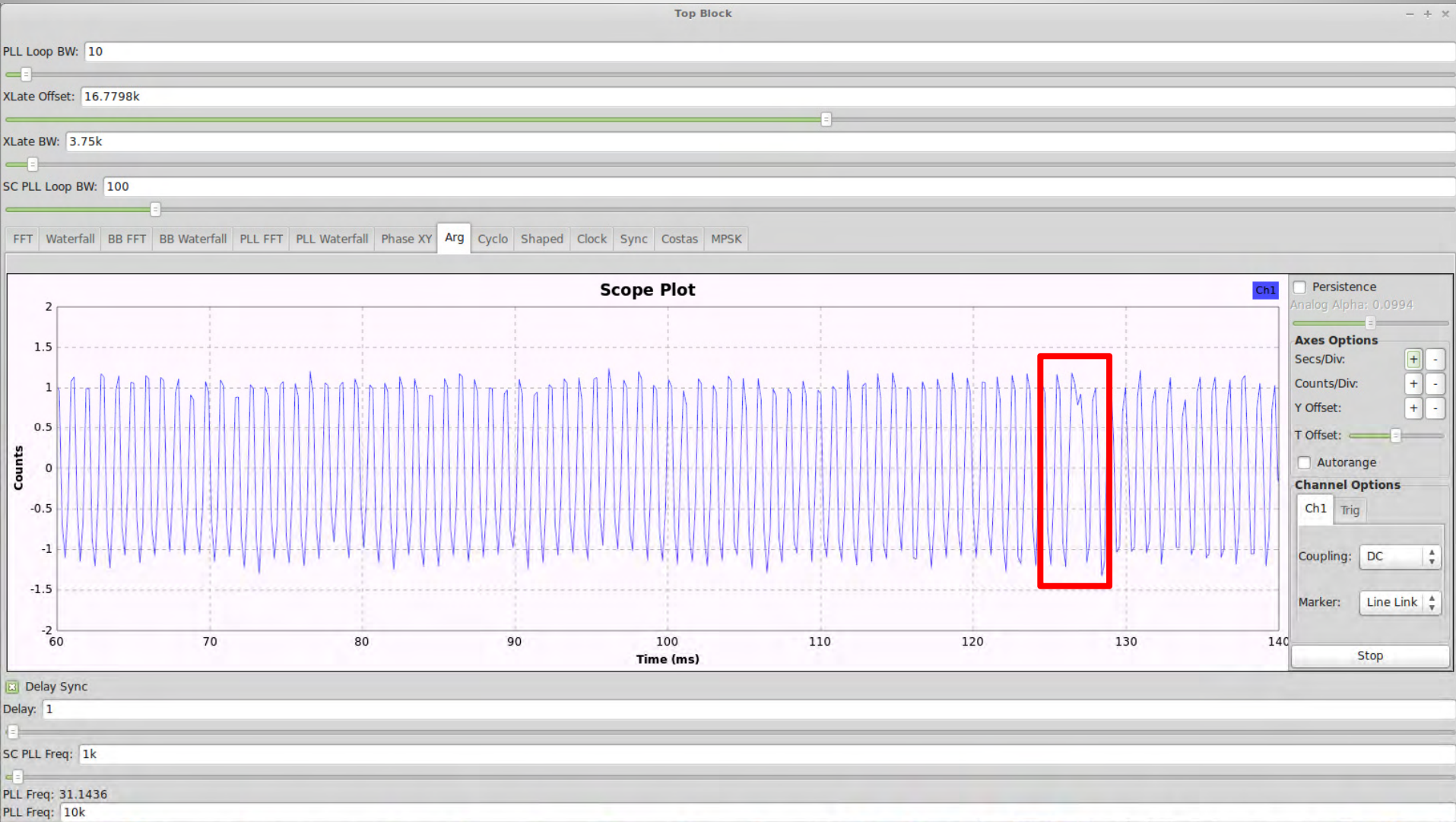
# PLL Lock



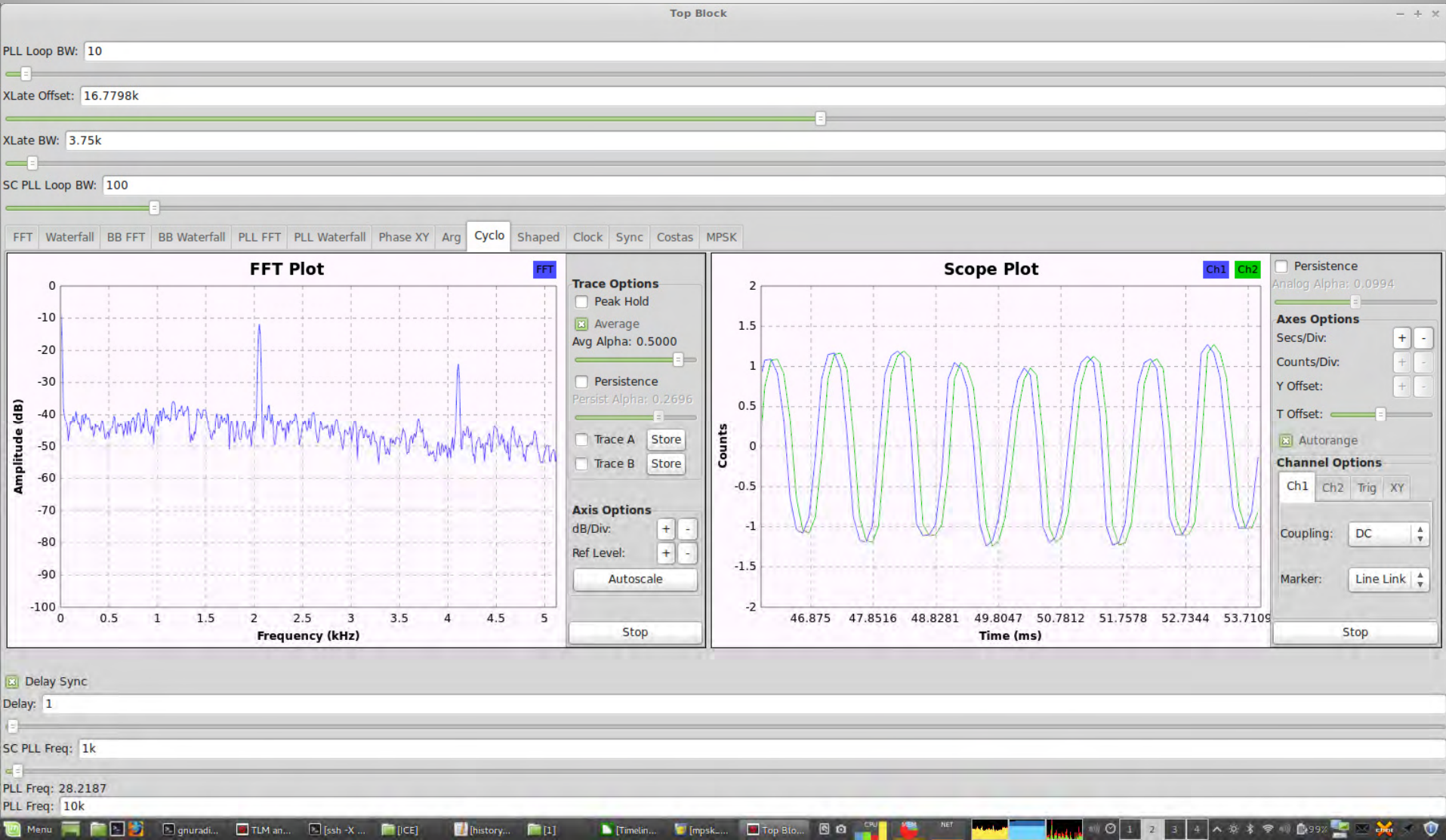
# IQ Scope Plot



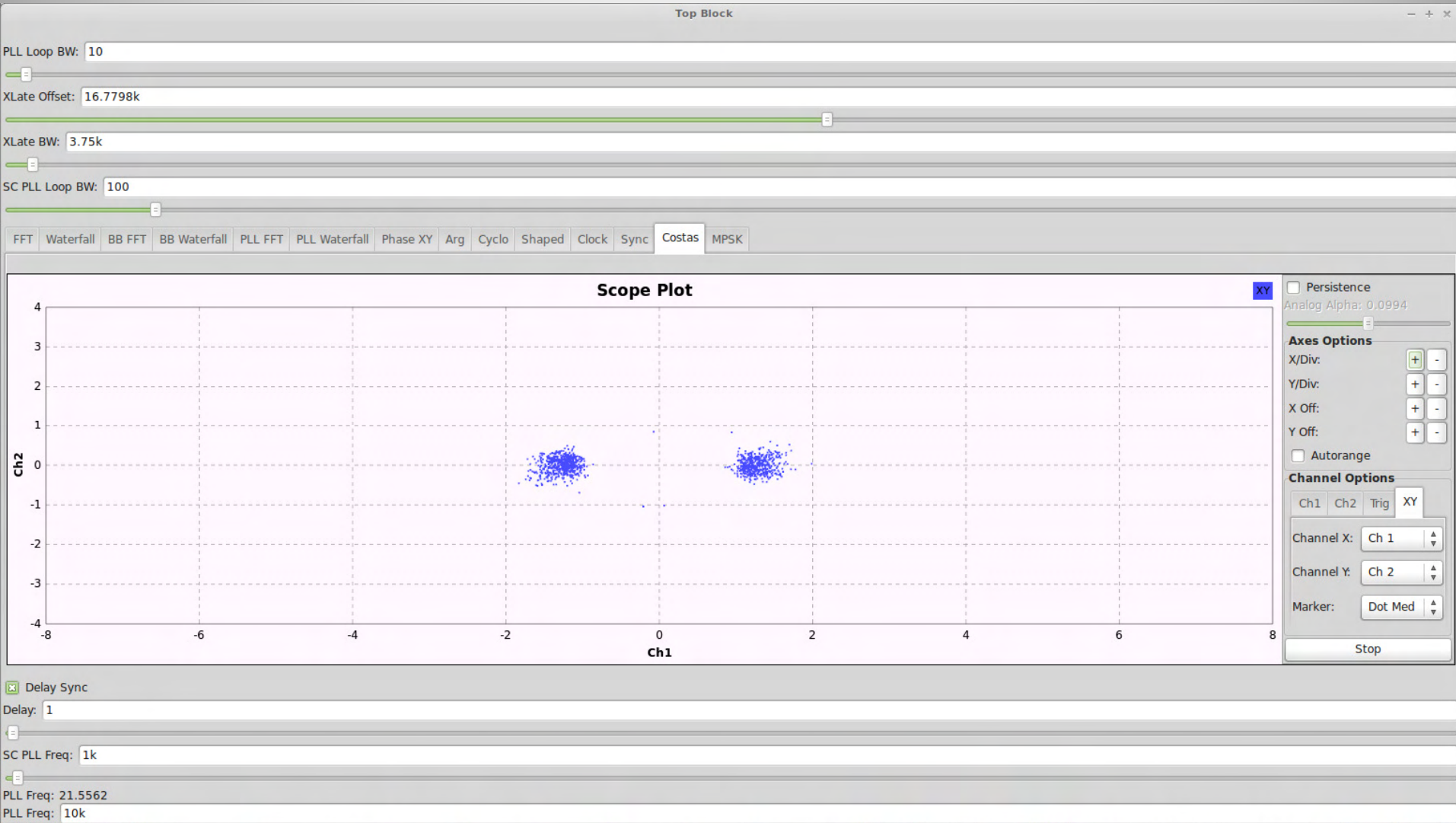
# 1024 Hz Sub-carrier



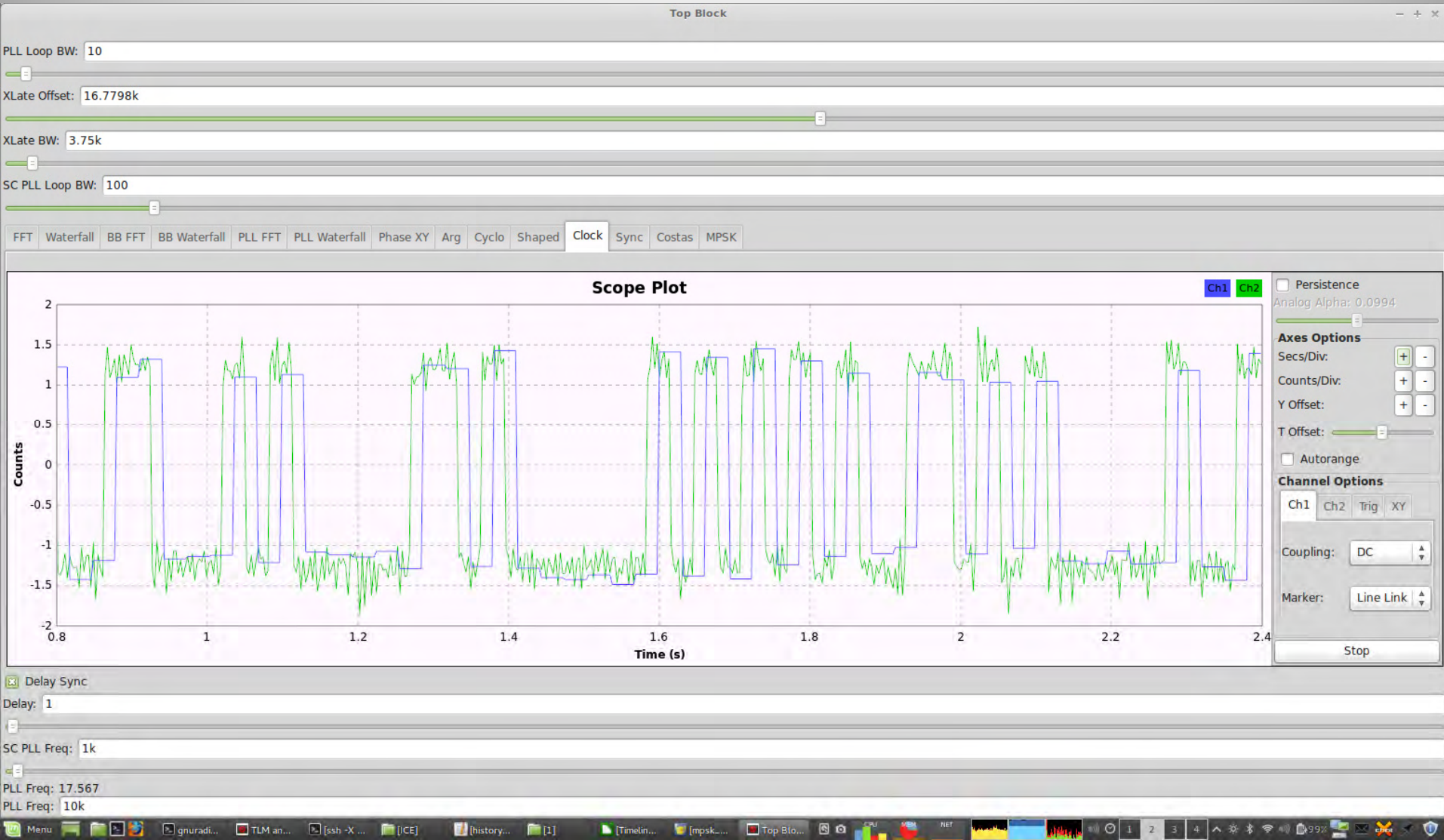
# Cyclostationary Analysis



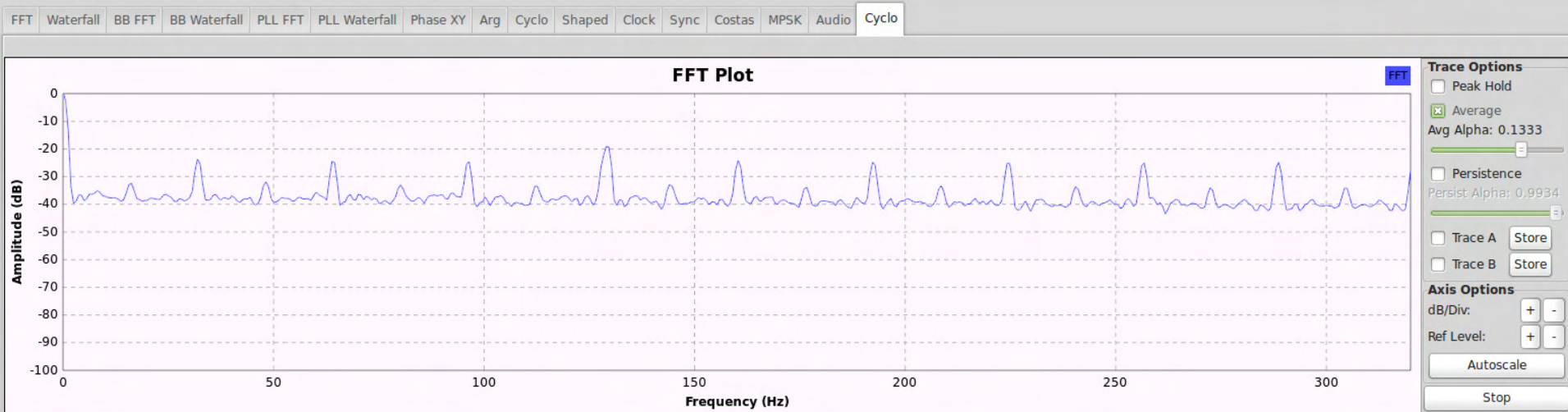
# Costas



# Clock Recovery & Sampling



# Sub-carrier Cyclostationary Analysis







# Telemetry

- The legendary Phil Karn (@ka9q) has contributed his considerable expertise & prototype convolutional decoder



Phil Karn @ka9q · Jun 2

I now claim the record for the largest Viterbi decoder ever used operationally. ISEE3's  $k=24$  is 131,072x more complex than the usual  $k=7$ .

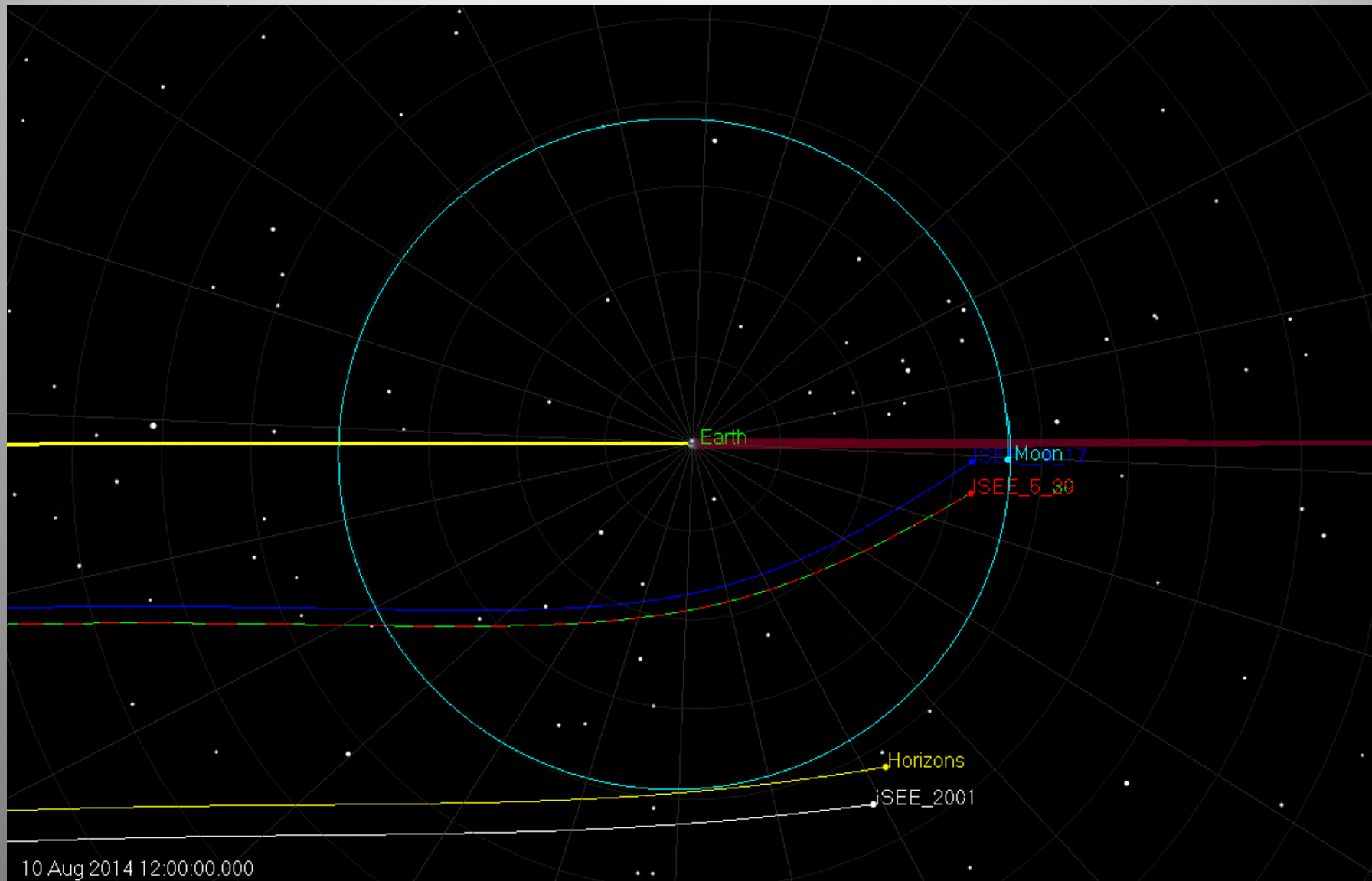
- Will attempt to GNU Radio-erise...



# Portion of Received Telemetry

- ISEE-3 Power System Status
  - Main bus voltage  $\geq 28$  volts
  - Solar array current  $\geq 5.2$  amps
  - Essential bus current  $\leq 0.25$  amps
  - Non-essential bus current = 4 amps
  - Shunt dump current  $\geq 1$  amp
  - Transponder A RF power = 5 watts
  - Transponder B RF power = 5.25 watts

# Improved Ephemeris



# Remote Operations

- RX & TX USRPs now hosted at Arecibo

The screenshot displays a software interface for remote operations, likely a GNU Radio GUI. The main window is titled "Record" and shows various parameters for recording:

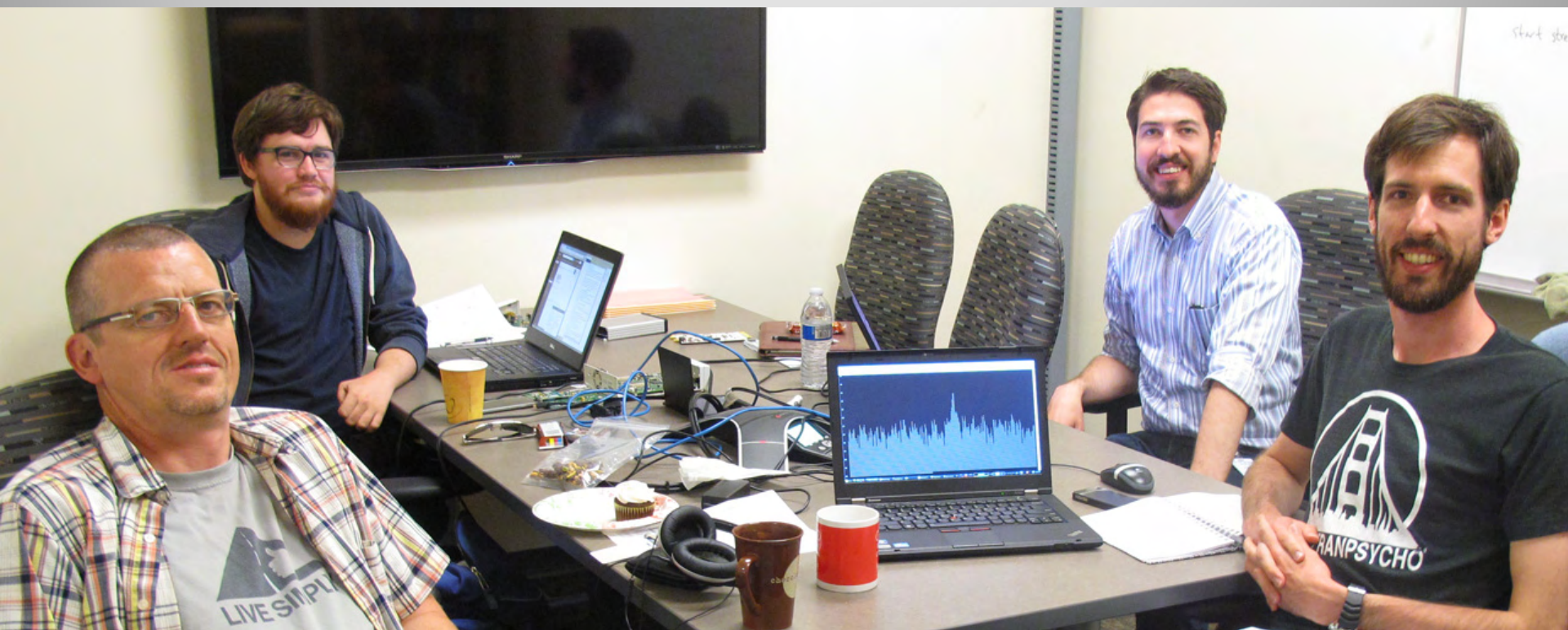
- LO Offset: 1M
- Gain: 15
- Freq: 260M
- Ant: TX/RX
- xlate\_decim: 16

The central part of the interface features an "FFT Plot" showing Amplitude (dB) versus Frequency (kHz). The plot displays a signal centered around 0 kHz, with a peak amplitude of approximately -60 dB. The x-axis ranges from -6 kHz to 6 kHz, and the y-axis ranges from -140 dB to -40 dB.

Below the FFT Plot, there is a terminal window showing the following output:

```
Click: 260.017M
RBW: 1.90735
bluegetlas ~/Desktop : ./record.py -p ~/Desktop --record-time-offset=0
Linux: GNU C++ version 4.8.1; Boost_185380; UHD_003.007.001-72-g383861d8
-- Opening a USRP2/N-Series device...
-- Current recv frame size: 1472 bytes
-- Current send frame size: 1472 bytes
UHD Warning:
Unable to set the thread priority. Performance may be negatively affected.
Please see the general application notes in the manual for instructions.
EnvironmentError: OSError: error in pthread_setschedparam
Using Volk machine: avx_64_mmx_orc
Exception: InvalidArgumentError: can't block on uhd user source (11)
```

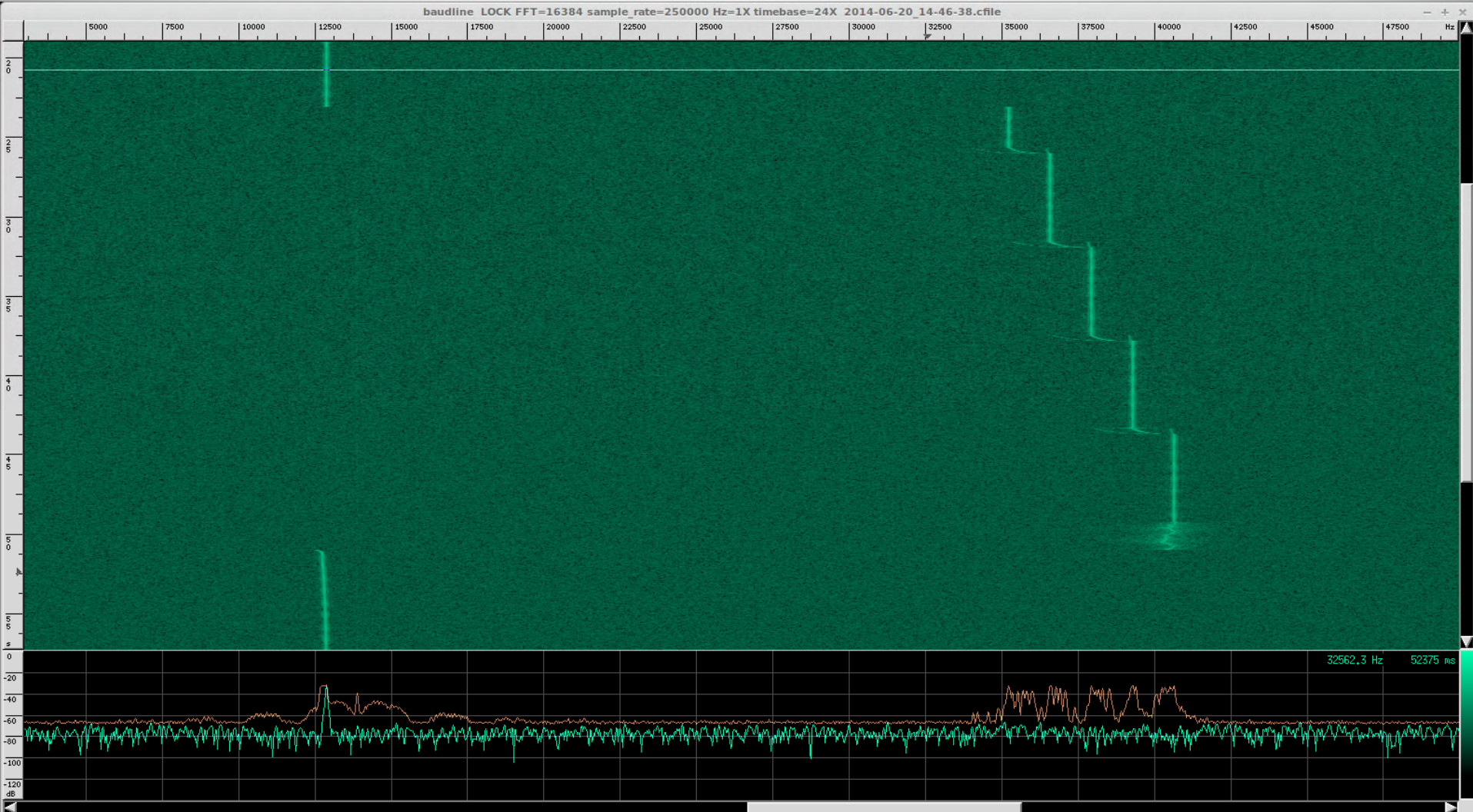
# ASCII-art FFT



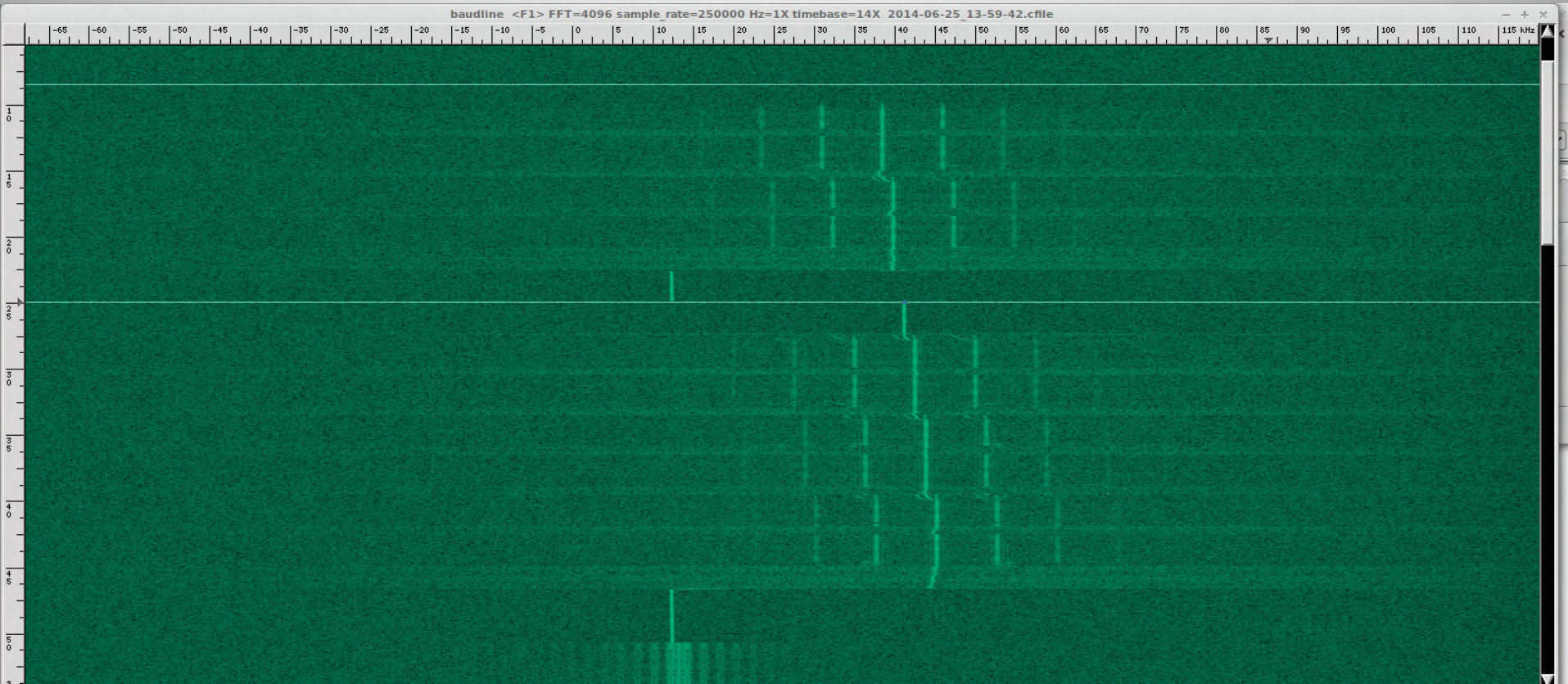
# Coherent Mode & Ranging

- Enable coherent & ranging modes to determine position & velocity of probe
- Coherent mode:
  - Downlink carrier frequency will track detected uplink carrier in fixed ratio (240/221)
- Ranging mode:
  - Downlink carrier will be modulated with demodulated uplink signal
  - Can send anything up (usually tones or PN code)

# Coherent Lock

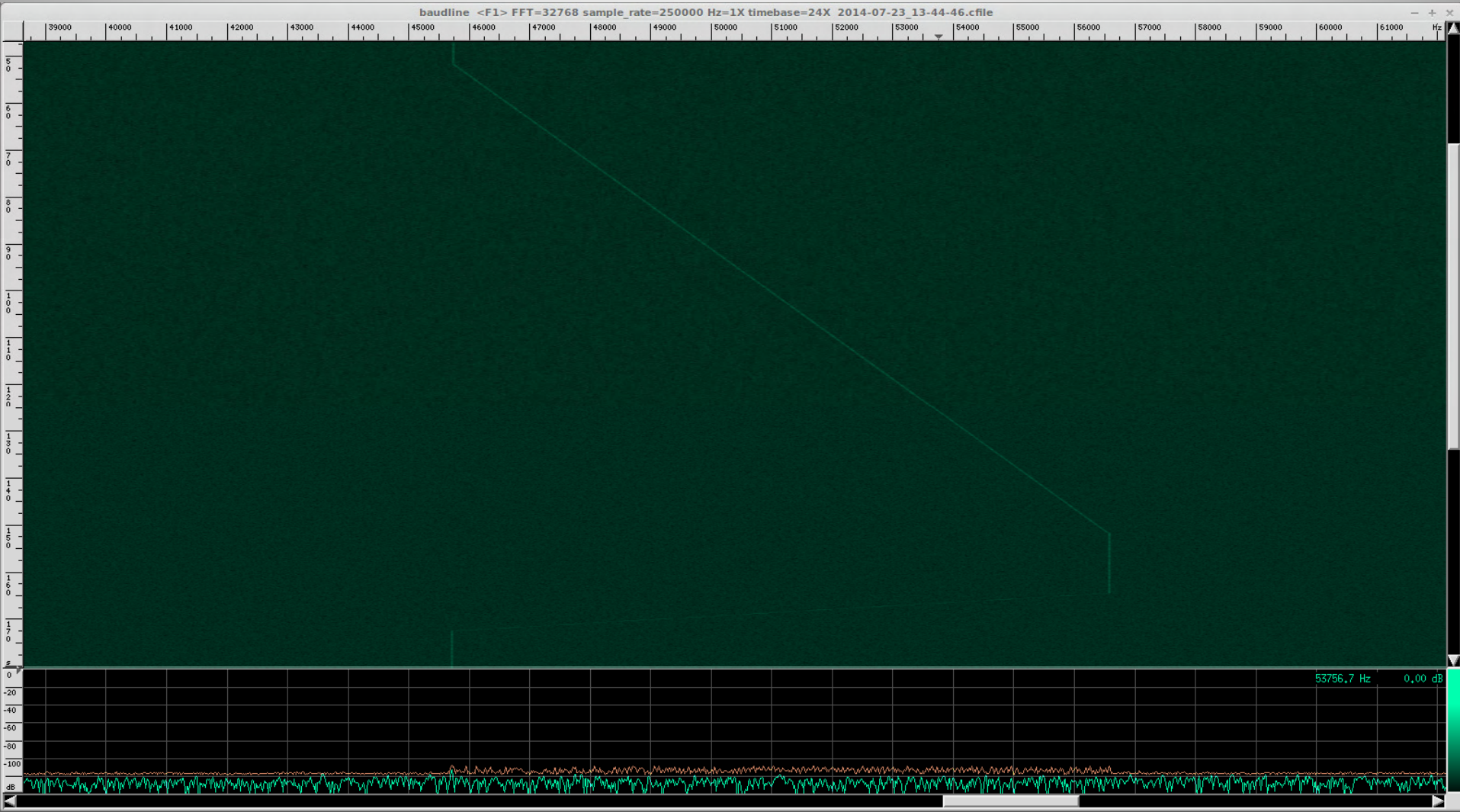


# Spectrum of Downlink

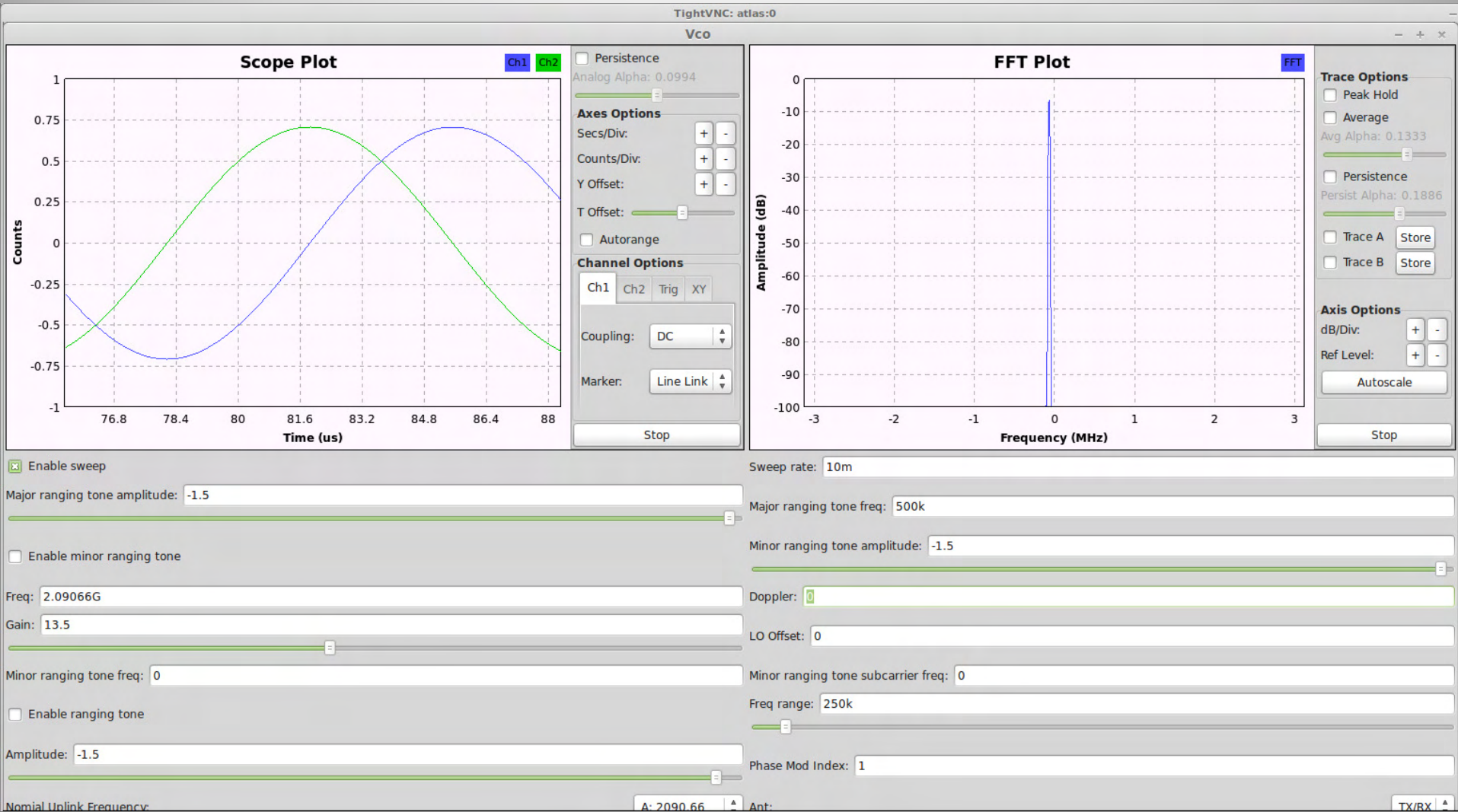




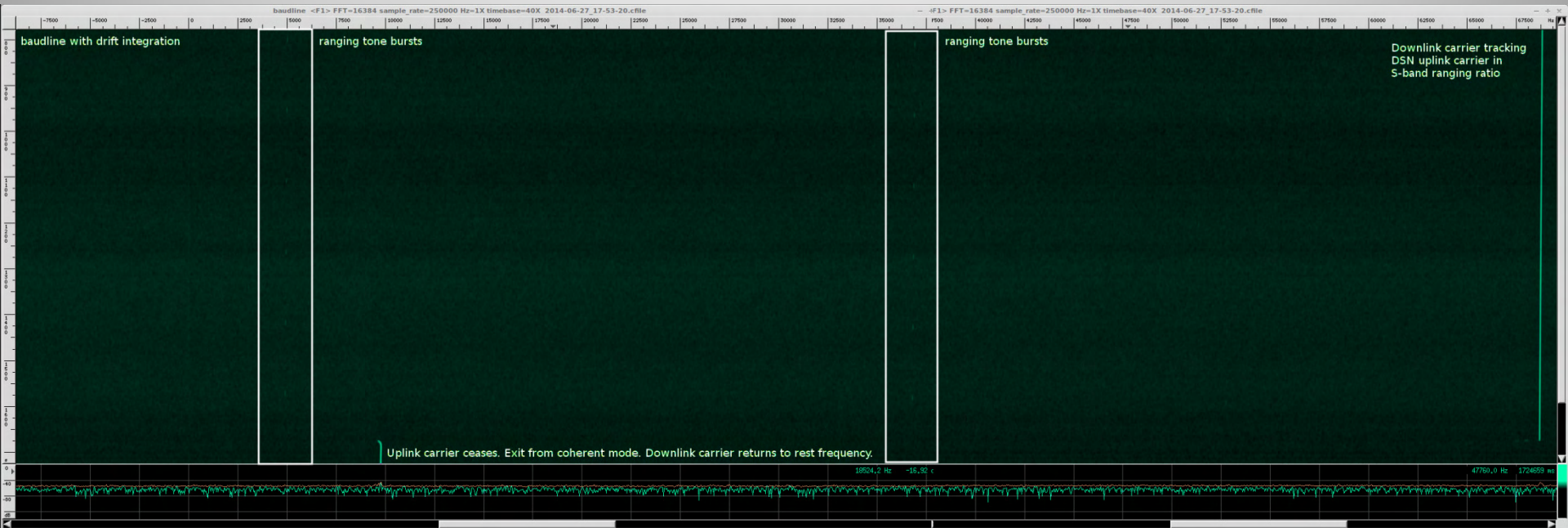
# Master Blaster Sweep

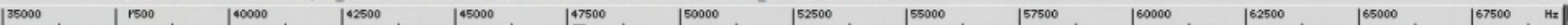


# Tone Generator



# Monitoring DSN Ranging

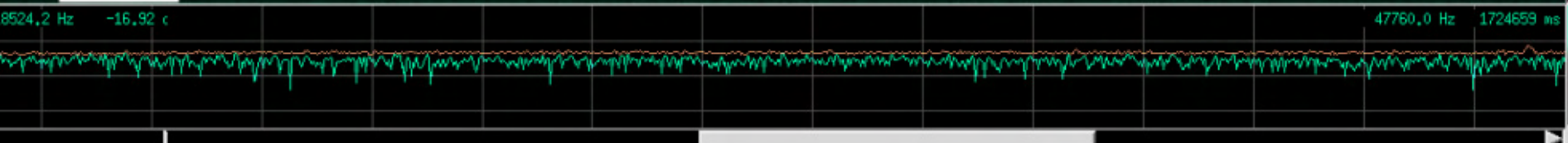




ranging tone bursts

Downlink carrier tracking  
DSN uplink carrier in  
S-band ranging ratio

cy.



# Real-time Telemetry

```
000 7c 001[02] 002 02 003 02 004 02 005 02 006 02 007 02 008 7c 009 02 010 02 011 02 012 02 013 02 014 02 015 02
016 7c 017 00 018 02 019 89 020 00 021 00 022 00 023 00 024 7c 025 1d 026 02 027 79 028 a4 029 00 030 00 031 00
032 7c 033 b2 034 02 035 00 036 7b 037 02 038 28 039 02 040 7c 041 79 042 02 043 76 044 0f 045 5e 046 64 047 36
048 7c 049 3a 050 02 051 76 052 00 053 02 054 e3 055 02 056 7c 057 dd 058 02 059 00 060 d7 061 45 062 23 063 47
064 7c 065 00 066 02 067 e0 068 00 069 00 070 00 071 00 072 7c 073 df 074 02 075 00 076 a0 077 00 078 00 079 00
080 7c 081 00 082 02 083 c9 084 b7 085 62 086 91 087 1d 088 7c 089 cc 090 02 091 02 092 00 093 08 094 00 095 00
096 7c 097 01 098 02 099 1d 100 00 101 85 102 84 103 02 104 7c 105 47 106 02 107 0e 108 a0 109 00 110 00 111 00
112 7c 113 0e 114 02 115 00 116 3f 117 f6 118 cb 119 9e 120 7c 121 47 122 02 123 12 124 fc 125 81 126 9f 127 be
```

Top Block

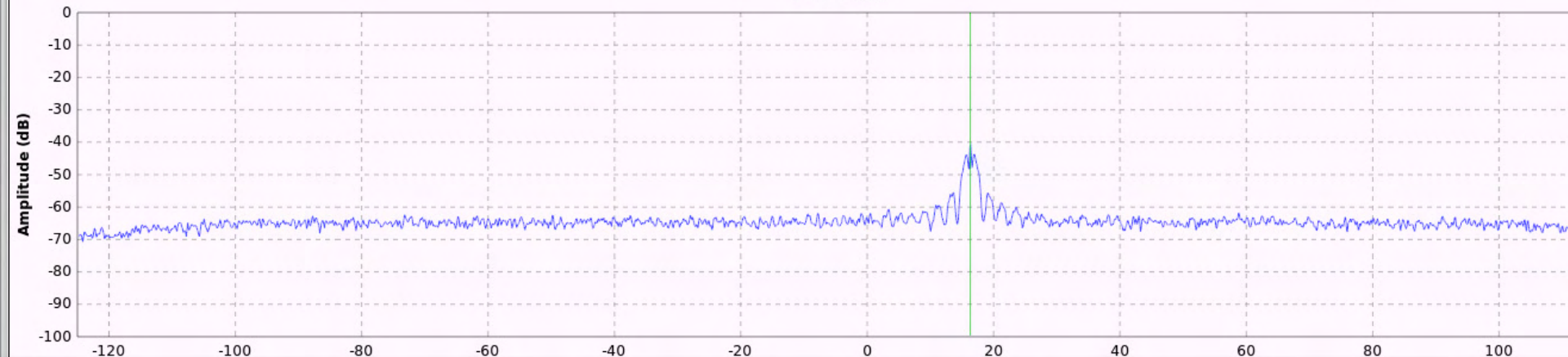
PLL Loop BW: 10

XLate Offset: 16.4714k

XLate BW: 20.48k

FFT Waterfall BB FFT BB Waterfall PLL FFT PLL Waterfall Phase XY Arg Cyclo Shaped Clock Sync Audio

FFT Plot



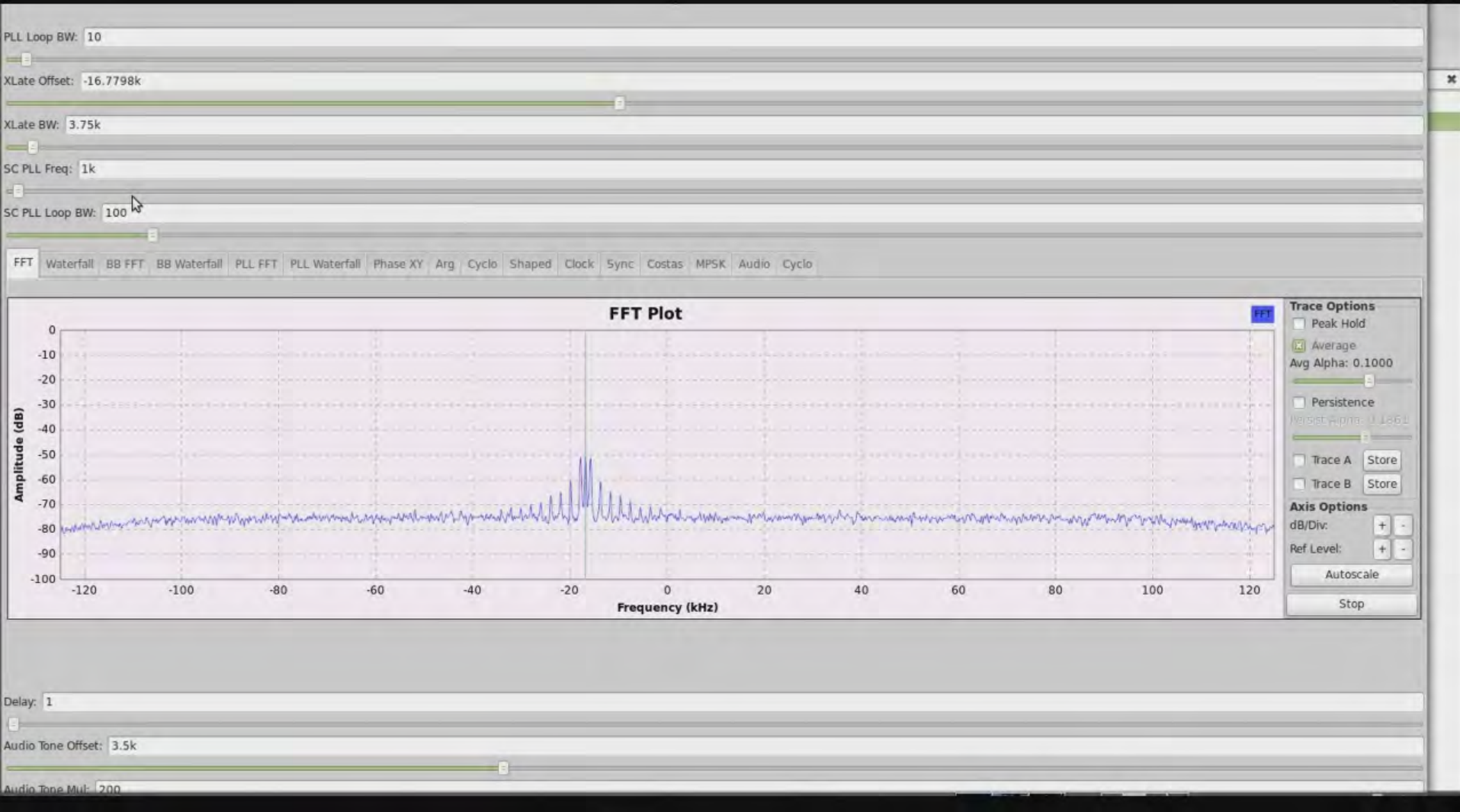


# Live Telemetry from Bochum

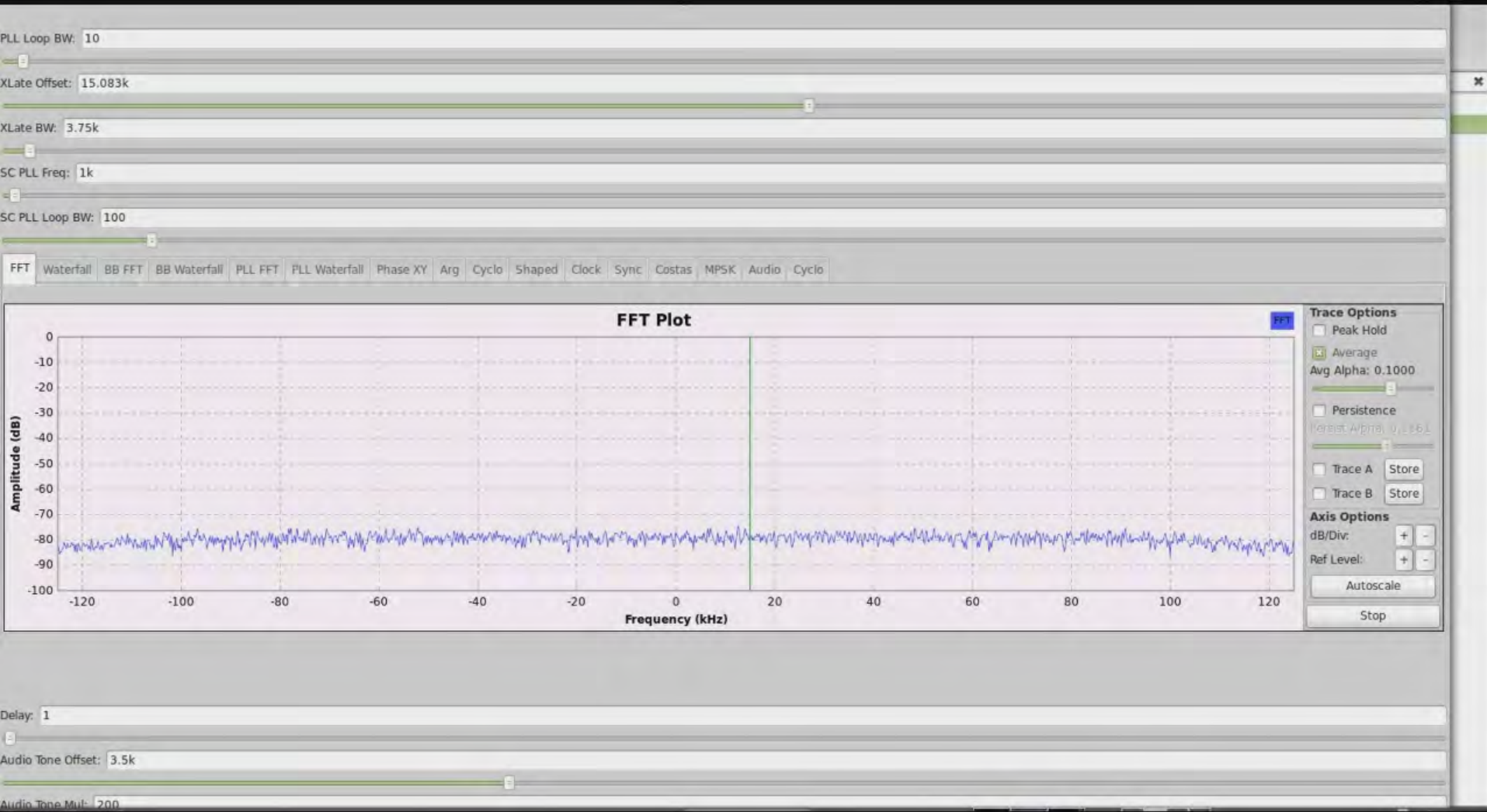
- Many thanks to our friends at AMSAT-DL



# Telemetry: 16 bps

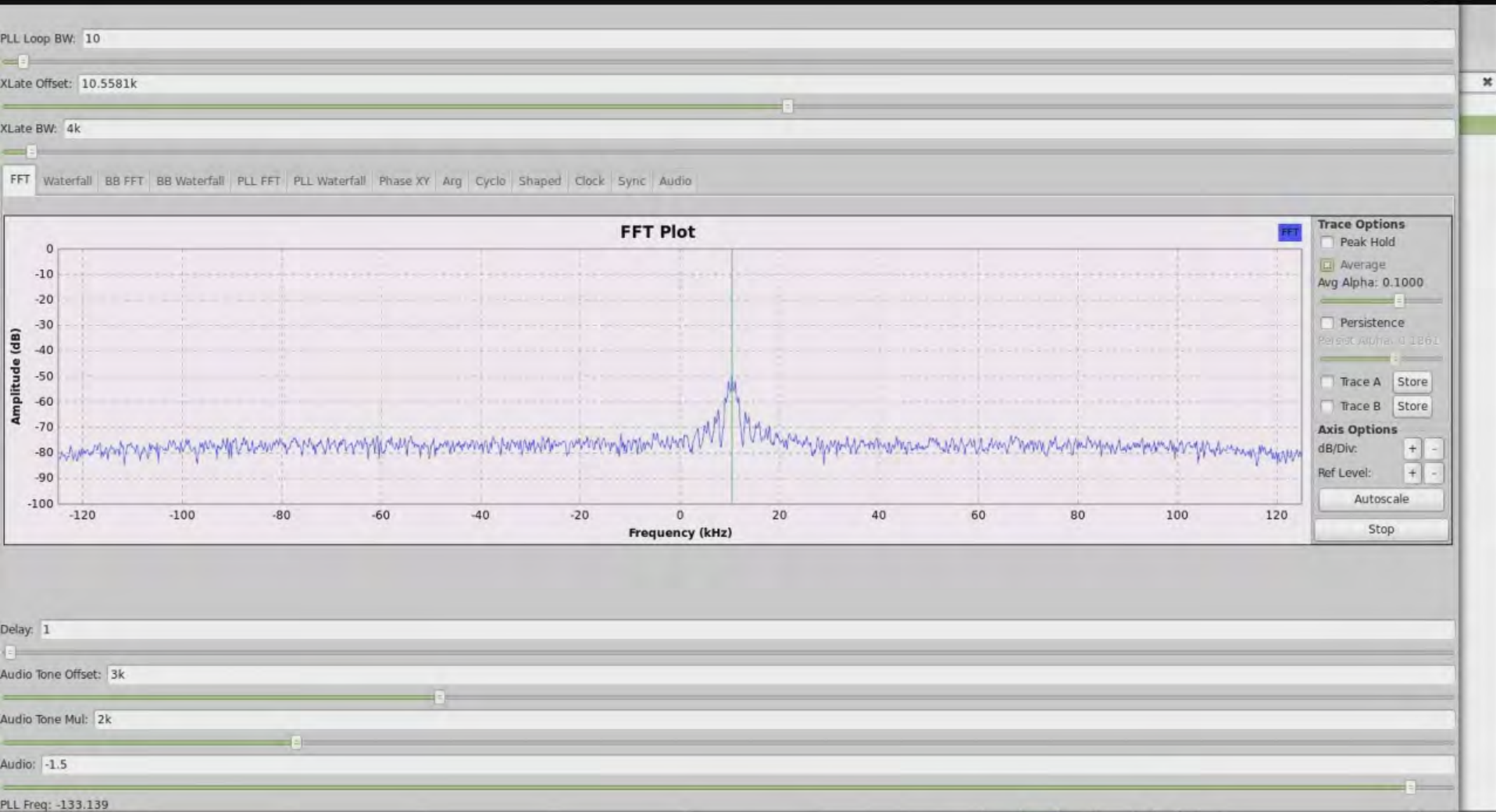


# Telemetry: 64 bps

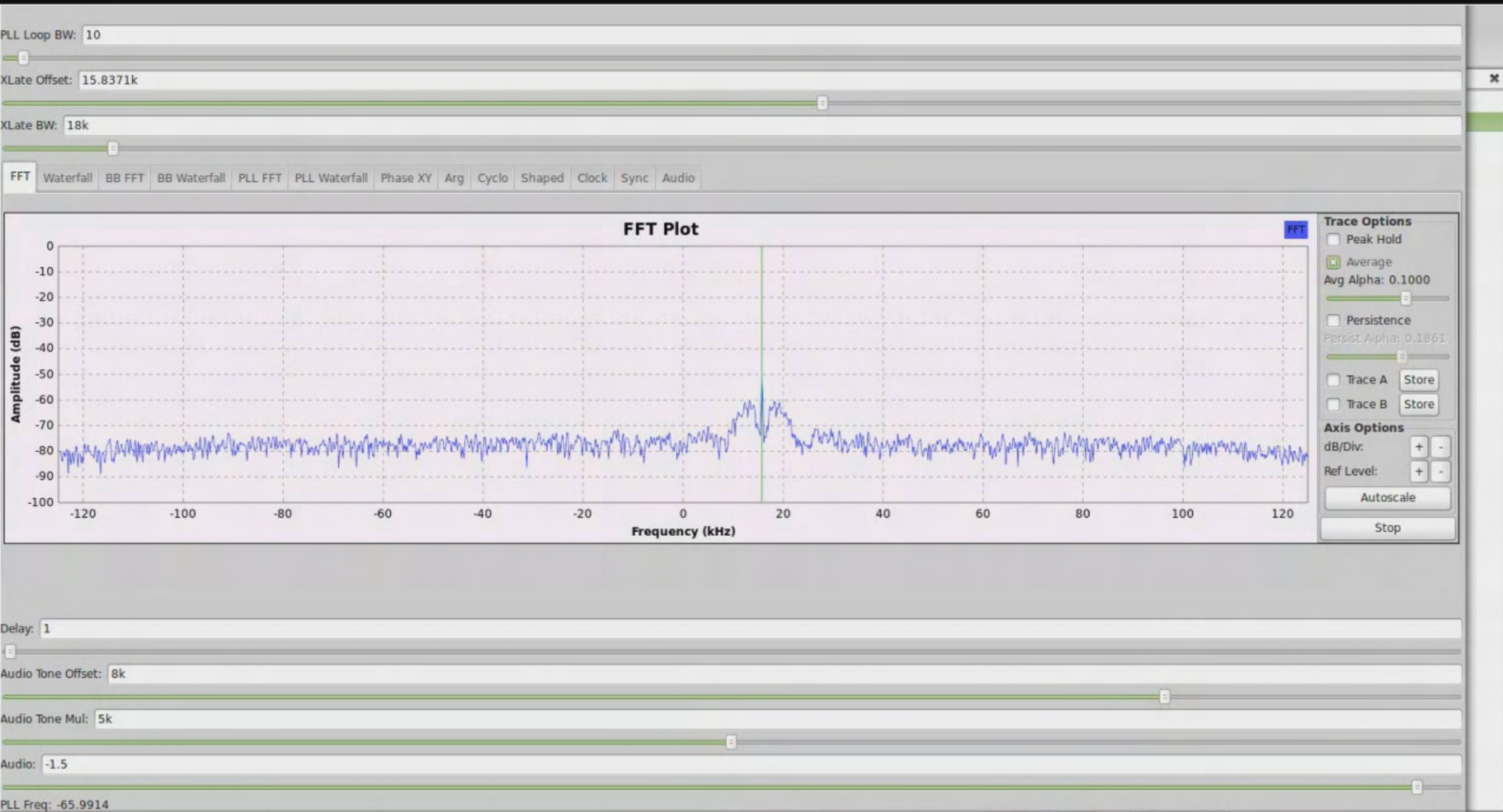




# Telemetry: 512 bps



# Telemetry: 2048 bps



# Subcoms

```
Current time: 2014-06-24 07:39:57.793797
Data arrived: 2014-06-24 07:39:57.794366
Data lag      : -0.000522      Data source: Rate: 1030, drops: 0000
Complete frame count: 5, sync reset count: 0, minor frame discontinuities: 0

AS2  000 d5  001 fe  002 fb  003 79  004 36  005 23  006 ff  007 7c  008 7d  009 fa  010 f6  011 ff  012 cd  013 90  014 00  015 02
000  016 33  017 56  018 4f  019 51  020 3a  021 00  022 5f  023 4e  024 50  025 45  026 48  027 cd  028 36  029 37  030 31  031 30
     032 9f  033 04  034 f9  035 7c  036 ff  037 00  038 6d  039 33  040 36  041 00  042 00  043 90  044 62  045 1c  046 89  047 1c
     048 00  049 27  050 74  051 77  052 e3  053 dd  054 e0  055 de  056 ca  057 cb  058 1d  059 47  060 00  061 46  062 b2  063[58]

AS1  000 7d  001 00  002 b1  003 71  004 0a  005 00  006 00  007 00  008 ca  009 2f  010 02  011 01  012 00  013 07  014 0b  015 41
000  016 52  017 ff  018 53  019 55  020 4c  021 50  022 4f  023[56] 024 55  025 4f  026 4f  027 0b  028 29  029 5a  030 40  031 46
     032 00  033 0c  034 0c  035 03  036 45  037 01  038 33  039 ed  040 00  041 ab  042 00  043 ff  044 8a  045 55  046 0f  047 0d
     048 02  049 08  050 31  051 07  052 00  053 00  054 47  055 00  056 47  057 ab  058 4c  059 33  060 ab  061 89  062 96  063 99

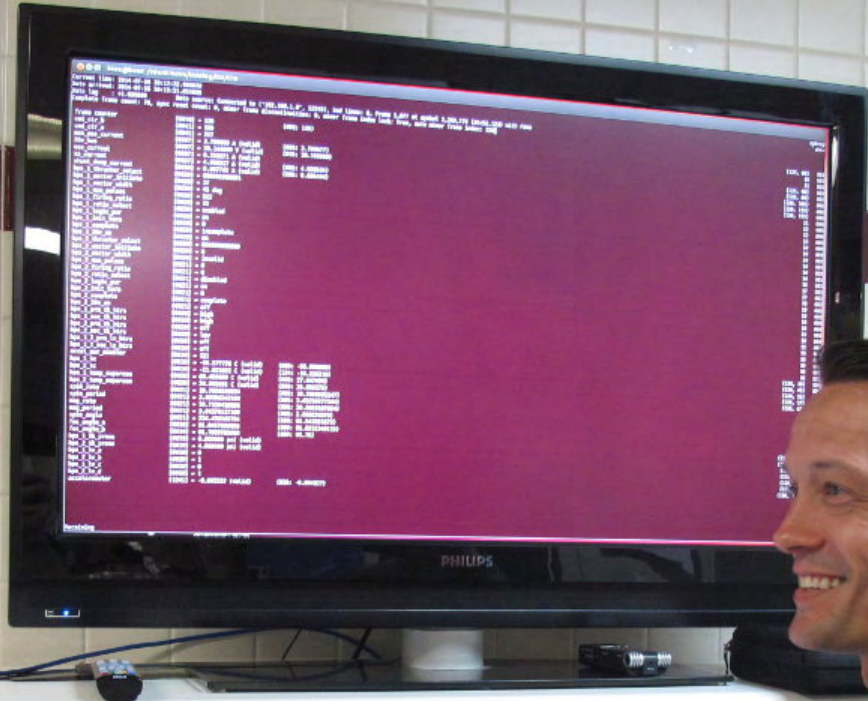
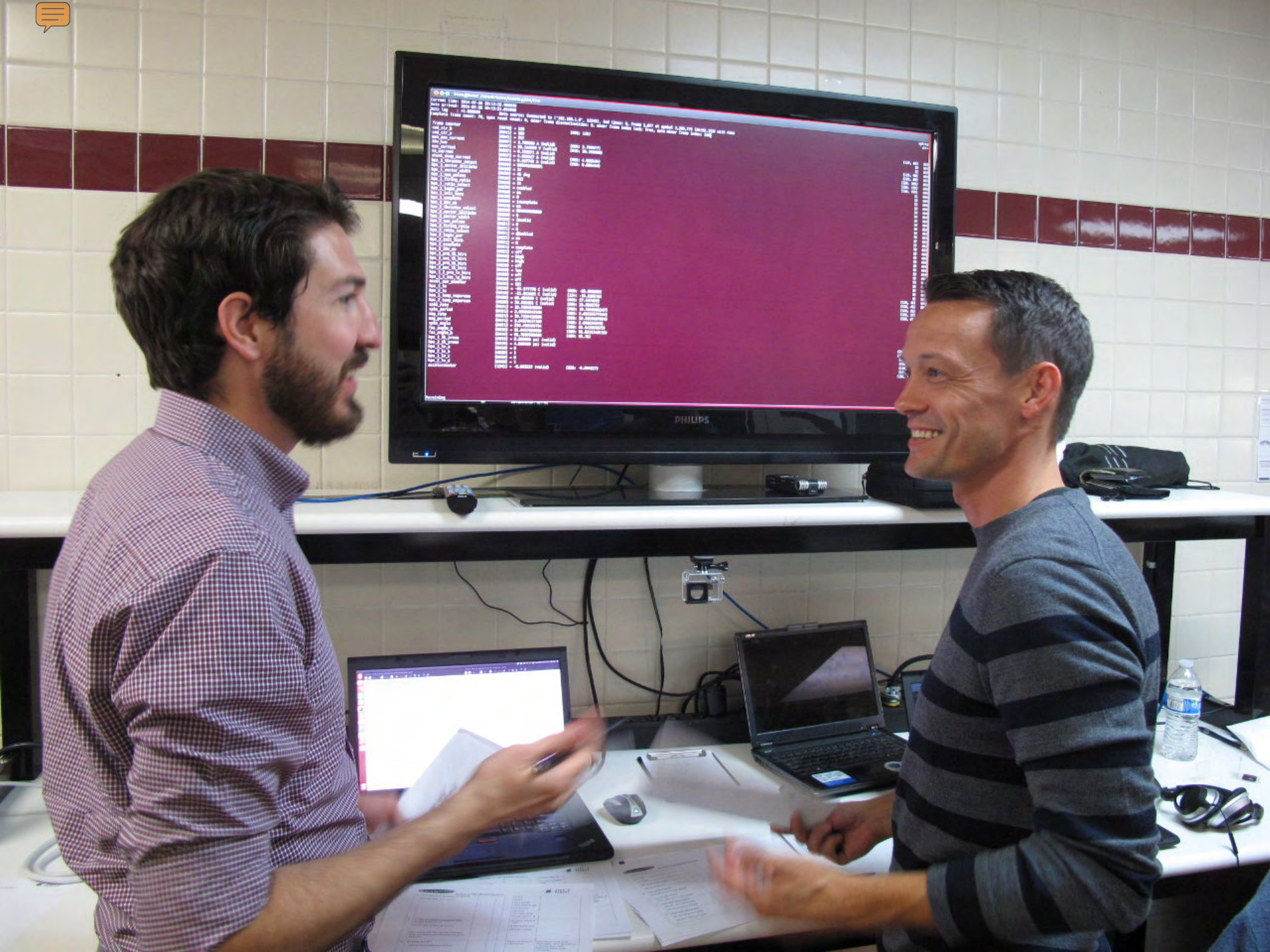
DS   000 00  001 00  002 00  003 00  004 00  005 00  006 00  007 18  008 00  009 00  010 00  011 07  012 6e  013 09  014 f8  015 00
000  016 00  017 00  018 00  019 0a  020 fb  021 54  022 00  023 f2  024 00  025 00  026 00  027 00  028 00  029 00  030 00  031 00
     032 63  033 f7  034 64  035 36  036 2f  037 b2  038 64  039[0c] 040 04  041 3c  042 0c  043 00  044 00  045 00  046 00  047 00
     048 13  049 00  050 00  051 04  052 fb  053 54  054 00  055 5c  056 00  057 aa  058 28  059 ff  060 ff  061 00  062 00  063 08
```

# Propulsion System

```
Current time: 2014-06-24 13:50:54.153003
Data arrived: 2014-06-24 13:50:54.161531
Data lag : -0.008515 Data source: Rate: 1027, drops: 0000
Complete frame count: 9, sync reset count: 3, minor frame discontinuities: 5

frame counter [0012] = 136 (001: 135) (136, 60) 001
cmd_ctr_b [0010] = 251 (008: 96) 20 001
cmd_ctr_a [0010] = 149 (008: 0) 21 000
non_ess_current [0015] = 3.951613 A (valid) (004: 3.911290) (136, 85) 000
28v_bus [0016] = 28.144000 V (valid) (008: 28.136000) (136, 86) 000
ess_current [0015] = 0.233871 A (valid) (002: 0.225806) (136, 87) 000
sa_current [0014] = 5.277778 A (valid) (000: 5.158730) (136, 101) 000
shunt_dump_current [0003] = 0.685484 A (valid) (004: 0.887097) (134, 121) 004
hps_1_thruster_select [0007] = 000000000000 (009: 010110000110) 11 001
hps_1_sector_initiate [0007] = 475 (009: 252) 12 001
hps_1_sector_width [0007] = 2 (009: 0) 12 001
hps_1_num_pulses [0008] = 4 (009: 2) 13 001
hps_1_firing_ratio [0008] = 15 (009: 1) 14 001
hps_1_ratio_select [0008] = enabled (013: disabled) 14 001
hps_1_logic_pwr [0008] = on (016: off) 14 001
hps_1_init_term [0008] = 0 (009: 1) 14 001
hps_1_complete [0008] = incomplete (017: on) 14 001
hps_1_28v_on [0008] = off (009: 011111100100) 16 001
hps_2_thruster_select [0007] = 000000000000 (008: 712) 17 001
hps_2_sector_initiate [0007] = 0 (008: 3) 17 001
hps_2_sector_width [0007] = 0 (008: 1070) 18 001
hps_2_num_pulses [0006] = 0 (015: 0) 19 001
hps_2_firing_ratio [0006] = disabled (015: enabled) 19 001
hps_2_ratio_select [0006] = on (008: off) 19 001
hps_2_logic_pwr [0006] = 0 (015: 1) 19 001
hps_2_init_term [0006] = complete (008: incomplete) 19 001
hps_2_complete [0006] = off (008: on) 19 001
hps_2_28v_on [0004] = off (003: off) 55 003
hps_1_prm_tk_htrs [0004] = low (011: low) 55 003
hps_1_sec_tk_htrs [0004] = low (003: off) 55 003
hps_2_prm_tk_htrs [0004] = low (003: off) 55 003
hps_2_sec_tk_htrs [0004] = low (003: off) 55 003
hps_1_2_prm_ln_htrs [0004] = low (003: off) 55 003
hps_1_2_sec_ln_htrs [0004] = low (003: off) 55 003
accel_pwr_monitor [0003] = 119 (136, 38) 001
hps_1_tc [0003] = -55.088889 C (valid) (001: -51.600000) (136, 41) 001
hps_2_tc [0003] = -10.810811 C (valid) (136, 51) 001
hps_1_temp_supercom [0005] = 1 (005: 145) (134, 57) 005
hps_2_temp_supercom [0004] = 1 (012: 253) (134, 67) 004
spin_rate [0003] = 19.1595852499 35 007
spin_period [0003] = 3.13159179688 35 007
mag_rate [0003] = 18.7810935769 (006: 19.2120075047) 39 006
mag_period [0003] = 3.19470214844 (006: 3.123046875) 39 006
spin_angle [0003] = 207.686910423 (007: 0.0) 35 007
fss_angle [0004] = 91.6409684294 (001: Data out of expected range) 58 001
hps_1_tk_press [0003] = 0.000000 psi (valid) (136, 115) 000
hps_2_tk_press [0003] = 4.800000 psi (valid) (136, 121) 000
hps_1_lv_a [0004] = 0 (134, 61) 004
hps_2_lv_b [0004] = 0 (134, 61) 004
hps_1_lv_c [0004] = 0 (134, 61) 004
hps_2_lv_d [0004] = 0 (134, 61) 004
accelerometer [0234] = 119 (010: 221) (137, 40) 000
```

Receiving





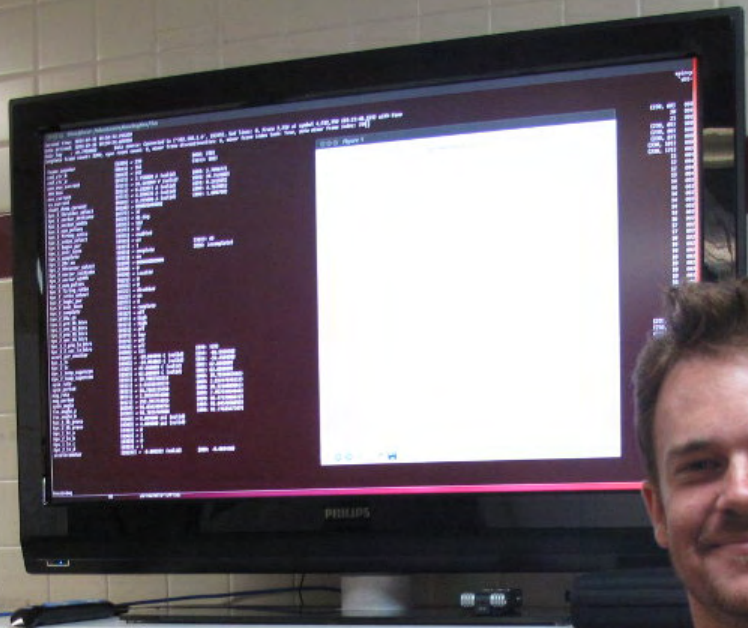




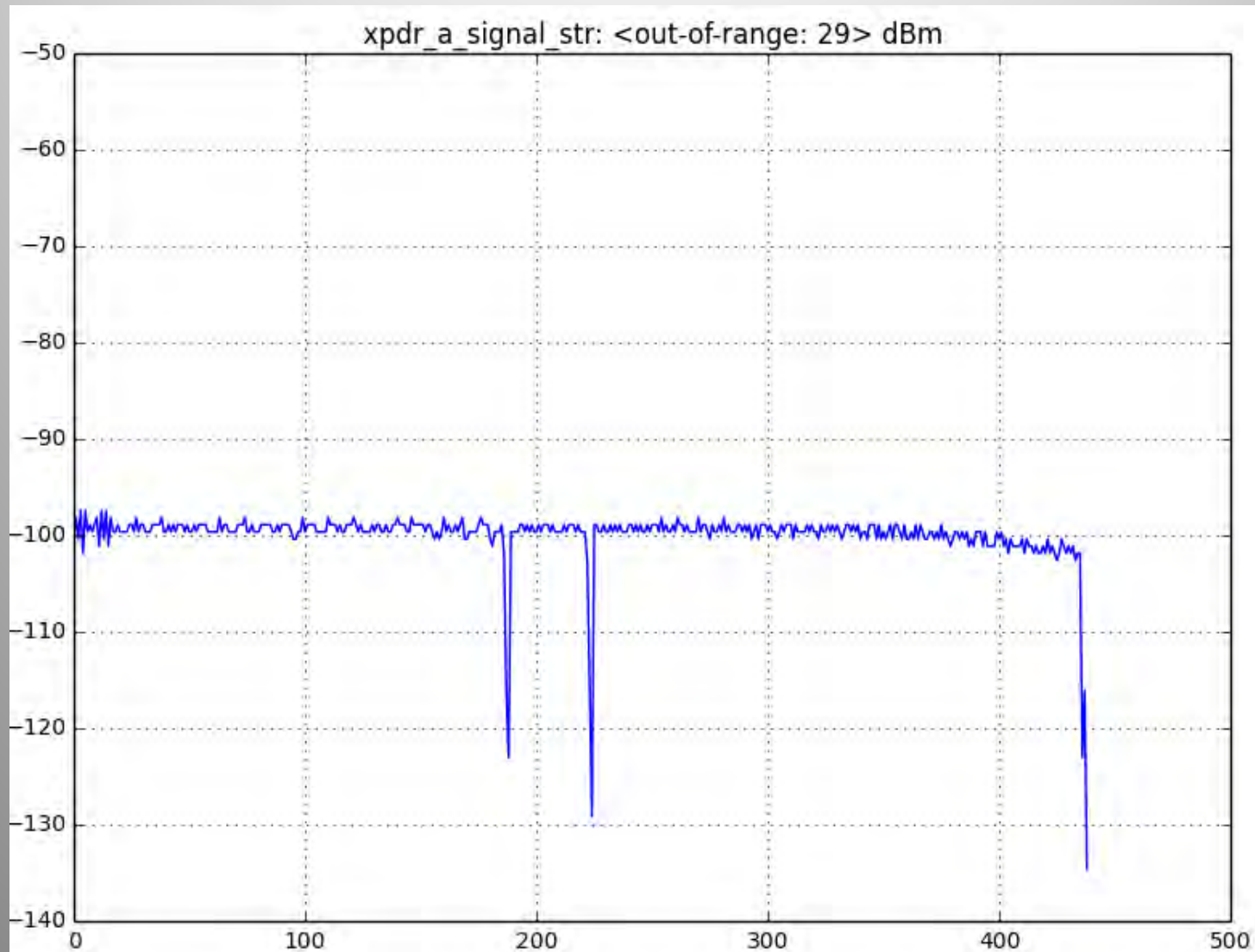




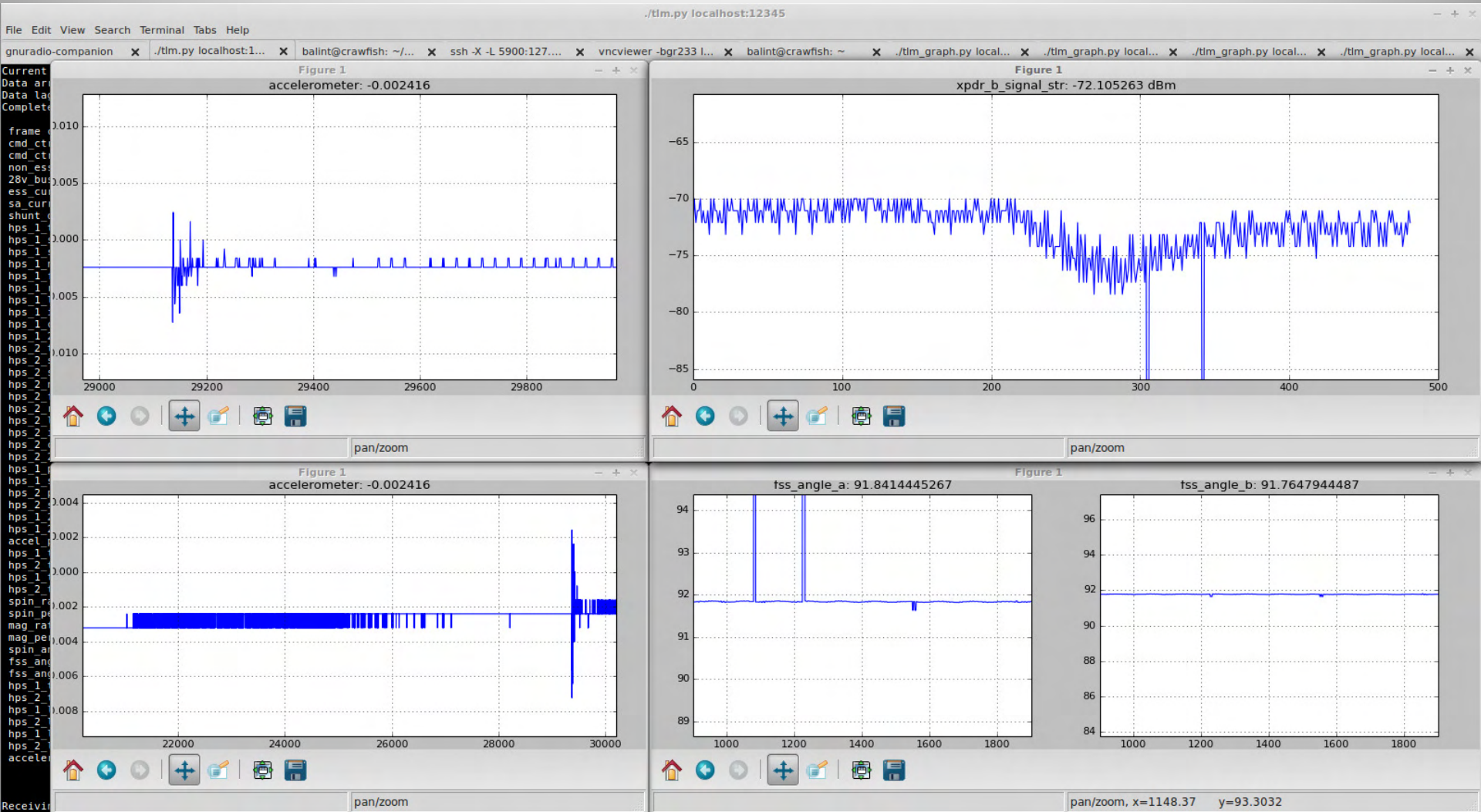
Telemetry 7/9  
B-  $\left\{ \begin{array}{l} 91.93^\circ \text{ FSS A} \\ 91.75^\circ \text{ FSS B} \end{array} \right.$   
 $\omega = 19.79 \text{ rpm}$   
 $T_1 = 255^\circ\text{C}$   
 $T_2 = 255^\circ\text{C}$   
B 90.6



# Reported Signal Strength

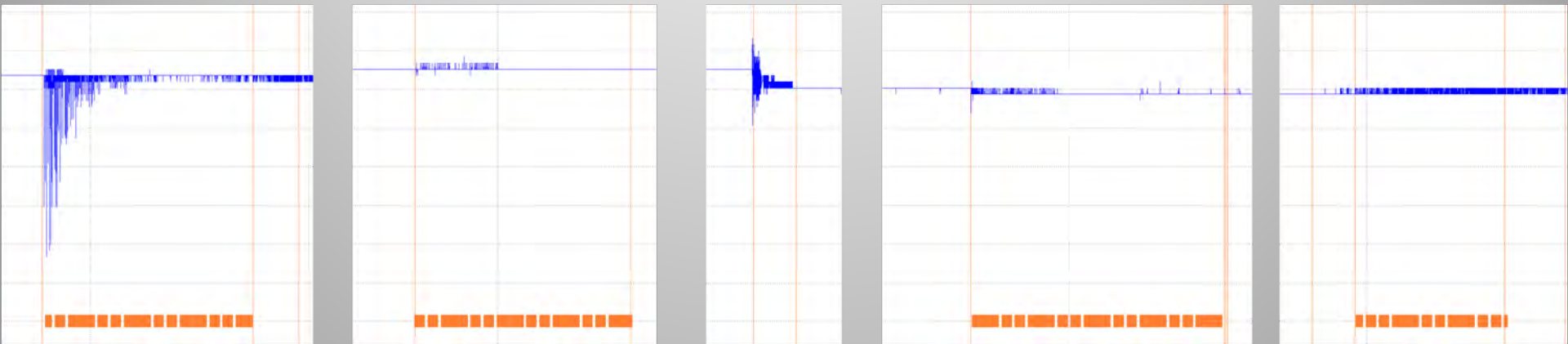


# Telemetry During Thruster Firing

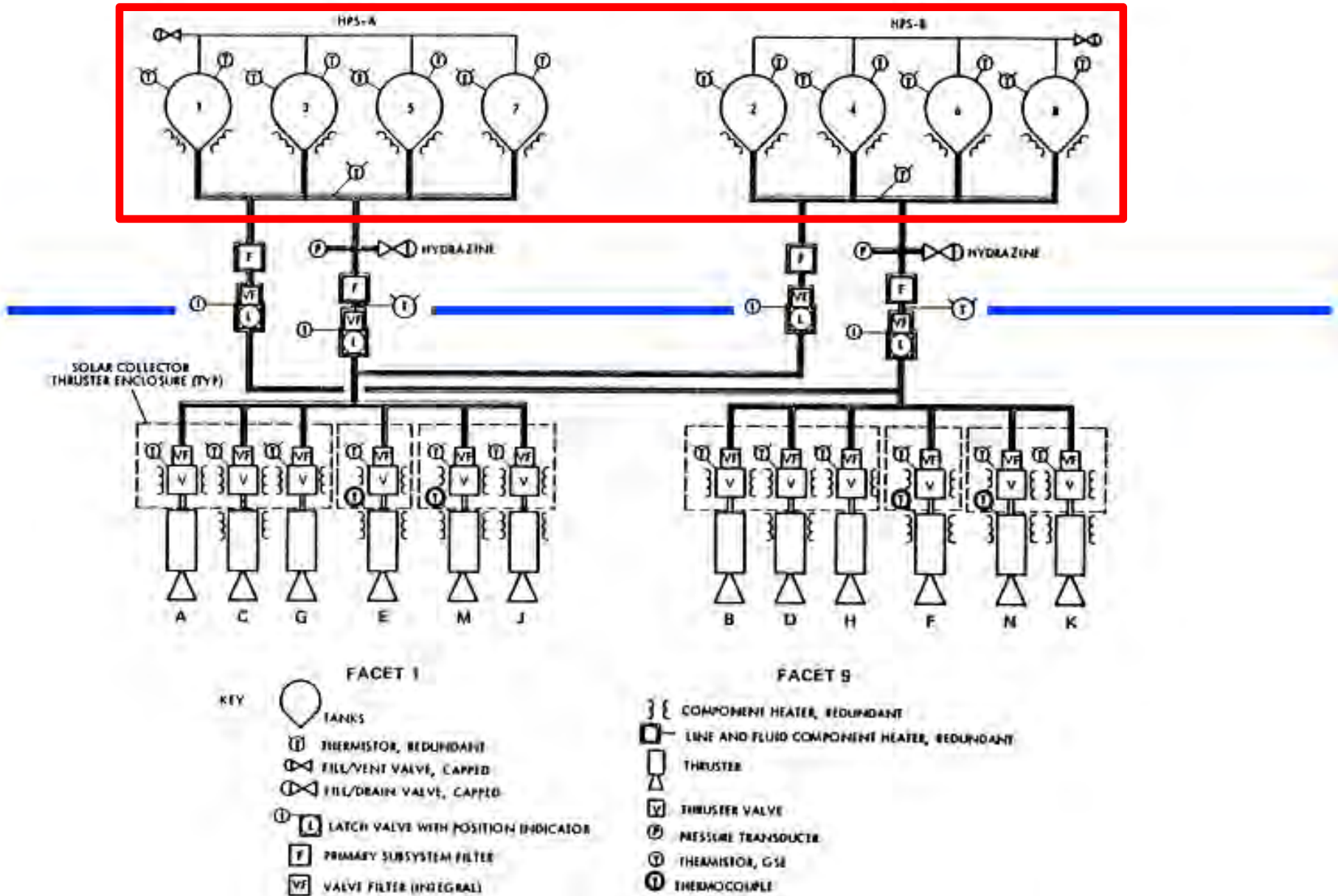




# No Thrust

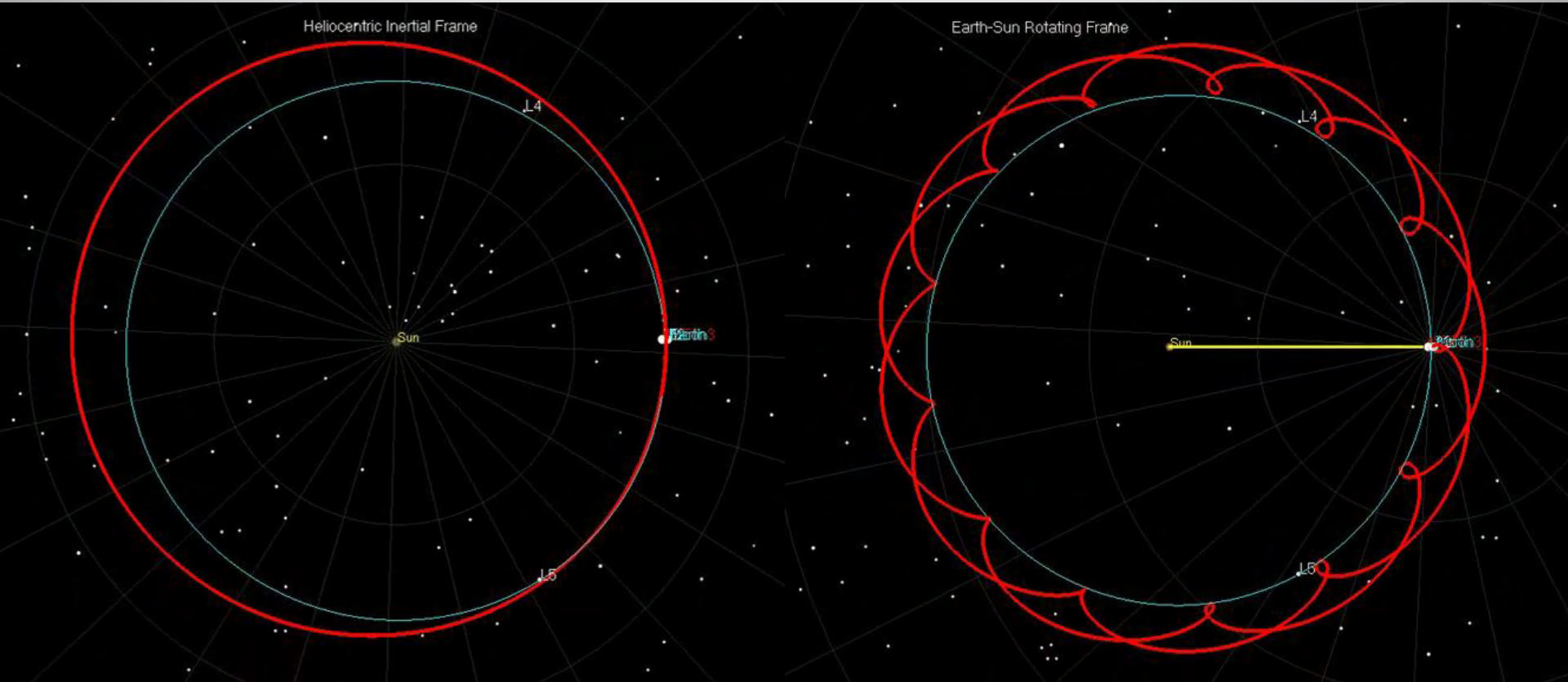


# Hydrazine Propulsion System

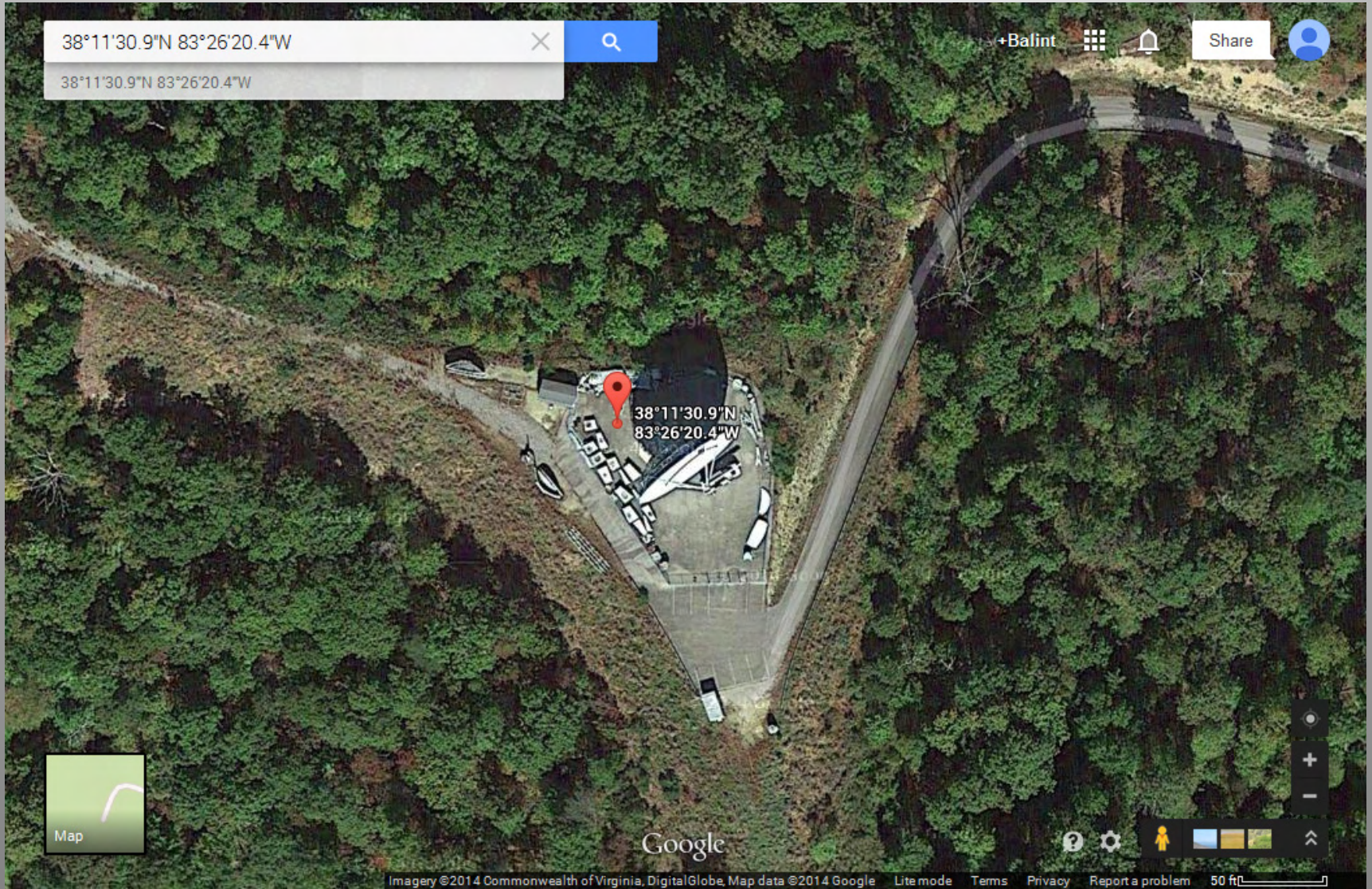




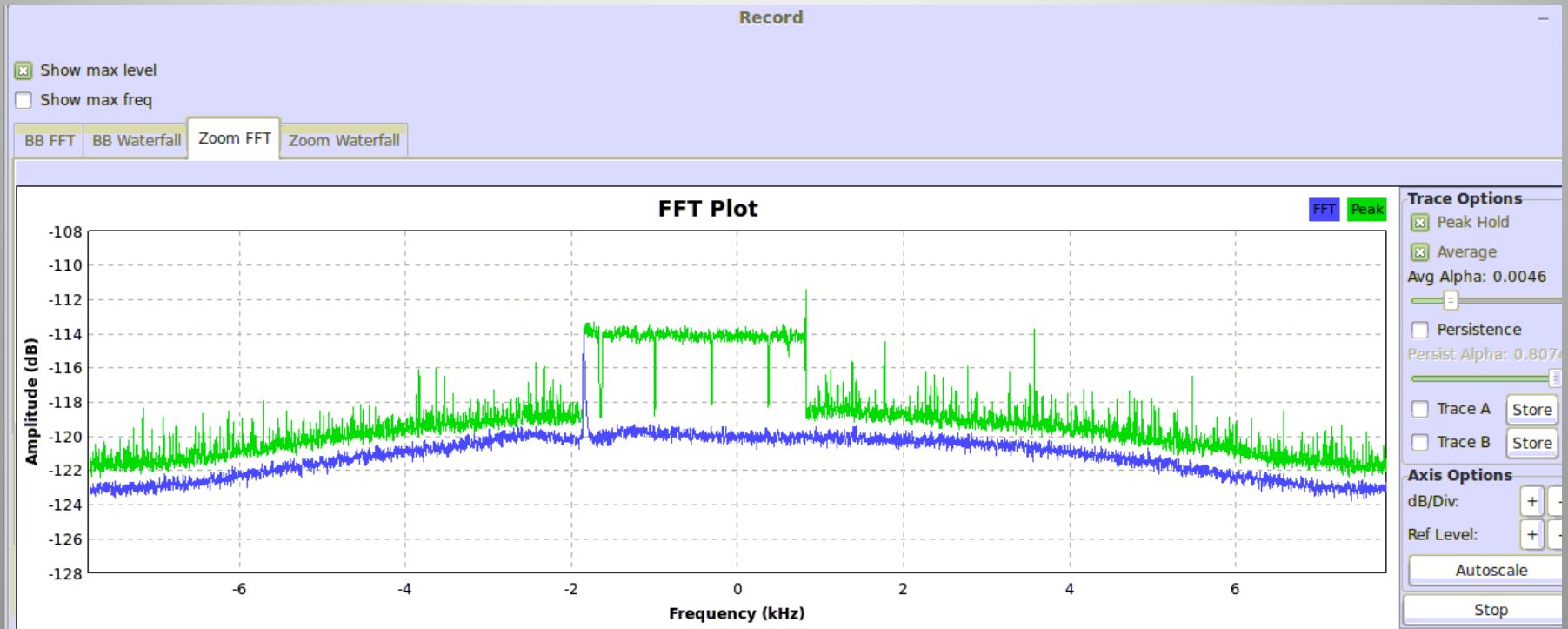
# New Orbit



# Morehead State University



# Morehead: Peak Hold



Click: 2.21748G

LO Offset: 1M

Freq: 2.2175G

XLate Decim: 16

Peak freq (BB): -24.4022k

Peak freq: -4.40216k

Peak magnitude (dB): -106.017

RBW: 1.90735

Gain: 15

Freq:

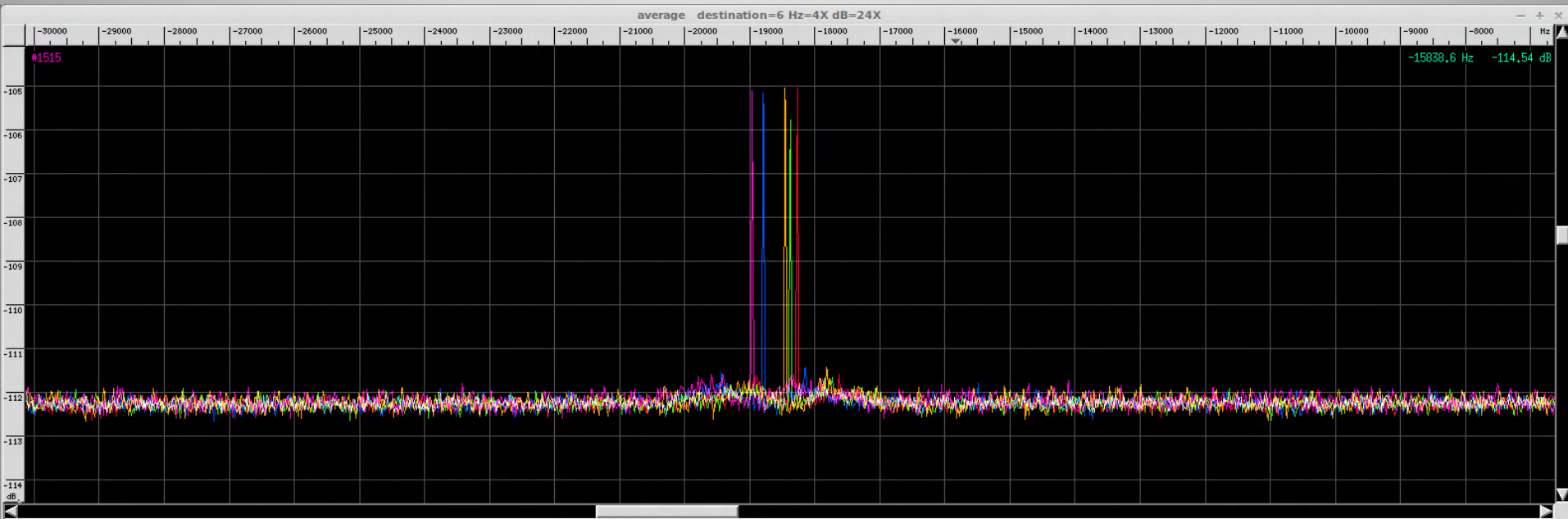
Ant:

2217.5

RX2



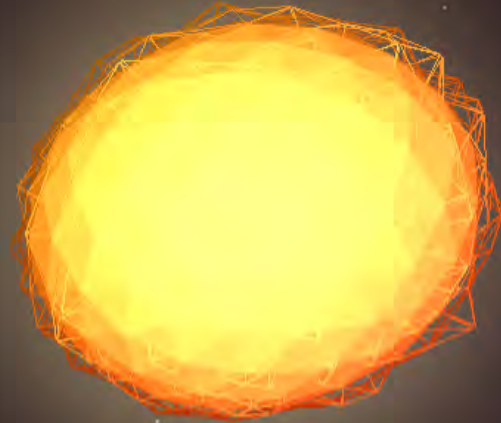
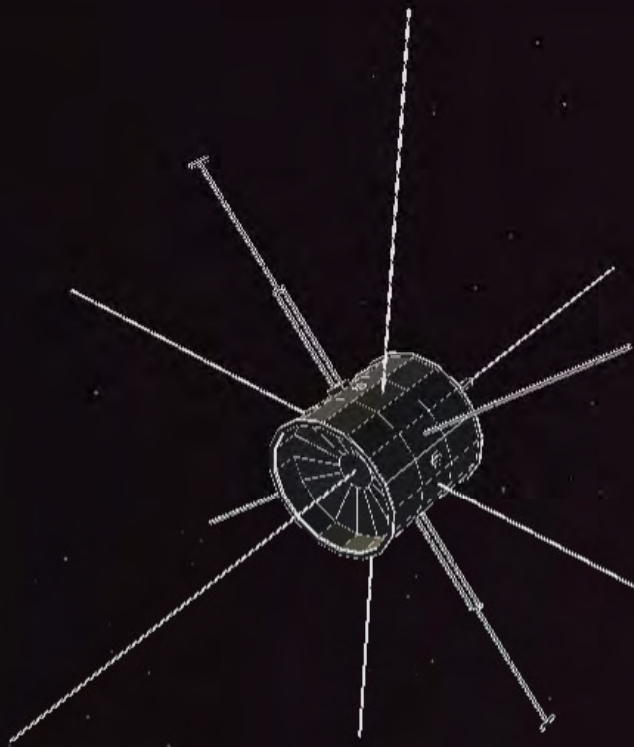
# Final Days: Weak Signal



# Mission Control Team



L-R: **Austin Epps** - Lead Engineer,  
**Jacob Gold** - Systems Engineer,  
**Cameron Woodman** - Flight Director,  
**Dennis Wingo** - Mission Director, Project Co-lead,  
**Marco Colleluori** - Attitude & Orbit Control Systems Engineer (and me).



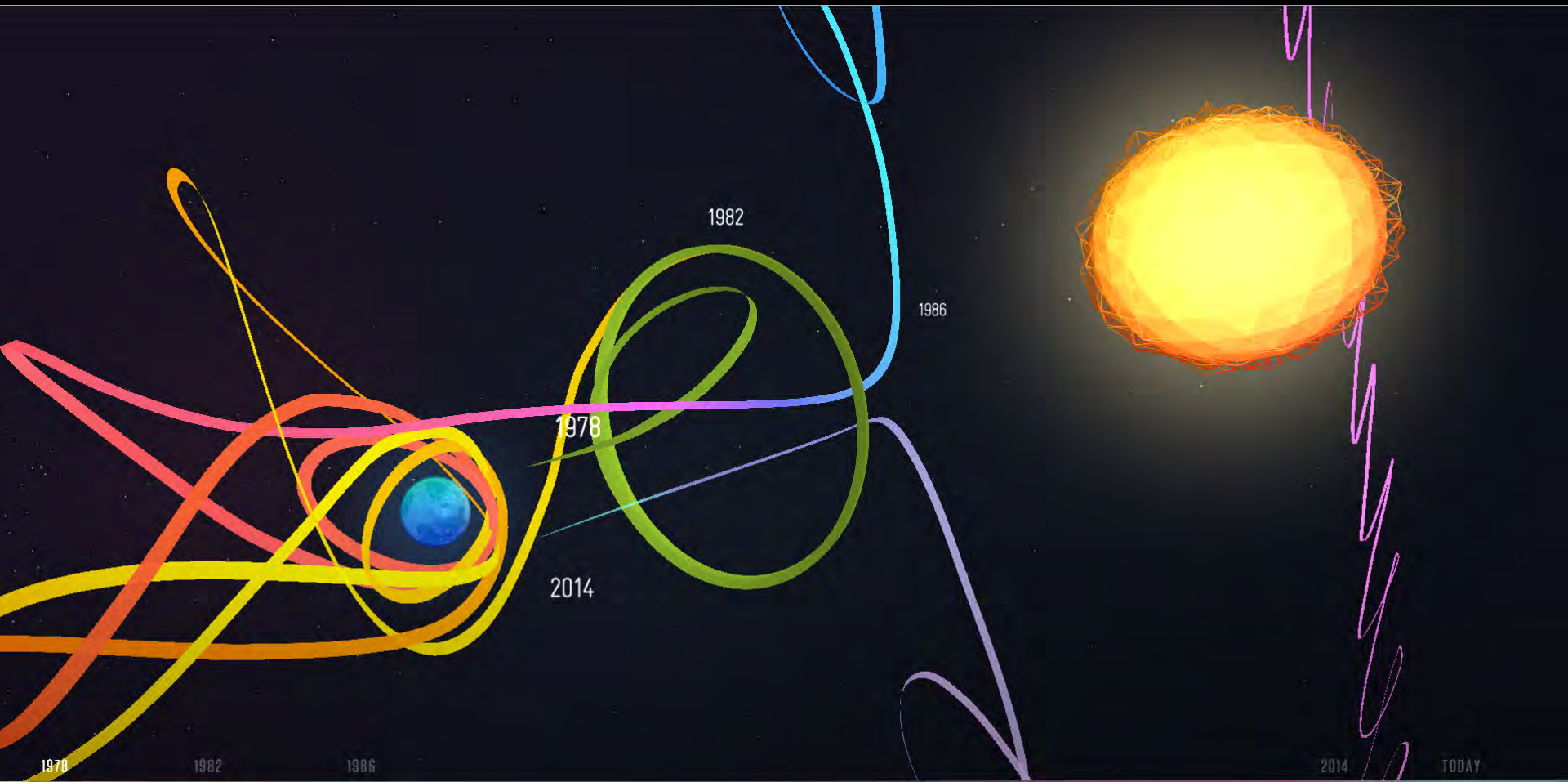
# A SPACECRAFT FOR ALL

The ISEE-3 was launched to study the Sun in 1978, but ended up redefining space flight. Now it's on a new mission to become citizen science's first spacecraft, with data accessible by everyone.

[SEE THE JOURNEY](#)[SEE LIVE VIEW](#)



# www.spacecraftforall.com







# Official Sources

- <http://spacecollege.org/isee3/>
- <http://denniswingo.wordpress.com/>
- [@ISEE3Reboot](#)
- <http://www.facebook.com/ISEE3Reboot>

# My Coverage

- <http://twitter.com/spenchdotnet>
- [http://www.youtube.com/playlist?list=PLPmwWVknVliUIPbkfBUY1ebP\\_8hA\\_4q8j](http://www.youtube.com/playlist?list=PLPmwWVknVliUIPbkfBUY1ebP_8hA_4q8j)
- <http://gallery.spench.net/v/Arecibo/>

# More Links

- Telemetry parser/display:  
<https://github.com/balint256/ice/>
- Archived telemetry frames:  
<http://spench.net/~balint/public/isee-3/>
- Phil Karn's decoder:  
<http://www.ka9q.net/isee.html>
- John Malsbury's post about the project:  
<http://www.jmalsbury.com/how-to-talk-to-a-36-year-old-space-probe-isee-3-with-gnu-radio-a-usrp-and-a-big-dish/>





Ooops...

